ANALYSIS OF CONTINUOUS BEAMS BY INFINITE SERIES

By

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ANALYSIS OF CONTINUOUS BEAMS BY INFINITE SERIES

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PREFACE

The author of this thesis was ordered by the United States Air Force to work towards his Master of Science Degree in Mechanical Engineering with a special emphasis on structural problems.

A special agreement between the School of Mechanical Engineering and the School of Civil Engineering has been reached to make the realization of the United States Air Force's order possible.

The School of Civil Engineering, having a large structural research program sponsored by the Robberson Steel Company, Oklahoma City, Oklahoma, permitted the author to take part in this research activity and to select a topic for his thesis connected with this program.

A special three man advisory committee, appointed to supervise the author's thesis writing, was composed of William H. Easton, Professor of Mechanical Engineering and chairman of the author's thesis committee; Jan J. Tuma, Associate Professor of Civil Engineering in charge of the Robberson Steel Research Project, and Raymond E. Chapel, Assistant Professor of Mechanical Engineering.

The writer wishes to acknowledge his indebtness to Professor Easton for his early reading of this thesis and for his helpful encouragement during the entire academic year, and to Professor Tuma, who originated this type of analysis in preliminary form, for his valuable guidance and careful checking of all the writer's derivations. He extends his thanks to Roger L. Flanders, Professor and acting head of the School of Civil Engineering, and to Professor Chapel, for their reading of the manuscript and much helpful advice.

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NOMENCLATURE

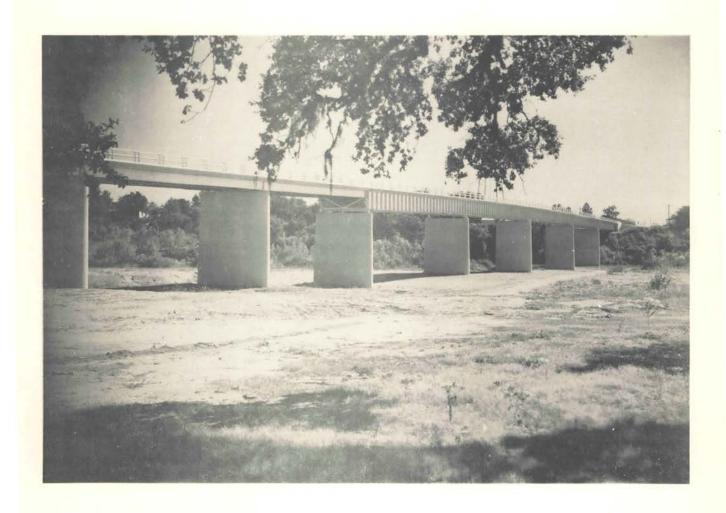
a,b	, c	d,	e,	f	g,	h,	j,	k,	m,	n,	p,	ŗ			•	×	Functions of New Distribution Factor
s,t	, u	, v,	W_g	S	т,	V	۰	٠	•		•	•			*	•	Equivalents
x,y	, Z		•	0		٠				۰	٠	0	0	٠			Denominators of convergency
A, B	, C	, D,	E,	F	, G,	Н,	J		0	٠,		٠			٥		Points or ends of beams
FM	0	٥	•	0	0	۰	0	o		0		0	0	0	0		Fixed end moment
EM*		0	•	•	0	0	o	0		۰		•		o		0	Propped beam end moment
I.	۰	0			0	•		0	0	o	0	0	۰	Q	۰		Moment of inertia
L.	•	0		0	0	•	0	•	p	0	0	0	•		•	•	Length
к.	۰	0	0	•	•	0	0	0	٠	0		•		0	0	۰	Stiffness factor
Μ.	0	0	.0	0	۰	0	•	0			0		۰	0			End moment

SIGN CONVENTION

The slope deflection sign convention is used throughout this paper.

According to this convention, all moments are positive if they act in a clockwise sense on the member, and are negative if they act in a counter-clockwise sense on the member.

Section 1



(Courtesy of Texas Highway Department)

Colorado River Bridge on State Highway 71 near Smithville, Texas.

SECTION I GENERAL THEORY

INTRODUCTION

Although the moment distribution method became a very important and widely used tool of structural analysis during the last two decades, no systematic effort has been made:

- 1. To investigate the internal functional mechanics of this procedure;
- 2. To define it in general algebraic terms;
- 3. To replace the n-cycle procedure by one direct function, clearly expressed in terms of structural identities and loads applied.

This paper is intended to be a contribution to the above mentioned effort presenting:

- A complete mathematical analysis of the moment distribution method applied to the computation of the end moments in continuous beams.
- 2. A derivation of the general end moment equations for a limited

This method was developed in connection with the calculation of secondary stresses in trusses and is described in the book by 0. Mohr, "Abhandlungen aus dem Gebiete der Technischen Mechanik", p. 429, 1906 and was extended to the analysis of statically indeterminate frames by K. A. Čalisev (Techničné Listy 1923 No. 17-21, Zagreb). The final form of the numerical method of successive approximations was obtained in the paper by H. Cross (Transactions, Am. Soc. C.E., Vol. 96, 1932) and is without a doubt the most important contribution to structural analysis in recent years.

number of cases (three, four, five, six, seven and eight span beams).

3. A direct, simple and workable procedure for the computation of end moments in continuous beams due to static or moving loads, due to settlement or rotation of supports.

Although only beams with a constant moment of inertia are analyzed, it is evident that all the principles apply to beams or frames with a variable moment of inertia.

Tables for twenty-four specific cases were prepared and their application demonstrated by three typical problems.

1. BASIC SERIES

If the method of moment distribution is used to compute the end moments for three or four span beams and the balancing procedure is expanded algebraically, each moment consists of a series that is:

- 1. Infinite (number of terms = \infinite),
- 2. Convergent (ratio of two successive terms < 1),
- 3. Geometric (ratio of two successive terms is constant).

The sum of all the terms in each series forms the final moment, which contains two separate and independent functions (2)(3):

- 1. New Distribution Factor,
- 2. Fixed End Moment.

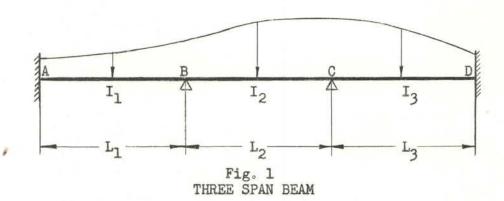
²Jan J. Tuma, <u>Wind Stress Analysis of One Story Bents by New Distribution Factor</u> (Oklahoma Engineering Experiment Station Publication No. 80, May, 1951).

May, 1951).

3Jan J. Tuma, Influence Lines For Frames With Constant Moment of Inertia and Sidesway Prevented (Oklahoma Engineering Experiment Station Publication No. 85, November, 1952).

2. BASIC SERIES FOR A THREE SPAN BEAM

In the following illustration of the Basic Series, the end moment equations will be derived for a three span cyclosymmetrical beam, loaded as shown in Fig. 1.



Denoting the fixed end moments due to the applied loads as $-FM_{AB}$, FM_{BA} , $-FM_{BC}$, FM_{CB} , $-FM_{CD}$, and FM_{DC} , the sums of the fixed end moments are at Point B

$$B = FM_{BA} = FM_{BC}, \tag{1a}$$

and at Point C

$$C = FM_{CB} - FM_{CD}. \tag{1b}$$

The distribution factors are denoted

at Point B as

$$D_{BA} = 2a, D_{BC} = 2b, \qquad (2a)$$

and at Point C as

$$D_{CB} = 2c, D_{CD} = 2d.$$
 (2b)

The distribution of the fixed end moments at Points B and C is presented in Table 1.

TABLE 1

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR A THREE SPAN BEAM

Points: • Distribution	0	2a	2b	2e	2d	0
Factors: Fixed End Moments:	~FM _{AB}	FM _{BA}	-FM _{BC}	FM _{CB}	-FM _{CD}	FM _{DC}
lst Cycle		-2aB	-2bB	-2cC	=2dC	
Carry-over	-aB		-cC	-bB		-dC
2nd Cycle		2acC	2bcC	2bcB	2bdB	1.
Carry-over	acC	-	bcB	bcC		bdB
3rd Cycle		-2abcB	−2b ² cB	-2bc ² C	-2bcdC	
Carry-over	-abcB		-bc ² C	-b ² cB		-bcdC
4th Cycle		2abc ² C	2b ² c ² C	2b ² c ² B	2b ² cdB	
Carry-over	abc ² C		b ² c ² B	b ² c ² C		b ² cdB
5th Cycle		-2ab ² c ² B	-2b ³ c ² B	-2b ² e ³ 0	-2b ² c ² dC	
Infinite Cycle	0	0	0	0	0	0
Sum of all the terms in each	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}

Summing all the terms in column \mathbf{M}_{AB} of Table 1,

$$M_{AB} = -FM_{AB} - \frac{a}{x}B + \frac{ac}{x}C, \qquad (3)$$

in which

$$\frac{1}{x} = 1 + bc + (bc)^2 + (bc)^3 + \dots + 0$$
 (3a)

and

$$x = 1 - bc$$
 (denominator of Convergency for \overline{AD}). (3b)

All moment equations similarly derived are presented in Section 2, Table C3.

3. BASIC SERIES FOR A FOUR SPAN BEAM

As a second illustration of the Basic Series, the end moment equations will be derived for a four span cyclosymmetrical beam, loaded as shown in Fig. 2.

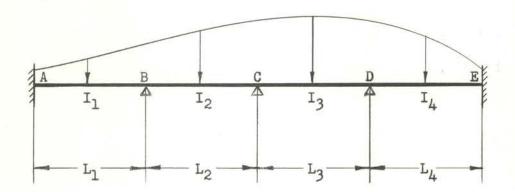


Fig. 2 FOUR SPAN BEAM

Using nomenclature similar to that in the previous chapter and summing all the terms in column $M_{\mbox{\scriptsize AB}}$ of Table 2, the moment equation becomes

$$M_{AB} = -FM_{AB} - \frac{av}{x} 2B + \frac{ac}{x}C - \frac{ace}{x}D$$
 (4)

in which

$$\frac{1}{x}$$
 = 1 + (bc + de) + (bc + de)² + (bc + de)³ + + 0, (4a)

$$x = 1 - bc - de$$
 (denominator of convergency for \overline{AE}), (4b)

$$y_1 = 1 - bc$$
 (denominator of convergency for \overline{AD}), (4c)

and

$$y_2 = 1 - de (denominator of convergency for \overline{BE}). (4d)$$

All moment equations similarly derived are presented in Section 2, Table C4.

TABLE 2

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR A FOUR SPAN BEAM

The state of the s	A		3		,)	
Distribution	0	2a	2b	2c	2d	2e	2f	0
Factors: Fixed End Moments	-FM _{AB}	FM _{BA}	-FM _{BC}	FM _{CB}	-FM _{CD}	FM _{DC}	-FM _{DE}	FM _{ED}
lst Cycle		-2aB	-2ъВ	-2cC	-2dC	-2eD	-2fD	
Carry-over	-aB		-cC	-bB	-eD	-dC		-fD
2nd Cycle		2acC	2bcC	2c5 ⁽⁴⁾	2dS	2deC	2dfC	
Carry-over	acC		cS	bcC	deC	dS		dfC
3rd Cycle		-2acS	-2bcS	-2ct/4c	-2dtC	-2deS	-2dfS	
Carry-over	-acS		-ctC	-bcS	-deS	-dtC		-dfS
4th Cycle		2actC	2bctC	2ctS	2dtS	2detC	2dftC	
Carry=over	actC		ctS	bctC	detC	dtS		dftC
5th Cycle		-2actS	-2bctS	-2ct ² C	-2dt ² C	-2detS	-2dftS	
								•
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}	M _{DE}	$M_{\rm ED}$

⁴Substituting bB + eD = S and bc + de = t simplifies the algebraic procedure. When summing the terms in each column, the values S and t are replaced by the original values.

4. INTRODUCTION TO SERIES OF SERIES

The Basic Series has proven itself adequate in the two preceding derivations. For beams with more than four spans, the series become very complex and it would be unadvisable to derive the moment equations by the previous procedure.

To illustrate the complexity of such a problem, consider an elementary case: a five span beam of equal spans, constant moment of inertia, and loaded as shown in Fig. 3.

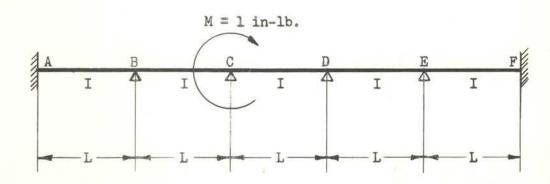


Fig. 3 FIVE SPAN BEAM - EQUAL SPANS

Denoting all the distribution factors as

$$D = 2a, (5)$$

from Table 3, the moment MAB is

$$M_{AB} = a^2 + 2a^4 + 5a^6 + 13a^8 + 24a^{10} + \dots + 0.$$
 (6)

The series forming the moment M_{AB} is evidently infinite in length and does converge, but it is not a simple geometric series. Although the algebraic terms form a perfect geometric series, their numerical factors follow a different generating law. By resolving each numerical

TABLE 3

ALGEBRAIC DISTRIBUTION OF APPLIED MOMENT (M = 1) FOR A FIVE SPAN BEAM

Points:	1		3		C]	D		3	
D's	0	2a	2a	2a	2a	2a	2a	2a	2a	0
1st Cycle				-2a	-2a					
2nd Cycle		2a ²	-a 2a ²			-a 2a ²	2a ²			
0,010	a ²		~~~	a ²	a ²	24	24	a ²		
3rd Cycle				-4a3	-4a3			-2a ³	-2a ³	
			-2a3	-		-2a3	_a3			_a3
4th Cycle		4a4	4a4			6a4	6a4			
	2a ⁴			2a4	3a4			3a4		
5th Cycle				-10a ⁵	-10a ⁵			-6a ⁵	-6a ⁵	
			-5a ⁵			-5a ⁵	-3a ⁵			-3a ⁵
6th Cycle		10a6	10a6			16a6	16a6			
	5a.6			5a6	8a6			8a6		
7th Cycle	PELLO III.			-26a ⁷	-26a ⁷			-16a ⁷	-16a ⁷	
Infinite Cycle Sum of all	0	0	0	0	0	0	0	0	0	0
the terms in each column = Final	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}	M _{DE}	M _{ED}	M _{EF}	M _{FE}

factor into basic components:

$$T_{1} = (1)a^{2} = 1a^{2},$$

$$T_{2} = (1 + 1)a^{4} = 2a^{4},$$

$$T_{3} = (1 + 2 + 2)a^{6} = 5a^{6},$$

$$T_{4} = (1 + 2 + 5 + 5)a^{8} = 13a^{8},$$

$$T_{5} = (1 + 2 + 5 + 13 + 13)a^{10} = 34a^{10},$$

$$T_{6} = (1 + 2 + 5 + 13 + 34 + 34)a^{12} = 85a^{12},$$

$$T_{n-1} = (T_{1} + T_{2} + T_{3} + \cdots + T_{n-3} + 2T_{n-2})a^{2(n-1)},$$

$$T_{n} = (T_{1} + T_{2} + T_{3} + \cdots + T_{n-2} + 2T_{n-1})a^{2n}.$$

$$(7)$$

Equation 7 demonstrates that each final moment is a combination of an infinite number of infinite series, generating parallel to zero. In order to simplify the analysis of the series in the following chapters, the investigated beams will be divided into isolated parts.

5. CARRY-OVER SERIES \propto AND β FOR A FIVE SPAN BEAM

In the following illustration of the Carry-over Series, the end moment equations will be derived for a five span cyclosymmetrical beam, loaded as shown in Fig. 4.

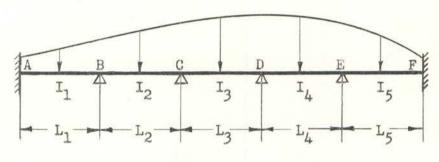
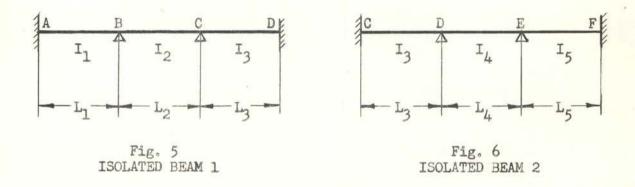


Fig. 4 FIVE SPAN BEAM



In order to simplify the algebraic procedure, only the fixed end moments at Points A and B will be considered. The same procedure must be repeated for the fixed end moments at Points C, D, E, and F, and the final moments will be the sums of the partial results.

The distribution factors are denoted by the algebraic terms shown in

TABLE 4

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR A FIVE SPAN BEAM (ISOLATED BEAM 1)

Points:	A	E	3	(3	D
Distribution	0	2a	2b	2c	2d	0
Factors: Fixed End Moments:	-FM _{AB}	FM _{BA}	-FM _{BC}			
1st Cycle		-2aB	-2bB			9.1
Carry-over	-aB			-bB		
2nd Cycle				2bcB	2bdB	
Carry-over			bcB			bdB
3rd Cycle		-2abcB	-2b ² cB			
Carry-over	-abcB			-b ² cB		
4th Cycle				2b ² c ² B	2b ² cdB	
Carry-over			b ² e ² B			b ² cdB
5th Cycle		-2ab ² c ² B	-2b ³ c ² B			
Infinite Cycle	0	0	0	0	0	0
Sum of all the terms in each column =	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}
Final Moment	-					

Table 4 and the sum of the fixed end moments at the points of support are designated by their respective symbols, as was done in Chapters Two and Three. Point D is temporarily locked against rotation; thus, isolating from a five span beam \overline{AF} , a three span beam \overline{AD} (Beam 1). From Table 4 the moments on Beam 1 are

$$M_{AB_0} = -FM_{AB} - \frac{a}{x}B_0$$
 (8)

$$M_{BA_0} = + FM_{BA} - \stackrel{2}{\sim} B, \tag{9}$$

$$M_{BC_0} = -FM_{BA} + 2\frac{a}{x} B, \tag{10}$$

$$M_{CB_D} = -2\frac{bd}{x} B, \tag{11}$$

$$M_{CD_0} = + 2\frac{bd}{x} B_{\theta} \tag{12}$$

$$M_{DC_{D}} = + \frac{bd}{x} B. \tag{13}$$

Hereafter, all series performing moments in Beam 1 will be called "Basic Series" because of their similarity with series derived in Chapters Two and Three.

By unlocking Point D and locking Point C, a new beam is isolated (Beam 2). Denoting

$$M_{DC_0} = + \frac{bd}{x} B = \Re O_9 \tag{14}$$

and summing the values in the various columns of Table 5,

$$M_{CD_3} = -\frac{9}{y} \approx_0, \tag{15}$$

$$M_{DC_1} = + \approx_0 - 2^{\underline{e}} \approx_0, \tag{16}$$

$$M_{DE_1} = - \otimes_0 + 2 \frac{g}{y} \otimes_0, \tag{17}$$

$$M_{ED_1} = 2 \frac{fh}{y} \approx 0.$$
 (18)

$$M_{EF_3} = + 2 \frac{fh}{v} e e_0$$
 (19)

$$M_{FE_1} = + \frac{fh}{y} \approx_0. \tag{20}$$

Unlocking Point C, locking Point D and denoting

$$M_{CD_{1}} = -\frac{e}{y} \approx_{0} = \beta_{0}, \tag{21}$$

the carry-over value from Table 6 is

TABLE 5

ALGEBRAIC DISTRIBUTION OF FUNCTION IN A FIVE SPAN BEAM (ISOLATED BEAM 2)

Points:	3	I)	E	G	
Distribution Factors:	0	2e	2f	2g	2h	0
Starting Moments:		ox.				
lst Cycle		-2e ≫ _o	-2f ∞			
Carry-over	-e 🗙			-f ∞₀		
2nd Cycle				2fg ⊘e	2fh җ	
Carry-over			fg 🗙 ,			fh ౫
3rd Cycle		-2efg ॐ	-2f ² g ≪			
Carry-over	-efg ∞			-f ² g ∝₅		
4th Cycle				2f ² g ²	2f ² gh≪	
Carry-over			f ² g ² ∞₀			f ² gh ≪
5th Cycle		-2ef ² g ² 0	-2f ³ g ² ∞			
Infinite Cycle	0	0	0	0	0	0
Sum of all the terms in each column =	M _{CD}	M _{DC}	M _{DE}	$^{ m M}_{ m ED}$	M _{EF}	M _{FE}

TABLE 6

ALGEBRAIC DISTRIBUTION OF FUNCTION (3) IN A FIVE SPAN BEAM (ISOLATED BEAM 1)

Points:	A	F		C		I
Distribution	0	2a	2b	2c	2d	0
Factors: Starting Moments:					Bo	
					7-5	
lst Cycle	-			-2c /3 _o	-2d/3。	
Carry-over			-c/3°			-d /3°
2nd Cycle		2ac /3,	2bc/3,			
Carry-over	ac/3o			be Bo		
3rd Cycle				-2bc ² /3 ₀	-2bcd/3	
Carry-over			-bc2Bo			-bcd /30
4th Cycle		2abc ² /3 ₈	-bc ² /3 _o 2b ² c ² /3 _o			
Carry-over	abc ² /3 _o			b2c2/3		
5th Cycle				b ² c ² /3 _e -2b ² c ³ /3 _e	-2b ² c ² d/3	
Infinite Cycle	0	0	0	0	0	0
Sum of all the	M	M	M	M	М	М
terms in each column = Final Moment	MAB	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}

TABLE 7

SERIES
AND
AS A FUNCTION OF
FOR A FIVE SPAN BEAM

Series CX - Function CX	Series /3 - Function 🗪
	$\beta_0 = -\left(\frac{\mathrm{de}}{\mathrm{xy}}\right)^0 = \infty$
	$\beta_1 = -\left(\frac{\det^2}{xy}\right)^{\frac{1}{y}} \propto$
	$\beta_2 = -\left \frac{\det}{xy}\right ^2 = \infty$
$ \propto_n = \left(\frac{\det}{xy}\right)^n \propto $	$\beta_n = -\frac{\text{de}}{\text{xy}} = \infty$
$\sum_{\alpha}^{\infty} = \sum_{\alpha}^{\infty}$	Sa = y de
$\sum_{0}^{\infty} \propto = \frac{1 - \frac{de}{xy}}{1 - \frac{de}{xy}}$	$\sum_{0}^{\infty} \beta = \frac{-\frac{e}{y}}{1 - \frac{de}{xy}}$

TABLE 8

SERIES
AND
AS A FUNCTION OF
FOR A FIVE SPAN BEAM

Series ← Function ß	Series /3 - Function /3
$\approx_0 = -\left(\frac{\text{dej}}{xy}\right)^0 \frac{\text{d}}{x} \beta$	30 = (de) 3
$ \mathcal{O}_1 = -\frac{\det^1}{\mathbf{x}\mathbf{y}} \frac{d}{\mathbf{x}} \beta $	$\beta_1 = \left(\frac{\text{de}}{xy}\right)^1 \beta$
$2 = -\left(\frac{de}{xy}\right)^2 \frac{d}{x} \beta$	$\beta_2 = \frac{ de ^2}{ xy ^2} \beta$ $\beta_n = \left(\frac{de ^n}{ xy ^n} \beta\right)$
$\propto_n = -\left(\frac{\mathrm{de}}{\mathrm{xy}}\right)^n \frac{\mathrm{d}}{\mathrm{x}} \beta$	$\beta_n = \left(\frac{de}{xy}\right)^n \beta$
$\sum_{0}^{\infty} \propto = \frac{\frac{d}{x} \beta}{1 - \frac{de}{xy}}$	$\sum_{0}^{\infty} \beta = \frac{\beta}{1 - \frac{de}{xy}}$

$$M_{DC_2} = -\frac{d}{x}/3_0 = +\frac{de}{xy} \propto_0 = \infty_1. \tag{22}$$

Systematically locking and unlocking Points C and D performs two functions:

- 1. Balancing the moments at each support,
- 2. Establishing equilibrium between the isolated beams.

The carry-over values from "Beam 1" to "Beam 2" and from "Beam 2" to "Beam 1" are denoted the functions "\(\infty \)" and "\(\beta \)" respectfully, and they are tabulated. Tables 7 and 8 demonstrate that these functions form perfect infinite, convergent, geometric series. Superimposing the Basic Series (Eq. 8) and the Carry-over Series (\(\beta \)) recently derived, the moment MAB due to the unbalance at Point B is

$$M_{AB} = -FM_{AB} - \frac{a}{x}B + \frac{ac}{x}\beta_0 + \frac{ac}{x}\beta_1 + \frac{ac}{x}\beta_2 + \dots + 0,$$
 (23a)

or

$$M_{AB} = -FM_{AB} - \frac{a}{x}B + \frac{ac}{x}\sum_{n}^{\infty} \beta , \qquad (23b)$$

in which (Table 7)

$$\sum_{0}^{\infty} /3 = \frac{-\frac{e}{y}}{1 - \frac{de}{xy}}.$$
 (23c)

With the notation 14, equation 23b becomes

$$M_{AB} = -FM_{AB} - \frac{a}{x}(1 + \frac{bc}{z}) B, \qquad (23)$$

in which

$$x = 1 - bc$$
 (Denominator of convergency for Beam 1), (23d)

and

$$z = \frac{xy}{de} - 1$$
 (Modified denominator of convergency⁵ of Carry-over
Series \propto and $/3$). (23f)

The above procedure is repeated for the fixed end moments at Points C, D, E and F. All moment equations similarly derived are presented in Section 2, Table C5.

$$\frac{1 - \frac{de}{xy}}{\frac{de}{xy}} = \frac{xy}{de} - 1 = z$$

⁵The actual denominator of convergency of the Carry-over Series 🗢

and /3 is

1 - de

1 - xv.

In order to simplify the algebraic form of the final equations, a modified term is used.

6. CARRY-OVER SERIES ∞, /3, 8 AND Y FOR A SIX SPAN BEAM

As a second illustration of the Carry-over Series, the end moment equations will be derived for a six span cyclosymmetrical beam, loaded as shown in Fig. 7. This beam could be divided into sections of unequal lengths, but the derivations of the moment equations are less complicated if the beams are isolated as shown in Figs. 8, 9, and 10.

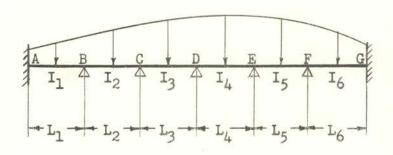
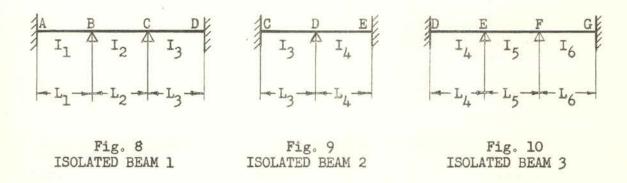


Fig. 7 SIX SPAN BEAM



Using nomenclature similar to that of the previous chapters and temporarily locking Point D, the moments on Beam 1 due to the fixed end moments

TABLE 9

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR A SIX SPAN BEAM (ISOLATED BEAM 1)

Points:	A	I	3		0	D
Distribution	0	2a	2b	2c	2d	0
Factors: Fixed End Moments:	-FM _{AB}	FM _{BA}	-FM _{BC}			
lst Cycle		-2aB	-2bB			
Carry-over	-aB			⊸bB		
2nd Cycle				2bcB	2bdB	
Carry-over			bcB			bdB
3rd Cycle		-2abcB	-2b ² cB			***************************************
Carry-over	-abcB			-b ² cB		
4th Cycle				2b ² c ² B	2b ² cdB	
Carry-over			b ² c ² B			b ² cdB
5th Cycle		-2ab ² c ² B	-2b ³ c ² B	-		
Infinite Cycle	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	M _{AB}	M _{BA}	M _{BC}	^M CB	M _{CD}	M _{DC}

at Points A and B are (Table 9)

$$M_{ABO} = -FM_{AB} - \frac{a}{x_1}B, \qquad (24)$$

$$M_{BAO} = FM_{BA} - 2\frac{a}{x_1}B, \qquad (25)$$

$$M_{BCO} = -FM_{BA} + 2\frac{a}{x_1}B, \qquad (26)$$

$$^{\mathrm{M}}_{\mathrm{CB_0}} = -2^{\mathrm{bd}}_{\mathrm{X_1}}^{\mathrm{B}}, \tag{27}$$

$$M_{CD_0} = 2\frac{bd}{x_1} B, \qquad (28)$$

$$M_{DC_0} = \frac{bd}{x_1} B. \tag{29}$$

Unlocking Point D and locking Points C and E, a second beam is isolated (Beam 2). Denoting

$$M_{DC_0} = \frac{bd}{x_1} B = \infty_0, \tag{30}$$

and summing the values in the various columns of Table 10,

$$M_{CD_1} = -e \sim_{O_2}$$
 (31)

$$M_{DC_1} = \infty_0 - 2e \infty_0, \tag{32}$$

$$M_{DE_1} = -2f \propto_0, \tag{33}$$

$$M_{ED_1} = -f \propto_0. \tag{34}$$

TABLE 10

ALGEBRAIC DISTRIBUTION OF FUNCTION CO. IN A SIX SPAN BEAM (ISOLATED BEAM 2)

Points: Distribution Factors: Starting Moments:

1st Cycle

Carry-over

Unlocking Points C and E and locking Point D, Beam 1 is again

TABLE 11

ALGEBRAIC DISTRIBUTION OF FUNCTION /3. IN A SIX SPAN BEAM (ISOLATED BEAM 1)

	A	I		(D
Distribution	0	2a	2b	2c	2d	0
Factors: Starting Moments:					0	
Dear offig nomenes.					Bo	
1-t 01-				2.0	21.0	
lst Cycle				-2c/3.	-2a/3 ₀	
027				\\		
Carry-over			-c/3.			-d/30
2nd Cycle		2ac/3	2bc/30			1
Carry-over	ac/3.			be /3.		1
	, ,			, , , , , ,		
3rd Cycle				-2bc ² /3	-2bcd/3,	
				100	700	
Carry-over			-bc ² /3.			-bed (3
oally-over			-00 /00			-bcd/3
143 0 - 2		0.1.20	2b ² c ² /3,			
4th Cycle		2abe /3	26 6 /3			
200	2			2.2		
Carry-over	abc ² /3			b ² c ² /30		
5th Cycle				-2b ² c ³ /3,	-2b2c2d/3	1
T. 61-11- 01-	_	0	0		0	
Infinite Cycle	0	0	0	0	0	0
Sum of all the						
terms in each	MAB	MBA	M_{BC}	M _{CB}	M_{CD}	M _{DC}
column =	A.D	DA	DO	ŲD.	OD	20
Final Moment		Carried Contraction				

TABLE 12

ALGEBRAIC DISTRIBUTION OF FUNCTION % IN A SIX SPAN BEAM (ISOLATED BEAM 3)

)]	7	
0	2g	2h	2j	2k	
	Y,				
	-2g Y,	-2h Y			
-g Y _o			−h Y₀		
			2hj Yo	2hk 🗸	
		hj Y.			hk 🍾
	-2ghj Y ₀	-2h ² j Y。			
-ghj ⅓			-h ² j Y _o		
			2h ² j ² / ₆	2h ² jk¥ _o	
		h ² j ² %			h²jk ⅓
	-2gh ² j ² / ₂	-2h ³ j ² ¥ _o			
0	0	0	0	0	0
M _{DE}	M _{ED}	$M_{ m EF}$	M _{FE}	$_{ m M_{FG}}$	$M_{ m GF}$
	-ghj Y _o	0 2g	0 2g 2h Y _o -2g Y _o -2h Y _o hj Y _o -2ghj Y _o -2h ² j Y _o h ² j ² Y _o -2gh ² j ² Y _o -2h ³ j ² Y _o 0 0 0	0 2g 2h 2j Y ₀ -2g Y ₀ -2h Y ₀ -h Y ₀ 2hj Y ₀ -h Y ₀ 2hj Y ₀ -h Y ₀ 2hj Y ₀ 2hj Y ₀ -h Y ₀ 2hj Y ₀ 2hj Y ₀ 2hj Y ₀ -2ghj Y ₀ -2h ² j Y ₀ 2h ² j Y ₀ -h ² j Y ₀ 2h ² j Y ₀ 0 0 0 0	0 2g 2h 2j 2k Y _o -2g Y _o -2h Y _o -h Y _o 2hj Y _o 2hk Y _o hj Y _o -2ghj Y _o -2h ² j Y _o -h ² j Y _o 2h ² j ² Y _o 2h ² jk Y _o 0 0 0 0 0

isolated, but also isolated is Beam 3. The following moments are denoted

$$M_{CD_{1}} = -e \propto_{0} = \beta_{0}, \tag{35}$$

$$M_{ED_{1}} = -f \propto_{0} = \gamma_{0}, \tag{36}$$

and the carry-over values from Tables 11 and 12 are

$$M_{DC_2} = -\frac{d}{x_1} \beta_0 = \infty_1, \tag{37}$$

$$M_{DE_2} = -\frac{g}{x_2} Y_0 = S_0. \tag{38}$$

TABLE 13

ALGEBRAIC DISTRIBUTION OF FUNCTION % IN A SIX SPAN BEAM (ISOLATED BEAM 2)

Points: Distribution Factors: Starting Moments:

1st Cycle

Carry-over

0	2e	2f	0
		8.	
	-2e %.	-2f &	
-е б.			-f &

The Points C, D, and E are locked and unlocked systematically and the values of \propto ,/3, δ and δ are tabulated (Table 14). Superimposing the Basic Series (Eq. 24) and the Carry-over Series (/3) recently derived, the moment M_{AB} due to the fixed end moments at Points A and B is

$$M_{AB} = -FM_{AB} - \frac{a}{x_1}B + \frac{ac}{x_1}\beta_0 + \frac{ac}{x_1}\beta_1 + \frac{ac}{x_1}\beta_2 + \dots + 0, \quad (39a)$$

or

$$M_{AB} = -FM_{AB} - \frac{a}{x_1}B + \frac{ac}{x_1}\sum_{n=0}^{\infty}\beta, \qquad (39b)$$

in which (Table 14)

$$\sum_{\alpha=0}^{\infty} \beta = -\frac{e}{(1-T)} \propto = -\frac{x_1}{dz_1} \propto.$$
 (39c)

TABLE 14 SERIES \varpropto , \nearrow , \lozenge AND \curlyvee AS A FUNCTION OF \varpropto FOR A SIX SPAN BEAM

Series ∝ - Function ∝	Series /3 → Function ∝
	Bosee
$\propto_1 = \frac{\text{de}}{x_1} \propto$	B ₁ = - e T \propto
	$\beta_2 = e T^2 \propto$
$ \propto_{n} = \frac{\text{de}}{x_{1}} \text{ Tn-l} \propto $	$\beta_n = -e^{n} \sim$
$\sum_{o}^{\infty} = \frac{\infty}{(1-T)}(1-\frac{fg}{x_2})$	$\sum_{\bullet}^{39} \beta = \frac{e \propto}{(1 - T)}$
	×
Series S - Function ∞	Series Y - Function <
Series S - Function ∞ $S_0 = \frac{gf}{x_2} \propto$	Series Y - Function \propto $Y_0 = -f \propto$
	Series Y - Function \propto $Y_0 = -f \propto$ $Y_1 = -f T \propto$
$\delta_0 = \frac{gf}{x_2} \propto$	Y ₀ = - f ∝
$\delta_0 = \frac{gf}{x_2} \propto$ $\delta_1 = \frac{gf}{x_2} T \propto$	$Y_0 = -f \propto$ $Y_1 = -f \propto$

⁶Substituting $\frac{de}{x_1} + \frac{fg}{x_2} = T$.

TABLE 15

SERIES \sim , \sim , \sim AND \sim AS A FUNCTION OF \sim FOR A SIX SPAN BEAM

Series ∝ - Function /3	Series /3 - Function /3
$\propto_0 = -\frac{d}{x_1} / 3$	B0=B
$\propto_1 = -\frac{d^2e}{x_1} 2/3$	$\beta_1 = \frac{\text{de}}{x_1} \beta$
$\simeq_2 = -\frac{d^2e}{x_1} 2T/3$	$\beta_2 = \frac{\text{de}}{x_1} T/3$
$ \approx_{n} = -\frac{d^{2}e}{x_{1}} 2^{n-1} / 3 $	$\beta_n = \frac{\text{de}}{x_1} T^{n-1} / 3$
$\sum_{0}^{\infty} = \frac{d\beta}{(1-T)x_1}(1-\frac{fg}{x_2})$	$\sum_{\delta} \beta = \frac{\beta}{(1-T)} (1 - \frac{fg}{x_2})$
Series 8 - Function /3	Series Y = Function /3
Series δ = Function β $\delta_0 = -\frac{\text{dfg}}{x_1 x_2} \beta$	Series $Y = Function /3$ $Y_0 = \frac{df}{x_1} /3$
$\delta_0 = -\frac{\mathrm{dfg}}{\mathrm{x}_1 \mathrm{x}_2} / 3$	$\gamma_0 = \frac{\mathrm{df}}{x_1} / 3$
$\delta_0 = -\frac{\text{dfg}}{x_1 x_2} / 3$ $\delta_1 = -\frac{\text{dfg}}{x_1 x_2} T / 3$	$Y_0 = \frac{df}{x_1} / 3$ $Y_1 = \frac{df}{x_1} T / 3$

TABLE 16 SERIES \leadsto , /3 , 8 AND Y AS A FUNCTION OF 8 FOR A SIX SPAN BEAM

Series ≪ - Function 8	Series /3 - Function 8
	/3 ₀ = - e 8
$\approx_1 = \frac{\text{de}}{x_1} T \delta$	$\beta_1 = -eT8$
$ \propto_2 = \frac{\text{de}}{x_1} T^2 S $	/32 = - e T ² 8
$ \sim_n = \frac{\text{de}}{x_1} \text{T}^n \delta $	$\beta_n = -e T^n S$
$\sum_{\infty}^{\infty} = \frac{\text{de } S}{(1-T)x_1}$	$\sum_{0}^{\infty} \beta = \frac{-eS}{(1-T)}$
Series 8 - Function 8	Series Y - Function 8
Series 8 - Function 8 $\delta_0 = 8$	Series γ - Function δ $\gamma_0 = - f \delta$
2 12	
8 ₀ = 8	Y ₀ = - f 8
$S_1 = \frac{fg}{x_2} S$	Y ₀ = - f 8 Y ₁ = - f T 8

TABLE 17 $\text{SERIES} \leadsto, \ \, /\! 3 \,\,, \,\, \otimes \,\, \text{ AND } \,\, \forall \,\, \text{ AS A FUNCTION OF } \,\, \forall \,\, \text{FOR A SIX SPAN BEAM }$

Series ≪ - Function Y	Series /3 - Function Y
$\approx_0 = -\frac{\deg}{x_1 x_2} \gamma$	$\beta_0 = \frac{eg}{x_2} Y$
$\approx_1 = -\frac{\deg}{x_1} x_2^T Y$	$\beta_1 = \frac{eg}{x_2} T Y$
$\alpha_2 = -\frac{\deg}{x_1 x_2} T^2 Y$	$\beta_2 = \frac{eg}{x_2} T^2 Y$
	$\beta_n = \frac{eg}{x_2} T^n Y$
$\sum_{0}^{\infty} \propto = -\frac{\deg Y}{(1-T)x_1x_2}$	$\sum_{\infty}^{\infty} \beta = \frac{\text{ge } Y}{(1-T)x_2}$
Series 8 - Function Y	Series Y - Function Y
$S_0 = -\frac{g}{x_2} \forall$	YO = Y
$\delta_1 = -\frac{fg^2}{x_2^2} Y$	$Y_1 = \frac{fg}{x_2} Y$
$\delta_2 = -\frac{fg^2}{x_2^2} T Y$	$Y_2 = \frac{fg}{x_2} T Y$
$S_n = -\frac{fg^2}{x_2^2} T^{n-1} Y$	$\gamma_n = \frac{fg}{x_2} T^{n-1} \gamma$
$\sum_{0}^{\infty} \delta = \frac{-gY}{(1-T)x_2} (1 - \frac{de}{x_1})$	$\sum_{o}^{\infty} Y = \frac{Y}{(1-T)}(1-\frac{de}{x_{1}})$

With notation 30, equation 39b becomes

$$M_{AB} = -FM_{AB} - \frac{a}{x_1}(1 + \frac{bc}{z_1}) B,$$
 (39)

in which

$$x_1 = 1 - bc$$
 (Denominator of convergency of Beam 1), (39d)

$$x_2 = 1 - hj$$
 (Denominator of convergency of Beam 3), (39e)

$$z_1 = \frac{x}{de^2} - \frac{fgx_1}{dex_2} - 1$$
 (Modified denominator of convergency of

Carry-over Series
$$\propto$$
 , β , δ and γ . (39f)

The above procedure is repeated for the fixed end moments at Points C, D, E, F and G. All moment equations similarly derived are presented in Section 2, Table C6.

/3, δ and Y is $1 - \frac{de}{dt} - \frac{fg}{ft}$.

In order to simplify the algebraic form of the final equations, modified

$$\frac{1 - \frac{de}{x_1} - \frac{fg}{x_2}}{\frac{de}{x_1}} = \frac{x}{de} - 1 - \frac{fgx}{dex_2} = z_1,$$

$$\frac{1 - \frac{de}{x_1} - \frac{fg}{x_2}}{\frac{fg}{x_2}} = \frac{x_2 - \frac{dex}{fgx_1^2} - 1 = z_2.$$

 $^{^{7}}$ The actual denominator of convergency of the Carry-over Series \propto ,

7. CARRY-OVER SERIES AND A FOR A SEVEN SPAN BEAM

The derivation of the end moment equations for the seven span cyclosymmetrical beam are considerably like those which are presented in Chapter Five. The beam will be loaded and divided as shown in Figs. 11, 12, and 13.

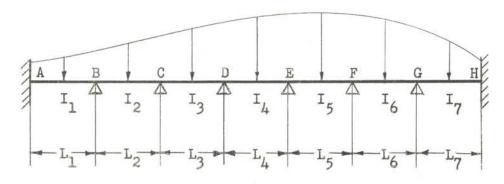
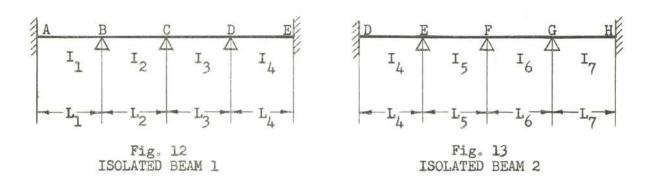


Fig. 11 SEVEN SPAN BEAM



Tables 18, 19, 20, 21, and 22 are formed by the same procedure that was used in Chapter Five.

TABLE 18

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR A SEVEN SPAN BEAM (ISOLATED BEAM 1)

Points:	A		В	(SECTION AND DESCRIPTION OF THE PARTY OF THE)	
Distribution Factors:	0	2a	2b	20	2d	2e	2f	0
Fixed End Moments:	-FM _{AB}	FMBA	-FM _{BC}					
lst Cycle		-2aB	-2bB					
Carry-over	-aB			-bB				
2nd Cycle				2bcB	2bdB			
Carry-over			beB			bdB		
3rd Cycle		-2abcB	-2b ² cB			-2bdeB	-2bdfB	
Carry-over	-abcB			-b ² cB	-bdeB			-bdfB
4th Cycle(8)				2bctB	2bdtB			
Carry-over			bctB			bdtB		
5th Cycle		-2ab ctB	-2b ² ctB			-2bd etB	-2bd ftB	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	MDC	MDE	M _{ED}

⁸Substituting bc + de = t.

TABLE 19

ALGEBRAIC DISTRIBUTION OF FUNCTION COIN A SEVEN SPAN BEAM (ISOLATED BEAM 2)

Points:	D	1	3	1	₹	(H
Distribution Factors:	0	2g	2h	2j	2k	2m	2n	0
Starting Moments:		∝.						
1st Cycle		=2go.	-2h ∝					
Carry-over	_g cx_			-h≪₀				
2nd Cycle				2hj≪₀	2hk≪₀			
Carry-over			hj∝.			hk≪₀		
3rd Cycle		-2ghj~	-2h ² j≪			-2hkm∽	-2hkn≪	
Carry-over	-ghj≪			-h ² j≪	-hkm≪			-hkn≪。
4th Cycle (9)				2hjv≪。	2hkv∽₀			
Carry-over			hjv∝.			hkv∝.		
5th Cycle		-2gh jv≪	-2h ² jv≪			-2hk lv%	-2hk mv∝	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	MDE	M _{ED}	^M EF	$^{ m M}_{ m FE}$	$^{ m M}_{ m FG}$	^M GF	M _{GH}	M _{HG}

 $⁹_{\text{Substituting hj} + \text{km} = \text{v}}$.

TABLE 20

ALGEBRAIC DISTRIBUTION OF FUNCTION /3, IN A SEVEN SPAN BEAM (ISOLATED BEAM 1)

	A		B		C	I		
Distribution Factors:	0	2a	2b	2c	2d	2e	2f	0
Starting Moments:							Bo	
1st Cycle						-2e/3 _o	-2f/3°	
Carry-over					-e/3°			-f/3 ₀
2nd Cycle				2ec/3,	2de/3			
Carry-over			ce/3 _o			de/3°		
3rd Cycle		-2ace/g	-2bce/3			-2de ² /3	-2def/3	
Carry-over	-ace/3			-bce/3	-de ² /3°			-def/3,
4th Cycle				2cet/3	2det/3			
Carry-over			cet/3,			det/3		
5th Cycle		=2ac et/3	-2bc et/3			-2de ²	-2de ft/3	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column =	M _{AB}	M _{BA}	M _{BC}	M _{CB}	McD	M _{DC}	MDE	$M_{ m ED}$

TABLE 21 SERIES \propto AND /3 AS A FUNCTION OF \propto FOR A SEVEN SPAN BEAM

Series ≪ - Function ≪	Series /3 = Function ∝
$ \approx_0 = \left(\frac{fgy_1y_4}{x_1x_2}\right)^0 \approx$	$\beta_0 = -\left(\frac{fgy_1y_1}{x_1x_2}\right)^0 \frac{gy_4}{x_2} \sim$
	$\beta_1 = -\left(\frac{fgy_1y_1}{x_1x_2}\right)^1 \frac{gy_1}{x_2} \propto$
	$\beta_2 = -\left(\frac{fgy_1y_4}{x_1x_2}\right)^2 \frac{gy_4}{x_2} \propto$
$ \sim_{n} = \left(\frac{fgy_1y_4}{x_1x_2}\right)^n \sim $	$\beta_n = -\left(\frac{fgy_1y_4}{x_1x_2}\right)^n \frac{gy_4}{x_2} \propto$
$\sum_{0}^{\infty} \propto = \frac{1 - \frac{fgy_1y_4}{x_1x_2}}$	$\sum_{0}^{\infty} \beta = -\frac{\frac{gy_{4}}{x_{2}} \propto}{1 - \frac{fgy_{1}y_{4}}{x_{1}x_{2}}}$

TABLE 22 SERIES \simeq AND \nearrow 3 AS A FUNCTION OF \nearrow 3 FOR A SEVEN SPAN BEAM

Series ∝ - Function /3	Series /3 - Function /3
$\approx_0 = -\left(\frac{fgy_1y_4}{x_1x_2}\right)^0 \frac{fy_1}{x_1} /3$	$\beta_0 = \left(\frac{fgy_1y_4}{x_1x_2}\right)^0 \beta$
$\approx_1 = -\left(\frac{fgy_1y_1}{x_1x_2}\right)^1 \frac{fy_1}{x_1} $	$\beta_1 = \left(\frac{fgy_1y_1}{x_1x_2}\right)^1 \beta$
$\approx_2 = -\left(\frac{fgy_1y_1}{x_1x_2}\right)^2 \frac{fy_1}{x_1}$	$\beta_2 = \left(\frac{fgy_1y_1}{x_1x_2}\right)^2 \beta$
	$\beta_n = \left(\frac{fgy_1y_4}{x_1x_2}\right)^n \beta$
$\sum_{0}^{\infty} = \frac{\frac{fy_1}{x_1} / 3}{1 - \frac{fgy_1y_4}{x_1x_2}}$	$\sum_{0}^{\infty} \beta = \frac{\beta}{1 - \frac{fgy_1y_4}{x_1x_2}}$

Combining the terms of Tables 18, 20 and 21, the moment MAB is

$$M_{AB} = -FM_{AB} - \frac{a}{x_1} (y_2 + \frac{bcde}{y_1 z}), \qquad (40)$$

in which

$$x_1 = 1 - bc - de,$$
 (40a)

$$x_2 = 1 - hj - km,$$
 (40b)

$$y_1 = 1 - bc, \tag{40c}$$

$$y_2 = 1 - de,$$
 (40d)

$$y_3 = 1 - hj,$$
 (40e)

$$y_{L} = 1 - km, \tag{40f}$$

and

$$z = \frac{x_1x_2}{fgy_1y_4} - 1$$
 (Modified denominator of convergency of Carry-over Series \propto and (3)). (40g)

All moment equations similarly derived are presented in Section 2, Table C7.

$$1 - \frac{fgy_1y_4}{x_1x_2}.$$

In order to simplify the final equation, a modified term is used.

$$\frac{1 - \frac{fgy_1y_4}{x_1x_2}}{\frac{fgy_1y_4}{x_1x_2}} = \frac{x_1x_2}{fgy_1y_4} - 1 = z.$$

 $^{^{10}}$ The actual denominator of convergency of the Carry-over Series \simeq and \nearrow 3 is

8. CARRY-OVER SERIES <, /3, 8 AND Y FOR AN EIGHT SPAN BEAM

As a final illustration of the Carry-over Series, the end moment equations will be derived for an eight span cyclosymmetrical beam, loaded as shown in Fig. 14.

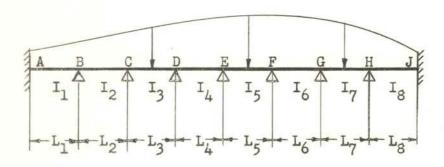
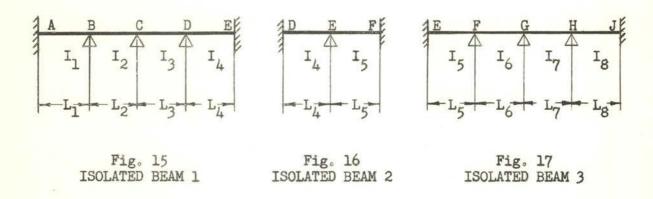


Fig. 14 EIGHT SPAN BEAM



Since the series would become complicated if beams of more than four spans were isolated, it is necessary to divide the eight span beam into two four span sections and a two span section, as shown in Figs. 15, 16 and 17.

TABLE 23

ALGEBRAIC DISTRIBUTION OF FIXED END MOMENTS FOR AN EIGHT SPAN BEAM (ISOLATED BEAM 1)

Points:	A	1	3		C	I)	
Distribution	0	2a	2b	2c	2d	2e	2f	0
Factors: Fixed End Moments:	-FM _{AB}	FM _{BA}	-FM _{BC}					
lst Cycle		-2aB	-2bB					
Carry-over	-aB			-bB				
2nd Cycle				2beB	2bdB			
Carrymover			bcB			bdB		
3rd Cycle		-2abcB	-2b ² cB		N.	-2bdeB	-2bdfB	
Carry-over	-abcB			-b ² cB	-bdeB			-bdfB
4th Cycle(11)				2bctB	2bdtB			F 5
Carry-over			bctB			bdtB		
5th Cycle		-2ab ctB	-2b ² etB			-2bd etB	-2bd ftB	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	M _{AB}	M _{BA}	M _{BC}	M _{CB}	MCD	M _{DC}	MDE	M _{ED}

llSubstituting bc + de = t.

Using nomenclature similar to that of the previous chapters and temporarily locking Point E, the moments in Beam 1 due to the fixed end moments at Points A and B are (Table 23)

$$M_{AB_0} = -FM_{AB} - \frac{ay_2}{x_1}B, \qquad (41)$$

$$M_{\text{BAO}} = FM_{\text{BA}} - 2\frac{\text{ay}2}{\text{x}_1}B, \tag{42}$$

$$M_{BCO} = -FM_{BA} + 2\frac{ay}{x_1} + 2\frac{ay}{x_1} + 2\frac{ay}{x_1}$$
 (43)

$$M_{CBO} = -\frac{bdu}{x_1} 2_B, \tag{44}$$

$$M_{CDO} = \frac{bdu}{x_1} 2_B,$$
 (45)
 $M_{DCO} = 2 \frac{bdf}{x_1} B,$ (46)
 $M_{DEO} = -2 \frac{bdf}{x_1} B,$ (47)

$$M_{DCO} = 2\frac{\text{bdf}}{x_1} B, \tag{46}$$

$$M_{DEO} = -2 \frac{\text{bdf}}{x_1} \text{ B}, \tag{47}$$

$$M_{EDO} = -\frac{bdf}{x_1} B. \tag{48}$$

Unlocking Point E and locking Points D and F, a second beam is isolated (Beam 2). Denoting

$$M_{EDO} = -\frac{bdf}{x_1} B = \infty_0, \tag{49}$$

and summing the values in the various columns of Table 24

$$M_{DE_1} = g^{\infty}_0, \tag{50}$$

$$M_{ED1} = 2h \sim_{O}, \tag{51}$$

$$M_{EF_1} = -2h \sim_{O_2} \tag{52}$$

$$M_{FE_1} = -h \propto_0. \tag{53}$$

TABLE 24

ALGEBRAIC DISTRIBUTION OF FUNCTION CIN AN EIGHT SPAN BEAM (ISOLATED BEAM 2)

Points: Distribution Factors: Starting Moments:

1st Cycle

Carry-over

0	2g ~	2h	0
	-2g ∝	-2h ∞₀	
-g ∞ _o			-h ∝

TABLE 25

ALGEBRAIC DISTRIBUTION OF FUNCTION /3, IN AN EIGHT SPAN BEAM (ISOLATED BEAM 1)

Points:	A	I			3	I		
Distribution	0	2a	2b	2c	2d	2e	2f	0
Factors: Starting							0	
Moments:							Bo	
lst Cycle						-2e /3°	-2f/3 _o	
Carry-over					-е/3。			-f/3°
2nd Cycle				2ce/3。	2de/3。			
Carry-over			ce/3。			de/3a		
3rd Cycle		-2ace/3	-2bce/3			-2de ² /3	-2def/3	
Carry-over	-ace/3			-bce/3	$-de^2/3$			-def/3
4th Cycle				2cet/3	2det/3			
Carry-over			cet/3			det/3		
5th Cycle		-2ac et/3	-2bc et/3			-2de ²	-2de ft/3	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	MAB	M _{BA}	M _{BC}	M _{CB}	M _{CD}	MDC	M _{DE}	M _{ED}

TABLE 26

ALGEBRAIC DISTRIBUTION OF FUNCTION % IN AN EIGHT SPAN BEAM (ISOLATED BEAM 3)

Points:	E		F	. (3	I	H	J
Distribution Factors: Starting	0	2j %.	2k	2m	2n	2p	2r	0
Moments:			-2k %					
Carry-over	-j ŏ.			-k 80				
2nd Cycle				2 km %.	2kn 🍇			
Carry-over			km %			kn Yo		
3rd Cycle		-2jkm Y	-2k ² m y			-2knp y	-2knr%	
Carry-over	-jkm %			-k ² m Y,	−knp ‰			-knr Y
4th Cycle(12)				2kmw %	2knw Y			
Carry-over			kmw %			knw Yo		
5th Cycle		-2jk mwy	-2k ²			-2kn pw%	-2kn rw %	
Infinite Cycle	0	0	0	0	0	0	0	0
Sum of all the terms in each column = Final Moment	M _{EF}	M _{FE}	M _{FG}	M _{GF}	M _{GH}	M _{HG}	M _H J	МЈН

 $^{^{12}}$ Substituting km + np = w.

Unlocking Points D and F and locking Point E, Beam 1 is again isolated, but also isolated is Beam 3. The following moments are denoted

$$M_{DE_1} = -g \propto_0 = /3_0,$$
 (54)

and

$$M_{FE_1} = -h \approx_0 = \chi_0. \tag{55}$$

The carry-over values from Tables 25 and 26 are

$$M_{ED_2} = -\frac{fy_1}{x_1} \beta_0 = \alpha_1,$$
 (56)

$$M_{EF2} = -\frac{Jy}{x^2} 4 y_0 = 80. \tag{57}$$

TABLE 27

ALGEBRAIC DISTRIBUTION OF FUNCTION & IN AN EIGHT SPAN BEAM (ISOLATED BEAM 2)

Points: Distribution Factors: Starting Moments:

1st Cycle

Carry-over

0	2g	2h	0
		8.	
	-2g &	-2h S.	
-g 8.			-h 8,

The Points D, E and F are locked and unlocked systematically and the values of \sim , /3, % and Y are tabulated (Tables 28, 29, 30 and 31). Superimposing the Basic Series (Eq. 41) and the Carry-over Series recently derived, the moment M due to the fixed end moments at Points A and B is

$$M_{AB} = -FM_{AB} - \frac{ay}{x_1} 2_B - \frac{ace}{x_1} \beta_0 - \frac{ace}{x_1} \beta_1 - \frac{ace}{x_1} \beta_2 - \dots - 0$$
 (58a)

or

$$M_{AB} = -FM_{AB} - \frac{ay_2}{x_1}B - \frac{ace}{x_1}\sum_{n=1}^{\infty}\beta, \qquad (58b)$$

in which (Table 28)

$$\sum_{0}^{6} \beta = \sum_{1 = 6}^{6} \frac{g}{x_1} + \sum_{1 = 6}^{6} \frac{g}{x_2}$$
(58c)

TABLE 28

SERIES ∞, /3, % AND Y AS A FUNCTION OF∞FOR AN EIGHT SPAN BEAM

Series ≪ - Function ≪	Series /3 - Function ≪
∞0 = ∞	/30 = - g ∝
	/3 ₁ = - g v ∝
$\approx_2 * \frac{fgy}{x_1} \lor \approx (13)$	$\beta_2 = -g V^2 \propto$
$ \propto_n = \frac{fgy}{x_1} v^{n-1} \approx$	$\beta_n = -g v^n \propto$
$\sum_{0}^{\infty} \propto = \frac{\infty}{1 - V} \left(1 - \frac{\text{hjy}}{x_2}\right)$	$\sum_{o}^{\infty} \beta = \frac{-g^{\infty}}{1-V}$
Series S - Function ∞	Series Y - Function ∝
$\delta_0 = \frac{hjy_4}{x_2} \sim$	$y_0 = -h \propto$
$S_1 = \frac{hjy_4}{x_2} \forall \infty$	Yl = - h voc
$S_2 = \frac{hjy_4}{x_2} v^2 \approx$	$Y_2 = -h V^2 \approx$
$\delta_n = \frac{hjy_4}{x_2} v^n >$	$\forall_n = -h \ V^n \infty$
$\sum_{0}^{\infty} 8 = \frac{\infty}{1 - V} \frac{hjy}{x_2}$	$\sum_{0}^{\infty} Y = \frac{-h \propto}{1 - V}$

^{13&}lt;sub>Substituting</sub> $\frac{fgy}{x_1} + \frac{hjy}{x_2} = V$.

TABLE 29 SERIES \propto , /3 , 8 AND Y AS A FUNCTION OF /3 FOR AN EIGHT SPAN BEAM

Series ∝ - Function /3	Series /3 - Function /3
	/3 ₀ = /3
$\alpha_1 = -\frac{f^2gy_1^2}{x_1^2}$	$\beta_1 = \frac{fgy_1}{x_1} \beta$
$\alpha_2 = -\frac{f^2gv_1^2}{x_1^2} \sqrt{3}$	$\beta_2 = \frac{fgy_1}{x_1} \sqrt{3}$
	$\beta_n = \frac{fgy_1}{x_1} \sqrt{n-1}/3$
$\sum_{x_1(1-v)}^{\infty} \left(1 - \frac{hjy_1}{x_2}\right)$	$\sum_{\alpha=1}^{\infty} \beta = \frac{\beta}{1-V} \left(1 - \frac{\text{hjy}_{1}}{x_{2}}\right)$
Series 8 - Function /3	Series Y - Function /3
Series 8 - Function /3 $8_0 = -\frac{\text{fhjy_1y_4}}{x_1 x_2} / 3$	Series Y - Function β $Y_0 = \frac{fhy_1}{x_1} \beta$
$\delta_0 = -\frac{\text{fhjy_1y_4}}{x_1 x_2} / 3$	$Y_0 = \frac{\text{fhy}_1}{x_1} / 3$
$8_0 = -\frac{\text{fhjy}_1 y_4}{x_1 x_2} / 3$ $8_1 = -\frac{\text{fhjy}_1 y_4}{x_1 x_2} \vee / 3$	$Y_0 = \frac{fhy_1}{x_1} / 3$ $Y_1 = \frac{fhy_1}{x_1} \vee / 3$

TABLE 30 SERIES \leadsto , \curvearrowright , $\mathrel{\$}$ AND $\mathrel{\forall}$ AS A FUNCTION OF $\mathrel{\$}$ FOR AN EIGHT SPAN BEAM

Series ∝ - Function 8	Series /3 - Function 8
	130 = - g 8
	/31 = - g V 8
$ \propto_2 = \frac{\text{fgy}_1}{x_1} v^2 S $	/3 ₂ = - g V ² 8
	/3 _n = − g V ⁿ S
$\sum_{\infty}^{\infty} \propto = \frac{fgy_1 \S}{x_1(1 - V)}$	$\sum_{s}^{\infty} \beta = \frac{-g \delta}{1 - V}$
Series 8 - Function 8	Series Y - Function 8
80 = 8	γ ₀ = - h δ
$\delta_1 = \frac{hjy_4}{x_2} \delta$	Y ₁ = - h V 8
$S_2 = \frac{hjy_4}{x_2} \vee S$	$Y_2 = - h V^2 S$
$\delta_n = \frac{hjy_4}{x_2} v^{n-1} \delta$	$Y_n = -h V^n S$
$\sum_{0}^{\infty} S = \frac{S}{1 - V} \left(1 - \frac{fgy_1}{x_1}\right)$	$\sum_{0}^{\infty} Y = \frac{-h \mathcal{E}}{1 - V}$

TABLE 31 SERIES \leadsto , /3 , 6 AND Y AS A FUNCTION OF Y FOR AN EIGHT SPAN BEAM

Series ✓ - Function Y	Series /3 - Function Y
$\sim_0 = -\frac{fgjy}{x_1x_2} \frac{y}{4} $	$\beta_0 = \frac{gjy_4}{x_2} \forall$
	$\beta_1 = \frac{gjy_4}{x_2} \vee \vee$
$\alpha_2 = -\frac{fgjy_1y_4}{x_1x_2} v^2 v$	$\beta_2 = \frac{gjy_4}{x_2} v^2 v$
$\alpha_{n} = -\frac{fg_{3}y_{1}y_{4}}{x_{1}x_{2}}v^{n} v$	$\beta_n = \frac{gjy_4}{x_2} v^n y$
$\sum_{0}^{\infty} \propto = -\frac{\text{fgjy}_{1}y_{4}}{x_{1}x_{2}(1-V)}$	$\sum_{i=1}^{\infty} \beta = \frac{gjy_4 \ \forall}{x_2(1-V)}$

Series 8 - Function Y	Series Y - Function Y
$S_0 = -\frac{Jy}{x}$	Y = Y
$S_1 = -\frac{hj^2y_4^2}{x_2^2} Y$	$Y_1 = \frac{hjy_4}{x_2} Y$
$S_2 = -\frac{hj^2y_1^2}{x_2^2} V Y$	$Y_2 = \frac{hjy_4}{x_2} \nabla Y$
$S_n = -\frac{hj^2y_4^2}{x_2^2} v^{n-1} v$	$Y_n = \frac{hjy_4}{x_2} v^{n-1} Y$
$\sum_{\alpha}^{\infty} \delta = \frac{-jy_4 \ \%}{x_2(1-\mathbb{V})} (1-\frac{fgy}{x_1})$	$\sum_{0}^{\infty} Y = \frac{Y}{1 - V} \left(1 - \frac{fgy}{x_1}\right)$

With notation 49, equation 58b becomes

$$M_{AB} = -FM_{AB} - \frac{a}{x_1}(y_2 + \frac{bcde}{y_1 z_1}) B,$$
 (58)

in which

$$x_1 = 1 - bc - de,$$
 (58d)

$$x_2 = 1 - km - np,$$
 (58e)

$$y_1 = 1 - bc,$$
 (58f)

$$y_2 = 1 - de, \tag{58g}$$

$$y_3 = 1 - km, \tag{58h}$$

$$y_{L} = 1 - np, \tag{58j}$$

and

$$z_1 = \frac{x_1}{fgy_1} - \frac{hjy_4x_1}{fgx_2y_1} - 1$$
 (Modified denominator of convergency) of Carry-over Series \propto , $(3, 8)$ and $(58k)$

The above procedure is repeated for the fixed end moments at Points C, D, E, F, G, H and J. All moment equations similarly derived are presented in Section 2, Table C8.

$$1 - \frac{fgy_1}{x_1} - \frac{hjy_4}{x_2}.$$

In order to simplify the algebraic form of the final equations, modified terms are used.

$$\frac{1 - \frac{fgy_1}{x_1} - \frac{hjy_4}{x_2}}{\frac{fgy_1}{x_1}} = \frac{x_1}{fgy_1} - 1 - \frac{hjy_4x_1}{fgy_1x_2} = z_1,$$

$$\frac{1 - \frac{fgy_1}{x_1} - \frac{hjy_4}{x_2}}{\frac{hjy_4}{x_2}} = \frac{x_2}{hjy_4} - 1 - \frac{fgy_1x_2}{hjy_4x_1} = z_2.$$

 $^{^{14}}$ The actual denominator of convergency of the Carry-over Series $^{\bowtie}$, $^{\bowtie}$ and $^{\vee}$ is

9. FINAL CONCLUSIONS

A general procedure for the derivation of the end moments by moment distribution has been demonstrated by using the summations of infinite, convergent, geometric series. The results are exact, direct and general. The final moments are functions of two separate and independent variables:

- The New Distribution Factor which is a function of the stiffness factors and the carry-over factors,
- 2. The Fixed End Moment which is a function of the applied loads or the performed deformation.

In general, the New Distribution Factor consists of two classes of infinite series:

- 1. Basic Series,
- 2. Carry-over Series.

The Basic Series are of two types:

- 1. Three Span Basic Series,
- 2. Four Span Basic Series.

The Carry-over Series are of two types:

- 1. Carry-over Series from left to right,
- 2. Carry-over Series from right to left.

The Basic Series consists of an infinite number of terms converging to zero and each term is a finite number.

Basic Series = $Ar^0 + Ar^1 + Ar^2 + Ar^3 + \dots + 0$. (59a) The Carry-over Series also consists of an infinite number of terms converging to zero. Each term in this series is not finite; it is another infinite series.

The sum of the Basic Series is the simple sum of an infinite, convergent, geometric series.

$$\sum_{n=0}^{\infty} A = \frac{A}{1-r}.$$
(59)

The sum of the Carry-over Series is the sum of an infinite series of infinite series.

$$\sum_{s}^{\infty} C = \frac{C}{1-s} r^{0} + \frac{C}{1-s} r^{1} + \frac{C}{1-s} r^{2} + \dots + 0, \quad (60b)$$

$$= \frac{C}{(1-s)(1-r)}. \quad (60)$$

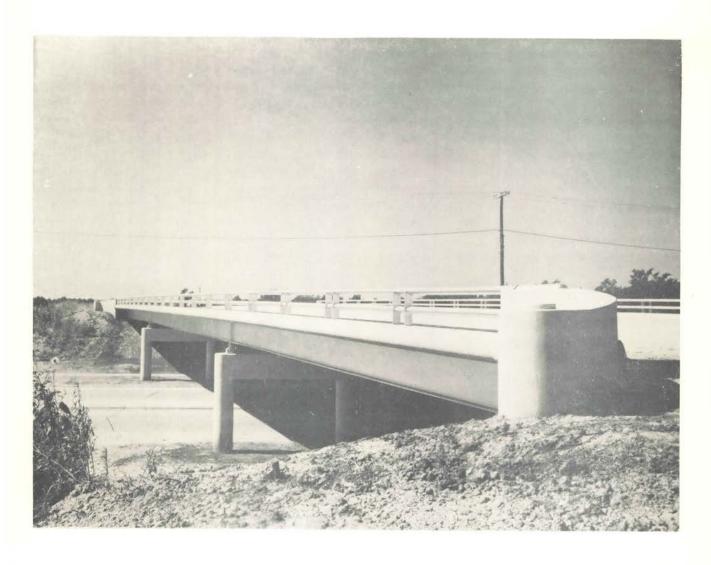
According to the performed investigations, the continuous beams may be divided into two groups:

- Three and four span beams containing New Distribution Factors formed by the Basic Series;
- 2. Five and more span beams containing New Distribution Factors formed by the Basic Series and the Carry-over Series.

Although the investigation was limited to continuous beams of three, four, five, six, seven and eight spans having a constant moment of inertia, it is evident that the principles are valid for all cases in which the Fixed End Moments may be distributed or the end angle changes may be bal-

anced. Any beam may be divided into an integral number of two, three and four span sections and the Carry-over Series tabulated. In many cases it becomes necessary to decide which division is the most appropriate. If the isolated sections are symmetrical with respect to the center of the beam, the derivations are simpler and the resulting equations are more concise than if the isolated sections are unbalanced. It must be noted that no matter how the beam is divided, the results will always be the same, although the algebraic form may differ.

Section 2



(Courtesy of Texas Highway Department)

California Street Overpass on U.S. Highway 77 in Gainesville, Texas.

SECTION 2

1. GENERAL NOTES

The moment tables presented in Section 2 are divided into the following groups:

Group 3 - Three Span Beams,

Group 4 - Four Span Beams,

Group 5 - Five Span Beams,

Group 6 - Six Span Beams,

Group 7 - Seven Span Beams,

Group 8 - Eight Span Beams.

Each group is subdivided into three subgroups:

- 1. Equal Span Beams,
- 2. Symmetrical Beams,
- 3. Cyclosymmetrical Beams.

Each table is composed of the following parts:

- 1. Description of the beam and structural identities,
- 2. Illustration of beam,
- 3. Constant functions of New Distribution Factors,
- 4. Table of final moments.

In the application of the following tables, follow this procedure:

- 1. Select the table for the case to be investigated,
- 2. Determine all the stiffness factors,
- 3. Determine all the required functions of the New Distribution Factors (a,b,c,d x,y and z),

- 4. Determine all the required load functions (fixed end moments, sums of the fixed end moments at the points of support or the applied couples),
- Substitute all the computed values into the Table of Final Moments,
- 6. Determine the final moments from the Table of Final Moments. For example, from Table C3, the final moment

$$\mathbf{M}_{\mathrm{AB}} = - \ \mathbf{F} \mathbf{M}_{\mathrm{AB}} - \frac{\mathbf{a}}{\mathbf{x}} (\mathbf{F} \mathbf{M}_{\mathrm{BA}} - \mathbf{F} \mathbf{M}_{\mathrm{BC}}) - \frac{\mathbf{ac}}{\mathbf{x}} (\mathbf{F} \mathbf{M}_{\mathrm{CB}} - \mathbf{F} \mathbf{M}_{\mathrm{CD}}).$$

The proper sign is incorporated in each formula in every table, based on the principles defined in "Nomenclature".

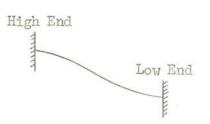
The tables may be used to compute the end moments due to the following:

- 1. Static or moving loads,
- 2. Rotations at the points of support,
- 3. Displacements of the supports.

All the tables were prepared to be used mainly for the computation of the end moments due to static or moving loads; therefore, the correct signs for each fixed end moment are included in each formula. Every fixed end moment is to be taken as positive when substituting into the Moment Table and the table will algebraically work out the proper sign.

If the end moments due to applied couples at the points of support are desired, the fixed end moments in the second vertical column of the selected table are placed equal to zero. The sum of the fixed end moments at the point of support where the couple is applied, is replaced by the numerical value of that couple. All couples are positive if they act in a clockwise direction, and negative if they act in a counterclockwise direction.

Sometimes the end moments due to the displacements of the supports are desired. In cases of this type a more complicated rule of signs must be followed.



- The fixed end moment of the high end is to be taken as positive.
- 2. The fixed end moment of the low end is to be taken as negative

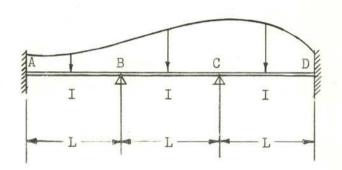
The above rules can be verified by basic relationships. If all the tables were derived for transverse loads and all the signs of the fixed end moments have been included in each formula, the multiplication factor (-1) has to be used to correct the sign for the low end. This modification must be used throughout the given table.

The tables were prepared for continuous beams with fixed ends, but may be modified for any end conditions. The modifications for two very common end conditions are explained in the following paragraphs.

If an exterior end is freely supported, the exterior span becomes a propped beam instead of a fixed end beam. The fixed end moment at the freely supported end becomes equal to zero, and the fixed end moment at the other end of the propped span is replaced by a propped end moment. The stiffness factor of the exterior span must be replaced by a modified stiffness factor ($K^{\circ} = \frac{3}{7} K$).

If the exterior span has an overhanging end, the fixed end moment at the freely supported end is equal to the cantilever moment and the fixed end moment of the other end is replaced by a propped end moment. This adjustment may be done numerically and the tabular procedure applied. The stiffness factor of the exterior span must be replaced by a modified stiffness factor ($K^2 = \frac{3}{4}$ K).

Beam with ends built in and freely supported at two points. Supports equally spaced.



CONSTANT FUNCTIONS OF NEW DISTRIBUTION FACTOR

$$K = \frac{I}{L}$$

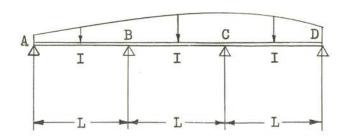
$$a = \frac{1}{4}$$

$$x = \frac{15}{16}$$

	TABLE OF FINAL MOMENTS			
Moment	Fixed End Moment	$FM_{BA} - FM_{BC}$	FM _{CB} - FM _{CD}	
M _{AB}	- FM _{AB}	2667	.0667	
M _{BA}	FM _{BA}	5333	.1333	
M _{BC}	- FM _{BA}	.5333	1333	
M _{CB}	FM _{CD}	1333	•5333	
^M CD	- FM _{CD}	.1333	5333	
^M DC	FM _{DC}	.0667	2667	

Beam freely supported at four points.

Supports equally spaced.



PROPPED BEAM END MOMENTS

CONSTANT FUNCTIONS OF NDF

$$EM_{BA}^* = \frac{FM_{AB}}{2} + FM_{BA}$$

$$EM_{CD}^* = FM_{CD} + \frac{FM_{DC}}{2}$$

$$K = \frac{I}{L}$$

$$K' = \frac{3I}{4L}$$

$$a = \frac{3}{14}$$

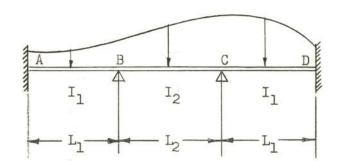
$$b = \frac{2}{7}$$

$$x = \frac{45}{49}$$

TABLE OF FINAL MOMENTS

Moment	Fixed End Moment	EM _{BA} - FM _{BC}	FM _{CB} EM _{CD}
M _{BA}	EM*BA	4667	.1333
^M BC	- EMBA	.4667	1333
^M CB	EM _{CD}	1333	.4667
^M CD	- EM _{CD}	-1333	4667

Beam with ends built in and freely supported at two points. Supports symmetrically spaced.



CONSTANT FUNCTIONS OF NEW DISTRIBUTION FACTOR

$$K_1 = I_1$$

$$K_2 = \frac{I}{L_2}$$

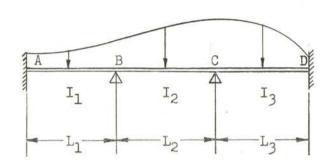
$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$

$$b = \frac{K_2}{2(K_1 + K_2)}$$

$$x = 1 - b^2$$

TABLE OF FINAL MOMENTS			
Moment	Fixed End Moment	${ m FM}_{ m BA} - { m FM}_{ m BC}$	FM _{CB} - FM _{CD}
M _{AB}	- FM _{AB}	- <u>a</u> x	ab X
M _{BA}	FM _{BA}	- 2 <u>a</u>	2 <u>ab</u>
M _{BC}	- FM _{BA}	2 <u>a</u>	= 2 <u>ab</u>
М _{СВ}	FM _{CD}	- 2 <u>ab</u>	2 <u>a</u>
$^{ m M}_{ m CD}$	- FM _{CD}	2 <u>ab</u>	- 2 a x
M _{DC}	$^{\mathrm{FM}}_{\mathrm{DC}}$	<u>ab</u>	- A

Beam with ends built in and freely supported at two points. Supports unequally spaced.



CONSTANT FUNCTIONS OF NEW DISTRIBUTION FACTOR

$$K_1 = \frac{I}{L_1}$$
 $K_2 = \frac{I}{L_2}$ $K_3 = \frac{I}{L_3}$

$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$

$$a = \frac{K_1}{2(K_1 + K_2)}$$

$$b = \frac{K_2}{2(K_1 + K_2)}$$

$$c = \frac{K_2}{2(K_2 + K_3)}$$

$$d = \frac{K_3}{2(K_2 + K_3)}$$

$$x = 1 - bc$$

TABLE OF FINAL MOMENTS			
Moment	Fixed End Moment	FM _{BA} - FM _{BC}	FM _{CB} - FM _{CD}
M _{AB}	- FM _{AB}	- <u>a</u>	ac x
M _{BA}	$^{ m FM}_{ m BA}$	- 2 <u>a</u> x	2 ac x
M _{BC}	- FM _{BA}	2 <u>a</u>	= 2 <u>ac</u>
M _{CB}	F ^M CD	- 2 <u>bd</u>	2 <u>d</u>
^M CD	- FM _{CD}	2 <u>bd</u>	2 <u>d</u>
M _{DC}	F ^M DC	bd x	- <u>d</u>

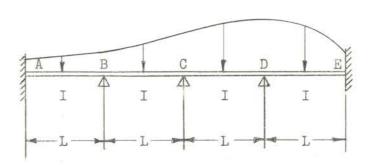
CASE A40

FOUR SPAN BEAM

EQUAL SPANS

DESCRIPTION:

Beam with ends built in and freely supported at three points. Supports equally spaced.



$$K = \frac{T}{I}$$

$$a = \frac{1}{h}$$

$$x = \frac{7}{8}$$

$$y = \frac{15}{16}$$

$$u = \frac{7}{2}$$

	TABLE OF FINAL MOMENTS									
Moment	Fixed End Moment	FM _{BA} FM _{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}						
M _{AB}	- FM _{AB}	2679	.0715	0179						
$^{ m M}_{ m BA}$	$^{ m FM}_{ m BA}$	5358	.1429	0357						
M _{BC}	- FM _{BA}	•5358	1429	•0357						
$^{ m M}_{ m CB}$	FM _{CB}	1250	5000	.1250						
$^{ m M}_{ m GD}$	- FM _{CD}	.1250	5000	1250						
M _{DG}	FM _{DE}	.0357	1429	•5358						
$^{ m M}_{ m DE}$	- FM _{DE}	0357	.1429	~ ₀ 5358						
$^{ m M}_{ m ED}$	$^{ m FM}_{ m ED}$	0179	.0715	2679						

CASE ALL

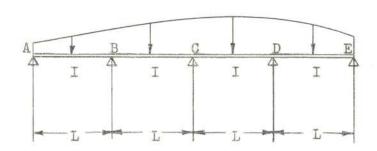
FOUR SPAN BEAM

EQUAL SPANS

DESCRIPTION:

Beam freely supported at five points.

'Supports equally spaced.



PROPPED BEAM END MOMENTS

$$\text{EM}_{\text{BA}}^{*} = \frac{\text{FM}_{\text{AB}}}{2} + \text{FM}_{\text{BA}}$$

$$EM_{DE}^* = FM_{DE} + \frac{FM_{ED}}{2}$$

$$K = \frac{I}{L}$$

$$A = \frac{3}{14}$$

$$A = \frac{3}{7}$$

$$A = \frac{6}{7}$$

$$A = \frac{13}{14}$$

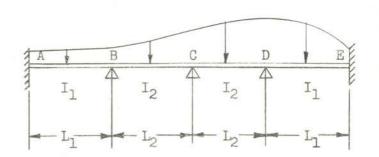
$$A = \frac{12}{7}$$

$$A = \frac{13}{14}$$

$$A = \frac{12}{7}$$

Moment	Fixed End Moment	EMBA - FMBC	FM _{CB} - FM _{CD}	FM _{DC} - EM [*] _{DE}
[™] BA	** EM _{BA}	4643	.1250	0357
M _{BC}	- EM _{BA}	•4643	1250	•0357
M _{CB}	FM _{CB}	1429	5000	•1429
^M CD	- FM _{CD}	•1429	5000	1429
M _{DC}	M _{DC} EM [*] DE		1250	•4643
$^{ m M}_{ m DE}$	EM _{DE}	0357	.1250	4643

Beam with ends built in and freely supported at three points. Supports symmetrically spaced.



$$K_{1} = I_{1}$$

$$I_2 = I_2$$

$$a = \frac{K_1}{2(K_1 + K_2)}$$

$$b = \frac{K_2}{2(K_1 + K_2)} \qquad c = \frac{K_2}{2(K_2 + K_2)} = \frac{1}{4}$$

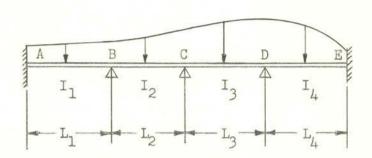
$$x = 1 - \frac{b}{2}$$

$$y = 1 - \frac{b}{4}$$

$$u = 2 - b$$

TABLE OF FINAL MOMENTS								
Moment	Fixed End Moment	FM _{BA} ~ FM _{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}				
M _{AB}	- FM _{AB}	ay x	<u>a</u> 4x	<u>ab</u> 4x				
$^{ m M}_{ m BA}$	FM _{BA}	- 2 ay	a 2x	ab 2x				
$^{ m M}_{ m BC}$	- FM _{BA}	2 ay	a 2x	ab 2x				
^M CB	F ^M CB	b 2	- 1 2	<u>b</u> 2				
$^{ m M}_{ m CD}$	- FM _{CD}	<u>b</u> 2	- ½	- <u>b</u> 2				
$^{ m M}_{ m DC}$	FM _{DE}	ab 2x	- <u>a</u> - <u>2</u> x	2 ay x				
$^{ m M}_{ m DE}$	- FM _{DE}	ab 2x	a 2x	- 2 ay				
$^{ m M}_{ m ED}$	$_{ m ED}$	$-\frac{ab}{4x}$	<u>a</u> 4x	ay x				

Beam with ends built in and freely supported at three points. Supports unequally spaced.



$$K_1 = \frac{I}{L_1}$$
 $K_2 = \frac{I}{L_2}$ $K_3 = \frac{I}{L_3}$ $K_4 = \frac{I}{L_4}$

$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$ $c = \frac{K_2}{2(K_2 + K_3)}$

$$d = \frac{K_3}{2(K_2 + K_3)} \quad e = \frac{K_3}{2(K_3 + K_4)} \quad f = \frac{K_4}{2(K_3 + K_4)}$$

$$x = 1 - bc - de$$

$$y_1 = 1 - bc$$

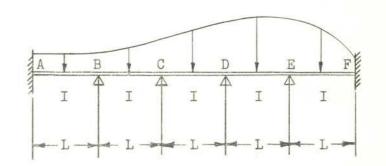
$$y_2 = 1 - de$$

$$u_1 = 2 - b$$

$$u_2 = 2 - e$$

7	TABLE OF FINAL MOMENTS									
Moment	Fixed End Moment	$_{ m BA}$ - $_{ m BC}$	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}						
$^{ m M}{}_{ m AB}$	- FM _{AB}	- ay x2	ac x	ace						
$^{ m M}_{ m BA}$	FM _{BA}	- 2 ay 2	2 <u>ac</u>	- 2 ace						
M _{BC}	- FM _{BA}	$2\frac{ay}{x}2$	- 2 ac	2 ace						
^M CB	FM _{CB}	- bdu _x	- cu	ceul						
^M GD	- FM _{CD}	bdu ₂	- <u>du</u> 2	ceu ₁						
^M DC	$_{ m DE}$	2 bdf	- 2 <u>df</u>	2 <u>fy</u> l						
$^{ m M}_{ m DE}$	- FM _{DE}	- 2 bdf	$2\frac{df}{x}$	- 2 <u>fy</u> 1						
$^{ m M}_{ m ED}$	$^{ m FM}_{ m ED}$	bdf x	df x	- fyl						

Beam with ends built in and freely supported at four points. Supports equally spaced.



$$K = \frac{I}{I}$$

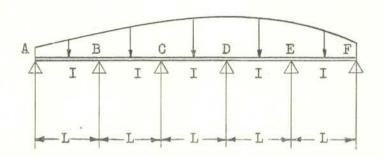
$$a = \frac{1}{4}$$

$$x = \frac{15}{16}$$

$$z = \frac{209}{16}$$

	TABLE OF FINAL MOMENTS								
Moment	Fixed End Moment	$^{\rm FM}_{\rm BA}$ \sim $^{\rm FM}_{\rm BC}$	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}	FM _{ED} - FM _{EI}				
M AB	- FM AB	2679	.0718	0191	.0048				
M BA	FM _{BA}	5359	.1436	0382	.0096				
M _{BC}	∞ FM _{BA}	.5359	1436	.0382	0096				
M _{CB}	· FM _{CD}	1242	.4976	.1340	0335				
M _{CD}	- FM _{CD}	.1242	4976	1340	.0335				
M _{DC}	$^{ m FM}_{ m DC}$.0335	1340	4976	.1242				
M _{DE}	- FM _{DC}	0335	.1340	.4976	1242				
MED	FM _{EF}	0096	.0382	1436	.5359				
${\rm M_{EF}}$	- FM _{EF}	.0096	0382	.1436	5359				
$^{ m M}_{ m FE}$	FMFE	.0048	0191	.0718	2679				

Beam freely supported at six points. Supports equally spaced.



PROPPED BEAM END MOMENTS

CONSTANT FUNCTIONS OF NDF

$$EM_{BA}^* = \frac{FM_{AB}}{2} + FM_{BA}$$

$$EM_{EF}^* = FM_{EF} + \frac{FM_{FE}}{2}$$

$$K = \frac{I}{L}$$

$$K' = \frac{3I}{4L}$$

$$a = \frac{3}{14}$$

$$b = \frac{2}{7}$$

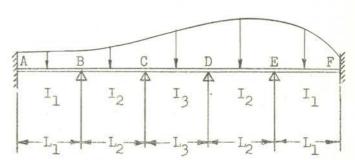
$$c = d = \frac{1}{4}$$

$$x = \frac{13}{14}$$

$$z = \frac{627}{49}$$

	TABLE OF FINAL MOMENTS									
Moment	Fixed End Moment	EMBA - FMBC	FM _{CB} - FM _{CD}	FM _{DG} - FM _{DE}	FM _{ED} - EM [*] _{EF}					
$^{ m M}_{ m BA}$	EM _{BA}	- 0.4641	0.1244	- 0.0334	0.0096					
M _{BC}	- EM _{BA}	0.4641	- 0.1244	0.0334	- 0.0096					
M _{CB}	FM _{GD}	~ 0.1435	0.5024	0.1334	- 0.0383					
^M CD	- FM _{CD}	0.1435	- 0,5024	- 0.1334	0.0383					
M _{DC}	FMDC	0.0383	- 0.1334	- 0.5024	0.1435					
$^{ m M}_{ m DE}$	- FM _{DC}	- 0.0383	0.1334	0.5024	- 0.1435					
$^{ m M}_{ m ED}$	EM _{EF}	- 0.0096	0.0334	- 0.1244	0.4641					
$^{ m M}_{ m EF}$	- em*	0.0096	- 0.0334	0.1244	- 0.4641					

Beam with ends built in and freely supported at four points. Supports symmetrically spaced.



$$\kappa_1 = \frac{I}{L_1} \qquad \kappa_2 = \frac{I}{L_2} \qquad \kappa_3 = \frac{I}{L_3}$$

$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$

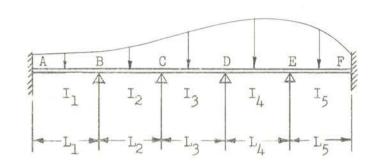
$$c = \frac{K_2}{2(K_2 + K_3)}$$
 $d = \frac{K_3}{2(K_2 + K_3)}$

$$x = 1 - bc$$

$$z = \left(\frac{x}{d}\right)^2 - 1$$

	TABLE OF FINAL MOMENTS								
Moment	Fixed End Moment	FM _{BA} - FM _{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}	FM _{ED} - FM _{EF}				
M _{AB}	- FM _{AB}	$-\frac{a}{x}(1 \div \frac{bc}{z})$	$\frac{\operatorname{ac}}{\mathbf{x}}(1+\frac{1}{\mathbf{z}})$	= ac dz	abc dz				
M _{BA}	FM _{BA}	$-\frac{2a}{x}(1+\frac{bc}{z})$	$\frac{2ac}{x}(1+\frac{1}{z})$	2ac dz	2abc dz				
M _{BC}	- FM _{BA}	$\frac{2a}{x}(1+\frac{bc}{z})$	$-\frac{2ac}{x}(1+\frac{1}{z})$	2ac dz	_ <u>2abc</u> dz				
$^{ m M}_{ m CB}$	FM _{CD}	$-\frac{b}{z}(\frac{2x}{d}-1)$	$\frac{1}{z}(\frac{2x}{d}-1)$	$\frac{1}{z}(\frac{x}{d}-2)$	$-\frac{b}{z}(\frac{x}{d}-2)$				
$^{\rm M}_{ m CD}$	- FM _{CD}	$\frac{b}{z}(\frac{2x}{d}-1)$	$=\frac{1}{z}(\frac{2x}{d}-1)$	$-\frac{1}{z}(\frac{x}{d}-2)$	$\frac{b}{z}(\underline{x}-2)$				
^M DC	FM _{DC}	$\frac{b}{z}(\frac{x}{d}-2)$	$-\frac{1}{z}(\frac{x}{d}-2)$	$-\frac{1}{z}(\frac{2x}{d}-1)$	$\frac{b}{z}(\frac{2x}{d}-1)$				
M _{DE}	- FM _{DC}	$-\frac{b}{z}(\frac{x}{d}-2)$	$\frac{1}{z}(\frac{x}{d}-2)$	$\frac{1}{z}(\frac{2x}{d}-1)$	$-\frac{b}{z}(\frac{2x}{d}-1)$				
MED	${ ilde FM}_{ m EF}$	= 2abc dz	2ac cz	$-\frac{2ac}{x}(1+\frac{1}{z})$	$\frac{2a}{x}(1+\frac{bc}{z})$				
$^{ m M}_{ m EF}$	- FM _{EF}	2abc dz	= 2ac dz	$\frac{2ac}{x}(1+\frac{1}{z})$	$-\frac{2a}{x}(1+\frac{bc}{z})$				
$M_{ m FE}$	FMFE	abc dz	= ac dz	$\frac{ac}{x}(1+\frac{1}{z})$	$-\frac{a}{x}(1*\frac{bc}{z})$				

Beam with ends built in and freely supported at four points. Supports unequally spaced.



$$K_{1} = \frac{I}{I_{1}} \quad K_{2} = \frac{I}{I_{2}} \quad K_{3} = \frac{I}{I_{3}} \quad K_{4} = \frac{I}{I_{4}} \quad K_{5} = \frac{I}{I_{5}}$$

$$a = \frac{K_{1}}{2(K_{1} + K_{2})} \quad b = \frac{K_{2}}{2(K_{1} + K_{2})} \quad c = \frac{K_{2}}{2(K_{2} + K_{3})}$$

$$d = \frac{K_{3}}{2(K_{2} + K_{3})} \quad e = \frac{K_{3}}{2(K_{3} + K_{4})} \quad f = \frac{K_{4}}{2(K_{3} + K_{4})}$$

$$g = \frac{K_{4}}{2(K_{4} + K_{5})} \quad h = \frac{K_{5}}{2(K_{4} + K_{5})}$$

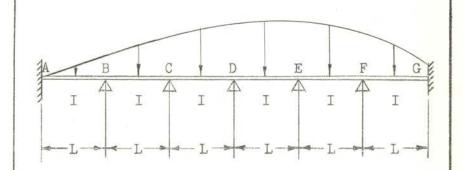
$$x = 1 - bc$$

$$y = 1 - fg$$

$$z = \frac{xy}{de} - 1$$

	TABLE OF FINAL MOMENTS								
Moment	Fixed End Moment	FM _{BA} - FM _{BC}	FM CB - FM CD	FM DC - FM DE	FM _{ED} - FM _{EF}				
M AB	- FM AB	$-\frac{a}{x}(1+\frac{bc}{z})$	$\frac{ac}{x}(1+\frac{1}{z})$	_ ac dz	acg dz				
M BA	FM BA	$-\frac{2a}{x}(1+\frac{bc}{z})$	$\frac{2ac}{x}(1+\frac{1}{z})$	- 2ac dz	2acg dz				
M BC	- FM BA	$\frac{2a}{x}(1+\frac{bc}{z})$	$-\frac{2ac}{x}(1+\frac{1}{z})$	2ac dz	_ 2acg dz				
Мсв	FM CD	$-\frac{b}{z}(\frac{2y}{e}-1)$	$\frac{1}{z}(\frac{2y}{e}-1)$	$\frac{1}{z}(\frac{x}{d}-2)$	$-\frac{g(x}{d}-2)$				
M CD	- FM CD	$\frac{b}{z}(\frac{2y}{e}-1)$	$-\frac{1}{z}(\frac{2y}{e}-1)$	$-\frac{1}{z}(\frac{x}{d}-2)$	$\frac{g(x)}{z} = 2$				
M DC	FM DC	$\frac{b}{z}(\frac{y}{e}-2)$	$-\frac{1}{z}(\frac{y}{e}-2)$	$-\frac{1}{z}(\frac{2x}{d}-1)$	$\frac{g}{z}(\frac{2x}{d}-1)$				
M DE	- FM DC	$-\frac{b}{z}(\frac{y}{e}-2)$	$\frac{1}{z}(\frac{y}{e}-2)$	$\frac{1}{z}(\frac{2x}{d}-1)$	$= \frac{g}{z}(\frac{2x}{d} - 1)$				
M ED	FM _{EF}	- 2bfh ez	2fh ez	$-\frac{2fh}{y}(1+\frac{1}{z})$	$\frac{2h}{y}(1+\frac{fg}{z})$				
M EF	- FM _{EF}	2bfh ez	2fh ez	$\frac{2fh}{y}(1+\frac{1}{z})$	$-\frac{2h}{y}(1+\frac{fg}{z})$				
M FE	FM FE	bfh ez	- fh ez	$\frac{\mathrm{hf}}{\mathrm{y}}(1+\frac{1}{\mathrm{z}})$	$-\frac{h}{y}(1+\frac{fg}{z})$				

Beam with ends built in and freely supported at five points. Supports equally spaced.



$$K = \frac{I}{L}$$

$$a = \frac{1}{4}$$

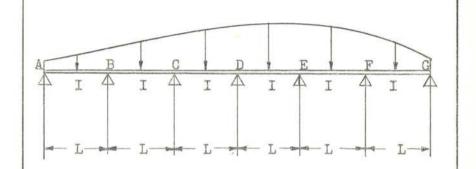
$$x = \frac{15}{16}$$

$$u = \frac{7}{4}$$

$$z = 13$$

	TABLE OF FINAL MOMENTS									
Moment	Fixed End Moment	FM _{BA} - FM _{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}	FM _{ED} - FM _{EF}	FM _{FE} - FM _{FG}				
M _{AB}	→ FM _{AB}	2680	.0718	0192	.0051	0013				
M _{BA}	FM _{BA}	5360	.1436	0384	.0102	0026				
M _{BC}	- FM _{BA}	.5360	1436	.0384	0102	.0026				
M _{CB}	FM _{CD}	1242	4973	.1345	0359	.0090				
M _{CD}	⊶ FM _{CD}	.1242	.4973	1345	.0359	∞ .0090				
MDC	FM _{DC}	.0333	1333	5000	.1333	0333				
$^{ m M}_{ m DE}$	- FM _{DE}	0333	•1333	5000	1333	.0333				
$^{ m M}_{ m ED}$	FMED	0090	.0359	1345	•4973	.1242				
$M_{\overline{EF}}$	~ FM _{ED}	,0090	0359	.1345	4973	1242				
MFE	FMFG	.0026	0102	.0384	1436	.5360				
$^{ m M}_{ m FG}$	- FM _{FG}	0026	.0102	0384	.1436	5360				
$^{ m M}_{ m GF}$	FMGF	0013	.0051	0192	.0718	2680				

Beam freely supported at seven points. Supports equally spaced.



PROPPED BEAM END MOMENTS

CONSTANT FUNCTIONS OF NDF

$$EM_{BA}^* = \frac{FM_{AB}}{2} + FM_{BA}$$

$$\text{EM}_{\text{FG}}^{*} = \text{FM}_{\text{FG}} + \frac{\text{FM}_{\text{GF}}}{2}$$

$$X = \frac{I}{L}$$

$$b = \frac{2}{7}$$

$$x = \frac{13}{14}$$

$$K' = \frac{3K}{4L}$$

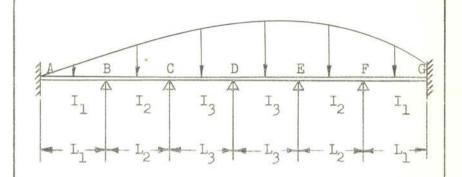
$$a = \frac{3}{14}$$
 $b = \frac{2}{7}$ $c = d = e = \frac{1}{4}$

$$z = \frac{90}{7}$$

$$u = \frac{7}{4}$$

		TAI	BLE OF FINAL MOM	ENTS		
Moment	Fixed End Moment	EM* - FMBC	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}	$FM_{ED} - FM_{EF}$	FM _{FE} - EM _{FG}
M _{BA}	EM* BA	4641	.1244	0333	.0090	0026
M _{BC}	- EMBA	.4641	1244	.0333	0090	.0026
M _{CB}	FM _{CB}	1436	5026	.1333	0359	.0103
^M CD	- FM _{CB}	.1436	.5026	1333	.0359	0103
M _{DC}	FMDC	.0385	1342	5000	.1342	0385
$^{\mathrm{M}}_{\mathrm{DE}}$	- FM _{DE}	0385	.1342	5000	1342	.0385
$^{\mathrm{M}}_{\mathrm{ED}}$	FMED	0103	.0359	1333	.5026	.1436
M _{EF}	- FMED	.0103	0359	.1333	5026	1436
M _{FE}	EM*	•0026	0090	.0333	1244	.4641
$^{ m M}_{ m FG}$	- EMFG	- ,0026	.0090	0333	.1244	4641

Beam with ends built in and freely supported at five points. Supports symmetrically spaced.



$$K_1 = \frac{I}{L_1}$$
 $K_2 = \frac{I}{L_2}$ $K_3 = \frac{I}{L_3}$

$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$ $c = \frac{K_2}{2(K_2 + K_3)}$

$$d = \frac{1_3}{2(K_2 + K_3)}$$
 $e = \frac{K_3}{2(K_3 + K_3)} = \frac{1}{4}$

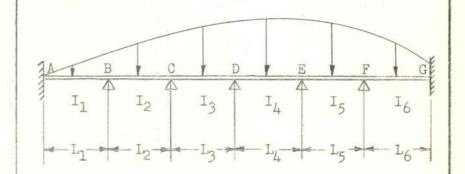
$$x = 1 - bc$$

$$u = \frac{7}{4}$$

$$z = 4 \frac{x}{d} - 2$$

Moment	Fixed End Moment	FM _{BA} - FL _{BC}	Fich - Fich	Fide - Fide	FMED - FMER	Hipe - Hipg
M _{AB}	- FIAB	$-\frac{a}{x}(1+\frac{bc}{2})$	$\frac{ac}{x}(1+\frac{1}{z})$	- ac	8.C 322	- abc
M _{BA}	Pic _{BA}	$-\frac{2a}{x}(1+\frac{bc}{z})$	$\frac{2ac}{x}(1+\frac{1}{2})$	- 2ac - dz	2ac xxx	- 2abc
H _{BC}	- FM _{EA}	$\frac{2a}{x}(1+\frac{bc}{z})$	$-\frac{2ac}{x}(1+\frac{1}{2})$	2ac dz	- 2ac	2abo xz
N _{CB}	Pi _{CD}	$-\frac{b}{z}(7-\frac{2d}{x})$	$-\frac{1}{2}(7-\frac{2d}{x})$	$\frac{1}{2}(\frac{x}{d}-2)$	$-\frac{1}{z}(1-\frac{2d}{x})$	$\frac{b}{z}(1-\frac{2d}{x})$
r _{cd}	- FM _{CD}	$\frac{b}{z}(7-\frac{2d}{x})$	$\frac{1}{2}(7-\frac{2d}{x})$	$-\frac{1}{\pi}(\frac{x}{d}-2)$	$\frac{1}{z}(1-\frac{2d}{x})$	$-\frac{b}{z}(1-\frac{2d}{x})$
1′DC	FM _{DC}	<u>bd</u> 2x	- <u>d</u>	- 1/2	₫ 2x	- bd 2x
NDE	- FM _{DE}	- <u>bd</u>	d 27:	- 1/2	- d /2x	bd 2x
MED	FM _{ED}	$-\frac{b}{z}(1-\frac{2d}{x})$	$\frac{1}{2}(1-\frac{2d}{x})$	$-\frac{1}{2}(\overset{\times}{\underline{c}}-2)$	$\frac{1}{2}(7-\frac{2d}{x})$	$\frac{b}{z}(7-\frac{2d}{x})$
EF	- Fi:ED	$\frac{b}{z}(1-\frac{2d}{x})$	$-\frac{1}{z}(1-\frac{2d}{x})$	$\frac{1}{z}(\frac{z}{d}-2)$	$-\frac{1}{2}(7-\frac{2d}{x})$	$-\frac{b}{z}(7-\frac{2d}{x})$
11FE	FF. FG	2abc x2	- 2ac	2ac dz	$-\frac{2ac}{x}(1+\frac{1}{z})$	$\frac{2a}{x}(1+\frac{bc}{z})$
N _{FG}	- Pigg	- <u>2abc</u>	2ac xx	- 2 <u>ac</u>	$\frac{2ac}{x}(1+\frac{1}{z})$	$-\frac{2a}{x}(1+\frac{bc}{z})$
l'GF	n'gr	- <u>abc</u>	ac xz	- ac dz	$\frac{ac}{c}(1+\frac{1}{2})$	$-\frac{a}{x}(1+\frac{bc}{z})$

Beam with ends built in and freely supported at five points. Supports unequally spaced.



$$K_{1} = \frac{I1}{L_{1}} \quad K_{2} = \frac{I2}{L_{2}} \quad K_{3} = \frac{I3}{L_{3}} \quad K_{4} = \frac{I4}{L_{4}} \quad K_{5} = \frac{I5}{L_{5}} \quad K_{6} = \frac{I6}{L_{6}}$$

$$x_{1} = \frac{I1}{L_{1}} \quad K_{2} = \frac{I2}{L_{2}} \quad K_{3} = \frac{I3}{L_{3}} \quad K_{4} = \frac{I4}{L_{4}} \quad K_{5} = \frac{I5}{L_{5}} \quad K_{6} = \frac{I6}{L_{6}}$$

$$x_{2} = 1 - hj$$

$$x_{1} = 1 - bc$$

$$x_{2} = 1 - hj$$

$$x_{2} = 1 - hj$$

$$x_{1} = 1 - bc$$

$$x_{2} = 1 - hj$$

$$x_{2} = 1 - hj$$

$$x_{1} = 1 - bc$$

$$x_{2} = 1 - hj$$

$$x_{2} = 1 - hj$$

$$x_{2} = 1 - hj$$

$$x_{3} = 1 - kc$$

$$x_{4} = 1 - kc$$

$$x_{5} = 1 - kc$$

$$x_{2} = 1 - kc$$

$$x_{2} = 1 - kc$$

$$x_{2} = 1 - kc$$

$$x_{3} = 1 - kc$$

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$$x_{2} = 1 - kc$$

$$x_{3} = 1 - kc$$

$$x_{4} = 1 - kc$$

$$x_{5} = 1 - kc$$

$$x_{6} = 1 - kc$$

$$x_1 = 1 - bc$$

$$x_2 = 1 - hj$$

$$u_1 = 2 - e$$

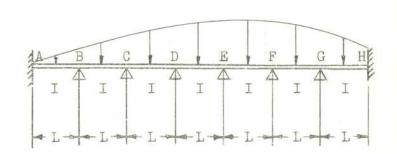
$$u_2 = 2 - f$$

$$z_1 = \frac{x_1}{de} - 1 - \frac{fgx_1}{dex_2}$$

$$z_2 = \frac{x_2}{fg} - 1 - \frac{dex_2}{fgx_1}$$

_	-			AL POPERILO		
Moment	Fixed End Moment	m_{BA} - m_{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FIL	M _{ED} - M _{EF}	Fi _{FE} - Fi _{FG}
M _{AB}	- fm _{ab}	$-\frac{a}{x_{\underline{1}}}(1+\frac{bc}{z_{\underline{1}}})$	$\frac{ac}{x_1}(1+\frac{1}{z_1})$	$-\frac{ac}{dz_1}$	acg dx ₂ ² 1	- aogj dx ₂ z ₁
M _{BA}	FM _{BA}	$-\frac{2a}{x_1}(1+\frac{bc}{z_1})$	$\frac{2ac}{x_1}(1+\frac{1}{z_1})$	$-\frac{2ac}{dz_1}$	2acg dx ₂ z ₁	- 2acgj dx ₂ z ₁
MBC	- FM _{BA}	$\frac{2a}{x_1}(1+\frac{bc}{x_1})$	$-\frac{2ac}{x_1}(1+\frac{1}{x_1})$	2ac dz ₁	- 2acg dx ₂ z ₁	2acgj dx ₂ z ₁
M _{CB}	FN _{CD}	$-\frac{b}{ez_1}(u_1-\frac{2fg}{x_2})$	$\frac{1}{ez_1}(u_1 - \frac{2fg}{x_2})$	$\frac{1}{z_1}(\frac{x_1}{d}-2)$	$-\frac{g}{x_2^2 z_1} (\frac{x_1}{d} - 2)$	$\frac{gj}{x_2z_1}(\frac{x_1}{d}-2)$
M _{CD}	- Filod	$\frac{b}{ez_1}(u_1-\frac{2fg}{x_2})$	$-\frac{1}{ez_1}(u_1 - \frac{2fg}{x_2})$	$-\frac{1}{z_1}(\frac{x_1}{d}-2)$	$\frac{g}{x_2^2 z_1} (\frac{x_1}{d} - 2)$	$-\frac{gj}{x_2z_1}(\frac{x_1}{d}-2)$
11 _{DC}	FILDC	$\frac{bd}{x_1 x_2} (\frac{2x_2}{g} - 1)$	$-\frac{d}{x_1x_2}(\frac{2x_2}{g}-1)$	$-\frac{1}{z_1}(\frac{2x_1}{d}-1)$	$\frac{g}{x_2^2 x_1^2} (\frac{2x_1}{d} - 1)$	$-\frac{4gj}{x_2^2 1}(\frac{2x_1}{d}-1)$
M _{DE}	- FM _{DE}	$-\frac{bd}{x_1x_2}(\frac{2x_2}{g}-1)$	$\frac{\frac{d}{x_1 x_2} (\frac{2x_2}{g} - 1)}{(\frac{2x_2}{g} - 1)}$	$-\frac{1}{z_2}(\frac{2x_2}{g}-1)$	$-\frac{g}{x_2^{z_1}}(\frac{2x_1}{d}-1)$	$\frac{gj}{x_2z_1}(\frac{2x_1}{d}-1)$
N _{ED}	FI _{ED}	$-\frac{bd}{x_1 x_2} (\frac{x_2}{g} - 2)$	$\frac{d}{x_1 x_2} (\frac{x_2}{g} - 2)$	$-\frac{1}{z_2}(\frac{x_2}{g}-2)$	$-\frac{1}{fz_2}(u_2 - \frac{2de}{x_1})$	$\frac{\mathrm{j}}{\mathrm{fz}_2}(\mathrm{u}_2-\frac{2\mathrm{de}}{\mathrm{x}_1})$
MEF	- FM _{ED}	$\frac{\mathrm{bd}}{\mathrm{x_1}\mathrm{x_2}}(\frac{\mathrm{x_2}}{\mathrm{g}}-2)$	$-\frac{d}{x_1^2}(\frac{x_2}{g}-2)$	$\frac{1}{z_2}(\frac{x_2}{g}-2)$	$\frac{1}{f^2 z_2} (u_2 - \frac{2de}{x_1})$	$-\frac{1}{12}(u_2 - \frac{2de}{x_1})$
M _{FE}	$^{ m FM}_{ m FG}$	2bdhk ^{gz} 1 ^z 2	- 2dhk gx ₁ z ₂	2hk E ² 2	$-\frac{2hk}{x_2}(1+\frac{1}{x_2})$	$\frac{2k}{x_2}(1+\frac{h\cancel{1}}{2})$
H _{FG}	- FM _{FG}	= 2bdhk E ^x 1 ² 2	2dhk ⁸⁷ 1 ² 2	- ^{2hk} / _{E²2}	$\frac{2hk}{x_2}(1+\frac{1}{x_2})$	$-\frac{2k}{\frac{x}{2}}(1+\frac{hj}{\frac{x}{2}})$
^M GF	FIGF	$=\frac{\text{bdhk}}{\mathbb{E}^{\chi}_{1}z_{2}}$	dhk E ^z 1 ^z 2	- hk/8 ² 2	$\frac{h!c}{x_2}(1+\frac{1}{x_2})$	$-\frac{k}{x_2}(1+\frac{hj}{x_2})$

Beam with ends built in and freely supported at six points. Supports equally spaced.



$$K = \frac{I}{L}$$

$$a = \frac{1}{L}$$

$$x = \frac{7}{8}$$

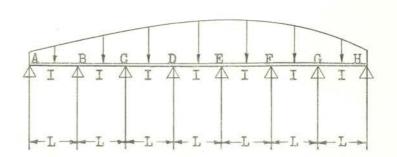
$$y = \frac{15}{16}$$

$$u = \frac{7}{4}$$

$$z = \frac{291}{22}$$

	TABLE OF FINAL MOMENTS									
Moment	Fixed End Moment	FM _{BA} - FM _{BC}	FM _{CB} - FM _{CD}	FM _{DC} - FM _{DE}	FM - FM EF	FM _{FE} - FM _{FG}	FM _{GF} - FM _{GI}			
$^{\mathrm{M}}_{\mathrm{AB}}$	- FM _{AB}	2681	.0718	0192	.0052	0014	.0003			
M _{BA}	FMBA	5362	.1436	0385	.0103	0028	.0007			
M _{BC}	- FM _{BA}	.5362	1436	.0385	0103	.0028	0007			
M _{CB}	FMCB	1244	5030	.1346	0361	.0096	0024			
$^{\mathrm{M}}_{\mathrm{CD}}$	- FM _{CB}	.1244	.5030	1346	.0361	0096	.0024			
MDG	FMDE	.0333	1333	.5000	.1340	0357	.0089			
$N_{\rm DE}$	- FM _{DE}	0333	.1333	5000	1340	.0357	0089			
"ED	FMED	0089	.0357	- :1340	5000	.1333	0333			
$M_{ m EF}$	- FM _{ED}	.0089	0357	.1340	.5000	1333	.0333			
M_{FE}	FMFG	.0024	0096	.0361	1346	.5030	.1244			
M_{FG}	- FM _{FG}	0024	.0096	0361	.1346	5030	1244			
$^{ m M}_{ m GF}$	F™GH	0007	.0028	0103	.0385	1436	.5362			
$^{\mathrm{M}}$ GH	- FM _{GH}	.0007	0028	.0103	0385	.1436	5362			
$^{\mathrm{M}}$ HG	FM _{HG}	.0003	0014	.0052	0192	.0718	2681			

Beam freely supported at eight points. Supports equally spaced.



PROPPED BEAM END MOMENTS

CONSTANT FUNCTIONS OF NDF

$$EM_{BA}^* = \frac{FM_{AB}}{2} + FM_{BA}$$

$$EM_{GH}^* = FM_{GH} + \frac{FM_{HG}}{2}$$

$$K = \frac{I}{L}$$

$$K^{\dagger} = \frac{3I}{4L}$$

$$a = \frac{13}{14}$$

$$b = \frac{2}{7}$$

$$a = \frac{13}{14}$$
 $b = \frac{2}{7}$ $c = d = e = f = \frac{1}{4}$

$$x = \frac{97}{112}$$

$$y_1 = \frac{13}{14}$$

$$x = \frac{97}{112}$$
 $y_1 = \frac{13}{14}$ $y_2 = \frac{15}{16}$

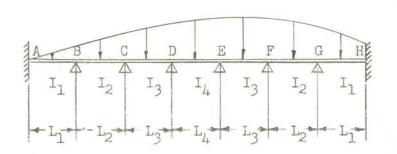
$$z = \frac{8733}{676}$$
 $u_1 = \frac{12}{7}$ $u_2 = \frac{7}{4}$

$$u_1 = \frac{12}{7}$$

$$u_2 = \frac{7}{4}$$

	+			1			
Moment	Fixed End Moment	EMBA - FMBC	FM _{CB} - FM _{CD}	FM _{DC} -FM _{DE}	FM _{ED} - FM _{EF}	FM _{FE} - FM _{FG}	FM _{GF} - EM _G
M _B A	EM*BA	4641	.1244	0333	.0089	0024	.0007
M _{BC}	- EM*	•4641	1244	.0333	0089	.0024	0007
M _{CB}	FM _{G3}	1436	· .4974	.1333	0357	.0096	0028
$^{\mathrm{M}}_{\mathrm{GD}}$	- FM _{CB}	.1436	.4974	1333	.0357	0096	.0028
$^{\rm M}_{ m DC}$	FMDE	.0385	1347	.5002	.1339	0361	.0103
MDE	- FM _{DE}	0385	.1347	5002	1339	.0361	0103
M _{ED}	FM_ED	0103	.0361	1339	5002	.1347	0385
M _{EF}	- FM _{ED}	.0103	036.1	. 133 9	.5002	1347	.0385
MFE	FMFG	.0028	0096	.0357	1333	•4974	.1436
$^{\mathrm{M}}_{\mathrm{FG}}$	- FM _{FG}	0028	.0096	0357	.1333	4974	1436
${ m M_{GF}}$	EM _{GH}	0007	.0024	0089	.0333	1244	.4641
M _{GH}	- EM [*] GH	.0007	0024	.0089	0333	.1244	4641

Beam with ends built in and freely supported at six points. Supports symmetrically spaced.



$$K_1 = \frac{I}{L_1}$$
 $K_2 = \frac{I}{L_2}$ $K_3 = \frac{I}{L_3}$ $K_4 = \frac{I}{L_4}$

$$a = \frac{K_1}{2(K_1 + K_2)}$$
 $b = \frac{K_2}{2(K_1 + K_2)}$ $c = \frac{K_2}{2(K_2 + K_3)}$

$$d = \frac{K_3}{2(K_2 + K_3)} \quad e = \frac{K_3}{2(K_3 + K_4)} \quad f = \frac{K_4}{2(K_3 + K_4)}$$

$$x = 1 - bc - de$$

$$y_1 = 1 - bc$$

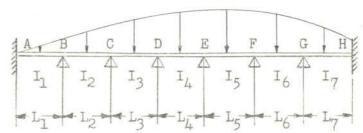
$$y_2 = 1 - de$$

$$u_1 = 2 - b$$

$$z = \left(\frac{x}{fy}\right)^2 - 1$$

Forent	fixed And Noment	FN _{BA} = FN _{BC}	F7 _{G3} - FE _{GD}	Fig-Fi _{DE}	FM _{ED} - FM _{EF}	Fi _{FE} - Fi _{FG}	FM _{GF} - FM _{GH}
M _{AB}	- FM _{AB}	$-\frac{a}{x}(y_2 + \frac{bcde}{y_1z})$	$\frac{ac}{x}(1 + \frac{de}{y_1 z})$	$-\frac{ace}{x}(1+\frac{1}{z})$	ace fy ₁ z	$-\frac{\text{acde}}{\text{fy}_{1}^{2}z}$	abcde fy ² ₁ z
M _{BA}	FM _{BA}	$-\frac{2a}{x}(y_2 + \frac{bcde}{y_1 z})$	$\frac{2ac}{x}(1+\frac{de}{y_1z})$	$-\frac{2ace}{x}(1+\frac{1}{2})$	2ace fy ₁ z	$-\frac{2acde}{fy_1^2z}$	2abcde fy2/z
M _{BC}	- FM _{BA}	$\frac{2a}{x}(y_2 + \frac{bcde}{y_1z})$	$-\frac{2ac}{x}(1+\frac{de}{y_1z})$	$\frac{2ace}{x}(1+\frac{1}{2})$	$-\frac{2ace}{fy_1z}$	2acde fy ² z	- ^{2abcde} fy ² ₁ z
MCB	FM _{CB}	$-\frac{bd}{x}(u_2-\frac{ceu_1}{y_1z})$	$-\frac{\operatorname{cu}_{1}}{\operatorname{x}}(1+\frac{\operatorname{de}}{\operatorname{y}_{1}\operatorname{z}})$	$\frac{\text{ceu}_1}{x}(1+\frac{1}{z})$	- ceul fylz	$\frac{\frac{\text{cdeu}_1}{\text{fy}_1^2 z}}$	$-\frac{\frac{bcdeu_1}{fy_1^2z}}$
M _{CD}	- FM _{CB}	$\frac{\mathrm{id}}{\mathrm{x}}(\mathrm{u}_2-\frac{\mathrm{ceu}_1}{\mathrm{y}_1\mathrm{z}})$	$\frac{cu_{1}}{x}(1 + \frac{de}{y_{1}z})$	$-\frac{\operatorname{ceu}_{1}}{x}(1+\frac{1}{z})$	ceu <u>l</u> fylz	$-\frac{\text{cdeu}_1}{\text{fy}_1^2 z}$	bcdeu ₁ fy ² ₁ z
M _{DC}	FM _{DE}	$\frac{\mathrm{bd}}{\mathrm{y_1}^{\mathrm{z}}}(\frac{2\mathrm{x}}{\mathrm{f}\mathrm{y_1}}-1)$	$-\frac{\mathrm{d}}{y_1 z} (\frac{2x}{f y_1} - 1)$	$\frac{1}{z}(\frac{2x}{fy_1}-1)$	$\frac{1}{z}(\frac{x}{fy_1}-2)$	$-\frac{\mathrm{d}}{y_1 z} (\frac{x}{i y_1} - 2)$	$\frac{\text{pd}}{\text{y}_1^2}(\frac{x}{\text{i}\text{y}_1}-2)$
M _{DE}	- FM _{DE}	$-\frac{\mathrm{bd}}{\mathrm{y}_1 \mathrm{z}} (\frac{2\mathrm{x}}{\mathrm{f} \mathrm{y}_1} - 1)$	$\frac{d}{y_1z}(\frac{2x}{fy_1}-1)$	$-\frac{1}{z}(\frac{2x}{iy_1}-1)$	$-\frac{1}{z}(\frac{z}{iy_1}-2)$	$\frac{d}{y_1^2}(\frac{x}{fy_1}-2)$	$-\frac{\mathrm{bd}}{\mathrm{y_{1}^{2}}}(\frac{\mathrm{x}}{\mathrm{i}\mathrm{y_{1}}}-2)$
M _{ED}	FM _{ED}	$-\frac{\mathrm{bd}}{\mathrm{y_1}\mathrm{z}}(\frac{\mathrm{x}}{\mathrm{fy_1}}-2)$	$\frac{1}{y_1^2}(\frac{x}{1y_1}-2)$	$-\frac{1}{2}(\frac{x}{f\overline{y}_1}-2)$	$-\frac{1}{z}(\frac{2x}{iy_1}-1)$	$\frac{d}{y_1 z} (\frac{2x}{1 y_1} - 1)$	$-\frac{\mathrm{bd}}{y_{1}^{2}}(\frac{2x}{ty_{1}}-1)$
MEF	- FM _{ED}	$\frac{\mathrm{bl}}{\mathrm{y}_1}\mathrm{z}(\frac{\mathrm{x}}{\mathrm{r}\mathrm{y}_1}-\mathrm{z})$	$-\frac{\mathrm{d}}{\mathrm{y}_1\mathrm{z}}(\frac{\mathrm{x}}{\mathrm{f}\mathrm{y}_1}-2)$	$\frac{1}{2}(\frac{x}{fy_1}-2)$	$\frac{1}{z}(\frac{2x}{iy_1}-1)$	$-\frac{\mathrm{d}}{y_1^2}(\frac{2x}{iy_1}-1)$	$\frac{\mathrm{bd}}{\mathrm{y_1^z}}(\frac{2\mathrm{x}}{\mathrm{fy_1}}-1)$
M _{FE}	$^{ m FM}_{ m FG}$	$\frac{\frac{bcdeu_1}{fy_1^2z}}$	$-\frac{\text{cdeu}_1}{\text{fy}_1^2 z}$	$\frac{\operatorname{ceu}_{\underline{1}}}{\operatorname{fy}_{\underline{1}}z}$	$-\frac{\operatorname{ceu}_{\frac{1}{2}}}{x}(1+\frac{1}{2})$	$\frac{\mathrm{cu}_{\underline{1}}}{\mathrm{x}}(1+\frac{\mathrm{de}}{\mathrm{y}_{\underline{1}}\mathrm{z}})$	$\frac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_2 - \frac{\mathrm{ceu}_1}{\mathrm{y}_1\mathrm{z}})$
M _{FG}	- FM _{FG}	$-\frac{bcdeu_1}{fy_1^2z}$	$\frac{\frac{\text{cdeu}_1}{\text{fy}_1^2 z}}$	$-\frac{\text{ceu}_{\underline{1}}}{\text{fy}_{\underline{1}}\overline{z}}$	$\frac{\operatorname{ceu}_{\underline{1}}}{x}(1+\frac{1}{z})$	$-\frac{cu_1}{x}(1+\frac{de}{y_1z})$	$-\frac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_2-\frac{\mathrm{ceu}_1}{\mathrm{y}_1\mathrm{z}})$
M _{GF}	FM _{GH}	$-\frac{2abcde}{fy_1^2z}$	2aced fy ² ₁ z	$-\frac{2ace}{fy_1z}$	$\frac{2ace}{x}(1+\frac{1}{z})$	$-\frac{2ac}{x}(1+\frac{de}{y_1z})$	$\frac{2a}{x}(y_2 + \frac{bcde}{y_1z})$
M _{GH}	- FM _{GH}	2abcde fy12	- 2aced fy2z	2°ce fy ₁ z	$-\frac{2ace}{x}(1+\frac{1}{z})$	$\frac{2ac}{x}(1+\frac{de}{y_1z})$	$-\frac{2a}{x}(y_2 + \frac{bcde}{y_1^2})$
M _{HG}	FM _{HG}	abcde fy12	$-\frac{\text{aced}}{\text{fy}_1^2 z}$	ace fy ₁ z	$-\frac{ace}{x}(1+\frac{1}{z})$	$\frac{ac}{x}(1 + \frac{de}{y_1z})$	$-\frac{a}{x}(y_2 + \frac{bcde}{y_1z})$

Beam with ends built in and freely supported at six points. Supports unequally spaced.



$$K_1 = \frac{I}{L_1}$$
 $K_2 = \frac{I}{L_2}$ $K_3 = \frac{I}{L_3}$ $K_4 = \frac{I}{L_4}$ $K_5 = \frac{I}{L_5}$ $K_6 = \frac{I}{L_6}$ $K_7 = \frac{I}{L_7}$ $X_1 = 1 - bc - de$ $X_2 = 1 - bj - km$

$$a = \frac{K_1}{2(K_1 + K_2)} \quad b = \frac{K_2}{2(K_1 + K_2)} \quad c = \frac{K_2}{2(K_2 + K_3)} \quad d = \frac{K_3}{2(K_2 + K_3)} \quad y_3 = 1 - hj \quad y_4 = 1 - km$$

$$e = \frac{K_3}{2(K_3 + K_4)}$$
 $f = \frac{K_4}{2(K_3 + K_4)}$ $g = \frac{K_4}{2(K_4 + K_5)}$ $h = \frac{K_5}{2(K_4 + K_5)}$ $u_3 = 2 - h$ $u_4 = 2 - m$

$$j = \frac{K_5}{2(K_5 + K_6)} \quad k = \frac{K_6}{2(K_5 + K_6)} \quad m = \frac{K_6}{2(K_6 + K_7)} \quad n = \frac{K_7}{2(K_6 + K_7)}$$

$$z = \frac{x_1 x_2}{fgy_1 y_4} - 1$$

$$x_1 = 1 - bc - de$$

$$x_2 = 1 - hj - kn$$

$$y_1 = 1 - bc$$
 $y_2 = 1 - de$

$$y_3 = 1 - hj$$

$$y_{i} = 1 - km$$

$$u_1 = 2 - b$$
 $u_2 = 2 - e$

$$u_2 = 2 - e$$

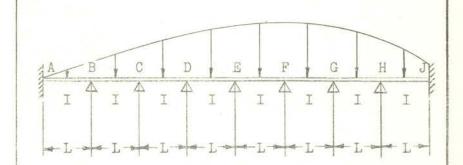
$$u_3 = 2 - h$$

$$u_4 = 2 - m$$

$$z = \frac{x_1 x_2}{f g y_1 y_1} - 1$$

Moment	Fixed End Moment	FM _{BA} - FN _{BC}	Mi _{CB} - Mi _{CD}	Fi _{DC} - FM _{DE}	FM _{ED} - FM _{EF}	FM _{FE} - FM _{FG}	FM _{GF} - FM _{GH}
MAB	- FMAB	$-\frac{a}{x_1}(y_2 + \frac{bcde}{y_1^2})$	$\frac{\operatorname{ac}}{x_1}(1+\frac{\operatorname{de}}{y_1z})$	$-\frac{ace}{x_1}(1+\frac{1}{2})$	ace fy ₁ z	- acej fy ₁ y ₄ z	acejm fy ₁ y ₄ z
MBA	FM _{BA}	$-\frac{2a}{x_1}(y_2 + \frac{bcde}{y_1^2})$	$\frac{2ac}{x_1}(1+\frac{de}{y_1z})$	$-\frac{2ace}{x_1}(1+\frac{1}{x})$	2ace fy ₁ z	- Zacej fyly4z	2acejm Tyjy,z
M _{BC}	- FM _{BA}	$\frac{2a}{x_1}(y_2 + \frac{bcde}{y_1^2})$	$-\frac{2ac}{x_1}(1+\frac{de}{y_1z})$	$\frac{2ace}{x_1}(1+\frac{1}{z})$	- 2ace	Zacej Ty ₁ y ₄ z	$-\frac{2acejn}{fy_1y_4^z}$
Исв	FM _{CB}	$-\frac{\mathrm{bd}}{\mathrm{x_1}}(\mathrm{u_2}-\frac{\mathrm{ceu_1}}{\mathrm{y_1^z}})$	$-\frac{\operatorname{cu}_{1}}{\operatorname{x}_{1}}(1+\frac{\operatorname{de}}{\operatorname{y}_{1}z})$	$\frac{\text{ceu}_{\underline{1}}}{x_{\underline{1}}}(1+\frac{\underline{1}}{z})$	- coul	ceju _l	$-\frac{\text{cejmu}_{\underline{1}}}{fy_{\underline{1}}y_{\underline{4}}\overline{z}}$
11 _{CD}	- Fi _{CB}	$\frac{\mathrm{pq}}{\mathrm{x^1}}(\mathrm{n^3}-\frac{\mathrm{cen^1}}{\mathrm{\lambda^1}\mathrm{z}})$	$\frac{\mathrm{cu}_{\underline{1}}}{\mathrm{x}_{\underline{1}}}(1+\frac{\mathrm{de}}{\mathrm{y}_{\underline{1}}\mathrm{z}})$	$-\frac{\operatorname{ceu}_{\underline{1}}}{x_{\underline{1}}}(1+\frac{1}{z})$	$\frac{\mathtt{ceu}_{\underline{1}}}{\mathtt{fy}_{\underline{1}} \mathtt{z}}$	$-\frac{\text{ceju}_{\underline{1}}}{\text{fy}_{\underline{1}}\text{y}_{\underline{4}}\overline{z}}$	$\frac{\text{cejrm}_1}{f y_1 y_4^z}$
MDC	FMDE	$\frac{\operatorname{bd}}{\operatorname{y}_1^{\mathbf{z}}}(\frac{2\operatorname{x}_2}{\operatorname{gy}_4}-1)$	$-\frac{\mathrm{d}}{y_1 z} (\frac{2x_2}{\mathrm{gy}_4} - 1)$	$\frac{1}{2}(\frac{2x_2}{gy_4}-1)$	$\frac{1}{2}(\frac{x_1}{fy_1}-2)$	$-\frac{1}{y_{\perp}^{\bullet}z}(\frac{x_{1}}{iy_{1}}-2)$	$\frac{jm}{y_{\underline{A}}z}(\frac{x_{\underline{1}}}{\overline{t}y_{\underline{1}}}-2)$
MDE	- Fi _{DE}	$-\frac{\mathrm{bd}}{y_{1}z}(\frac{2x_{2}}{\mathrm{g}y_{4}}-1)$	$\frac{\mathrm{d}}{y_1 z} (\frac{2 x_2}{g y_4} - 1)$	$-\frac{1}{z}(\frac{2x_2}{gy_4}-1)$	$-\frac{1}{z}(\frac{x_1}{f\overline{y}_1}-2)$	$\frac{1}{y_4^2}(\frac{x_1}{ty_1}-2)$	$-\frac{jm}{y_{4}z}(\frac{x_{1}}{Ty_{1}}-2)$
MED	FM _{ED}	$-\frac{\mathrm{bd}}{\mathrm{y_1^2}}(\frac{\mathrm{x_2}}{\mathrm{gy_4}}-2)$	$\frac{\mathrm{d}}{y_1^z}(\frac{x_2}{gy_4}-2)$	$-\frac{1}{z}(\frac{x_2}{gy_4}-2)$	$-\frac{1}{2}(\frac{2x_1}{7y_1}-1)$	$\frac{1}{y_{\lambda}^{2}}(\frac{2x_{1}}{1y_{1}}-1)$	$-\frac{jm}{y_{\underline{A}}z}(\frac{2x_{\underline{1}}}{\overline{i}y_{\underline{1}}}-1)$
MEF	- Fig	$\frac{\text{bd}}{\text{$y_1$}^2}(\frac{\text{$x_2$}}{\text{$gy_4$}}-2)$	$-\frac{\mathrm{d}}{y_1 z} (\frac{x_2}{\mathrm{g} y_4} - 2)$	$\frac{1}{z}(\frac{x_2}{\epsilon y_4} - 2)$	$\frac{1}{z}(\frac{2x_1}{fy_1}-1)$	$-\frac{1}{y_{4}z}(\frac{2x_{1}}{1y_{1}}-1)$	$\frac{jm}{y_4^2}(\frac{2x_1}{Ty_1}-1)$
MFE	FK _{FG}	bdhku ₄	$-\frac{\mathrm{dhku}_{\lambda}}{\mathrm{gy}_{1}\mathrm{y}_{\lambda}^{\mathrm{z}}}$	hku ₄ Ey ₄ ²	$-\frac{hku_{\underline{4}}}{\kappa_{\underline{2}}}(1+\frac{1}{2})$	$\frac{ku_{4}}{x_{2}}(1+\frac{hj}{y_{4}z})$	$\frac{jn}{x_2}(u_3 - \frac{hku_4}{y_4^z})$
M_{FG}	- Fit _{FG}	$-\frac{\mathrm{bdhku}_{\lambda}}{\mathrm{gy}_{1}\mathrm{y}_{\lambda}^{\mathrm{z}}}$	$\frac{\overline{\mathrm{gy_1}}\overline{\mathrm{y_4}^2}}{\mathrm{dhku_4}}$	$-\frac{hku_{4}}{gy_{4}^{2}}$	$\frac{hku_{4}}{x_{2}}(1+\frac{1}{2})$	$-\frac{\mathrm{ku}_{4}}{\mathrm{x}_{2}}(1+\frac{\mathrm{hj}}{\mathrm{y}_{4}\mathbf{z}})$	$-\frac{jm}{x_2}(u_3 - \frac{hku_2}{y_4z})$
M _{GF}	FNGH	- 2bdhlen gy ₁ y ₄ z	2dhkn gy _l y ₄ z	- 2hkm 5742	$\frac{2hlm}{x_2}(1+\frac{1}{z})$	$-\frac{2\mathrm{kn}}{x_2}(1+\frac{\mathrm{h}_1^*}{y_4^2})$	$\frac{2n}{x_2}(y_3 + \frac{hfkm}{y_4^2})$
M _{GH}	- FM _{GH}	2bdhkn gy ₁ y ₄ z	$-\frac{2\mathrm{dhlcn}}{\mathrm{gy}_1\mathrm{y}_4\mathrm{z}}$	2hkn gy ₄ ²	$-\frac{2hkn}{x_2}(1+\frac{1}{z})$	$\frac{2kn}{x_2}(1+\frac{hj}{y_4z})$	$-\frac{2n}{x_2}(x_3 + \frac{h_1 km}{y_4 z})$
11HG	FM _{HG}	bdhkm gy ₁ y ₄ z	$-\frac{\mathrm{dhlm}}{\mathrm{gy_1y_4}^2}$	hkn gy ₄ z	$-\frac{\mathrm{hkm}}{\mathrm{x}_2}(1+\frac{1}{2})$	$\frac{kn}{x_2}(1+\frac{hj}{y_{\ell}z})$	$-\frac{n}{x_2}(y_3 + \frac{hjlm}{y_4z})$

Beam with ends built in and freely supported at seven points. Supports equally spaced.



$$K = \frac{I}{L}$$

$$a = \frac{1}{4}$$

$$x = \frac{7}{3}$$

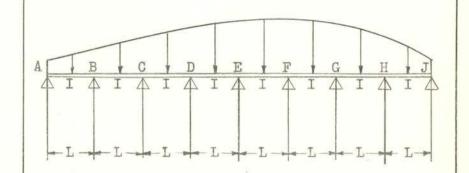
$$y = \frac{15}{16}$$

$$u = \frac{7}{4}$$

$$z = \frac{194}{15}$$

Moment	Fixed End	EM EM	EM EM	EM EM	EM EM	PM PM	EM EM	EM EM
roment	Moment		FMCB - FMCD	FM _{DC} - FM _{DE}	rmED - rmEF	rmFE - rmFG	FM _{GF} - FM _{GH}	rmHG - rmH.
MAB	- FM _{AB}	2681	.0718	0192	.0052	0014	.0004	0001
M_{BA}	FM_{BA}	5362	.1436	0385	.0103	0028	.0007	0002
M_{BC}	- FM _{BA}	.5362	1436	.0385	0103	.0028	0007	.0002
McB	FM _{CB}	1243	5022	.1346	0351	.0097	0026	.0007
McD	- FM _{CB}	.1243	.5022	1346	.0351	0097	.0026	0007
M_{DG}	$FM_{ m DE}$	-0333	1334	-5Q00	.1340	0359	.0096	0024
$M_{ m DE}$	- FM _{DE}	= .0333	-1334	5000	1340	.0359	0096	.0024
MED	FMED	0089	.0357	1340	5000	.1340	0357	.0089
MEF	- FMEF	.0089	0357	.1340	5000	1340	.0357	0089
MFE	FMFE	,0024	0096	.0359	1340	5000	-1334	0333
M_{FG}	- FMFE	0024	.0096	0359	.1340	- 5000	1334	-0333
M_{GF}	FMGH	0007	.0026	0097	.0351	1346	.5022	.1243
M_{GH}	- FM _{GH}	.0007	0026	.0097	0351	.1346	5022	1243
$^{\rm M}{ m HG}$	FM _{HJ}	.0002	0007	.0028	0103	.0385	1436	.5362
M_{HJ}	- FM _{HJ}	0002	.0007	0028	.0103	0385	.1436	5362
$^{\mathrm{M}}$ JH	FMJH	0001	.0004	0014	.0052	0192	.0718	2681

Beam freely supported at nine points. Supports equally spaced.



PROPPED BEAM END MOMENTS

CONSTANT FUNCTIONS OF NDF

$$EM_{BA}^* = \frac{FM_{AB}}{2} + FM_{BA}$$

$$EM_{HJ}^* = FM_{HJ} + \frac{FM_{JH}}{2}$$

$$K = \frac{I}{L}$$

$$K^{\dagger} = \frac{3K}{4L}$$

$$a = \frac{13}{14}$$

$$b = \frac{2}{5}$$

$$a = \frac{13}{14}$$
 $b = \frac{2}{7}$ $c = d = e = f = g = \frac{1}{4}$

$$x = \frac{97}{112}$$
 $y_1 = \frac{13}{14}$ $y_2 = \frac{15}{16}$

$$y_1 = \frac{13}{14}$$

$$y_2 = \frac{15}{16}$$

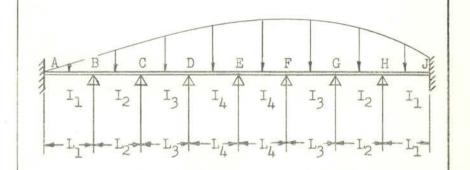
$$z = \frac{168}{13}$$
 $u_1 = \frac{12}{7}$ $u_2 = \frac{7}{4}$

$$u_2 = \frac{7}{4}$$

TABLE OF FINAL MOMENTS Fixed End $\text{EM}_{\text{BA}}^* - \text{FM}_{\text{BC}} | \text{FM}_{\text{CB}} - \text{FM}_{\text{CD}} | \text{FM}_{\text{DC}} - \text{FM}_{\text{DE}} | \text{FM}_{\text{ED}} - \text{FM}_{\text{EF}} |$ Moment FM_{FE} - FM_{FG} FMGF - FMGH Moment EMBA - .4641 .1244 - .0333 .0089 - .0023 - .0002 .0006 MBA - EMBA MBC .4641 - .1244 .0333 - .0089 .0006 .0023 .0002 MCB FMCB - .1436 - .4973 .1333 - .0357 .0096 - .0026 .0007 - FM_{CB} $^{\rm M}_{ m CD}$.1436 .4973 - .1333 .0357 - .0096 - .0007 .0026 MDC FMDE .0385 - .1348 .5002 .1339 - .0356 .0097 M_{DE} - .0385 - FM_{DE} .1348 - .5002 - .I.339 .0356 - .0097 .0028 FM_{ED} MED - .0103 .0361 - .1337 - .5000 .1337 - .0361 .0103 - FMEF MEF .0103 - .0361 .1337 - .1337 - .0103 - .5000 .0361 $\mathrm{FM}_{\mathrm{FE}}$.0028 MFE .0356 - .1339 - .5002 .1348 - .0385 - FM M_{FG} - .0028 - .0356 .1339 .0097 .5002 - .1348 .0385 FMGH MGF - .0007 .0026 - .0096 .0357 - .1333 .4973 .1436 ∞ FM_{GH} M_{GH} - .0026 .0007 .0096 - .0357 - .4973 .1333 - .1436 EM HJ MHG - .0006 .0002 .0023 - .0089 .0333 - .1244 .4641 - EMHJ MHJ - .0002 .0006 - .0023 - .4641 .0089 - .0333 .1244

DESCRIPTION:

Beam with ends built in and freely supported at seven points. Supports symmetrically spaced.



CONSTANT FUNCTIONS OF NEW DISTRIBUTION FACTOR

$$K_{1} = \frac{I}{L_{1}} \qquad K_{2} = \frac{I}{L_{2}} \qquad K_{3} = \frac{I}{L_{3}} \qquad K_{4} = \frac{I}{L_{4}} \qquad \qquad x = 1 - bc - de \qquad y_{1} = 1 - bc$$

$$a = \frac{K_{1}}{2(K_{1} + K_{2})} \qquad b = \frac{K_{2}}{2(K_{1} + K_{2})} \qquad c = \frac{K_{2}}{2(K_{2} + K_{3})} \qquad u_{1} = 2 - b \qquad y_{2} = 1 - de$$

$$d = \frac{K_{3}}{2(K_{2} + K_{3})} \qquad e = \frac{K_{3}}{2(K_{3} + K_{4})} \qquad f = \frac{K_{4}}{2(K_{3} + K_{4})} \qquad u_{2} = 2 - e$$

$$g = \frac{K_{4}}{2(K_{4} + K_{4})} = \frac{1}{4} \qquad u_{3} = \frac{7}{4}$$

$$u_{3} = \frac{7}{4}$$

$$x = 1 - bc - de$$
 $y_1 = 1 - bc$

$$y_2 = 1 - de$$

$$u_2 = 2 - e$$

$$z = 4 \frac{x}{fy_1} - 2$$

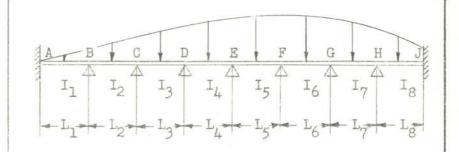
TABLE OF FINAL MOMENTS

l'oment	Fixed End Moment	HL _{BA} - FM _{BC}	FM _{CD} - FM _{CD}	FM _{DC} - FM _{DE}	FM _{ED} - FM _{EF}	Fi _{FE} - Fi _{FG}	FM _{GF} - FM _{GH}	Fig - Fig
N _{AB}	- FM _{AB}	$-\frac{a}{x}(y_2 + \frac{bcde}{y_1^2})$	$\frac{\operatorname{ac}}{\operatorname{x}}(1+\frac{\operatorname{de}}{\operatorname{y}_{1}^{2}})$	$-\frac{ace}{x}(1+\frac{1}{z})$	ace Ty ₁ z	- ace	acde xy12	- abode xy ₁ z
МВЛ	FH _{BA}	$-\frac{2a}{x}(y_2 + \frac{bode}{y_1^z})$	$\frac{2ac}{x}(1+\frac{de}{y_1^2})$	$-\frac{2ace}{x}(1+\frac{1}{z})$	2ace fy ₁ z	_ 2ace	2acde xy ₁ z	- Zabode Xyl²
H _{BC}	- Fil	$\frac{2a}{x}(y_2 + \frac{bcde}{y_1^2})$	$-\frac{2ac}{x}(1+\frac{de}{y_1z})$	$\frac{2ace}{x}(1+\frac{1}{z})$	- Zace	2ace XZ	$-\frac{2acde}{xy_1z}$	+ 2abcde xy ₁ z
и _{св}	Mice	$-\frac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_{2}-\frac{\mathrm{ceu}_{1}}{\mathrm{y}_{1}^{\mathrm{x}}})$	$-\frac{\mathrm{cu}_{1}}{x}(1+\frac{\mathrm{de}}{y_{1}z})$	$\frac{\operatorname{ceu}_1}{\pi}(1+\frac{1}{2})$	- ceul fylz	ceu ₁	- cdeu ₁	bcdeu ₁
M _{CD}	- Fi _{GB}	$\tfrac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_2-\tfrac{\mathrm{ceu}_1}{\mathrm{y}_1\mathrm{z}})$	$\frac{\mathrm{cu}_{1}}{\mathrm{x}}(1+\frac{\mathrm{de}}{\mathrm{y}_{1}\mathrm{z}})$	$-\frac{\operatorname{ceu}_1}{x}(1+\frac{1}{z})$	ceu _l fy _l z	- ceul	odeu <u>n</u>	- bcdeu ₁
I ^C DC	PLE	$\frac{bd}{y_1^2}(7-\frac{2fy_1}{x})$	$-\frac{\mathrm{d}}{y_1 z} (7 - \frac{2fy_1}{x})$	$\frac{1}{2}(7 - \frac{2fy_1}{x})$	$\frac{1}{z} \left(\frac{x}{1y_1} - 2 \right)$	$-\frac{1}{2}(1-\frac{2fy_1}{x})$	$\frac{\mathrm{d}}{y_1^2}(1-\frac{2\mathrm{f}y_1}{x})$	$-\frac{\mathrm{bd}}{y_1^2}(1-\frac{2\mathrm{f}y_1}{x})$
N _{DE}	- M _{JE}	$-\frac{\mathrm{bd}}{\mathrm{y}_{\underline{1}}\mathrm{z}}(7-\frac{\mathrm{2fy}_{\underline{1}}}{\mathrm{x}})$	$\frac{\mathrm{d}}{y_1^2}(7-\frac{2\mathrm{f}y_1}{x})$	$-\frac{1}{3}(7-\frac{2fy_1}{x})$	$-\frac{1}{2}(\frac{x}{iy_1}-2)$	$\frac{1}{2}(1-\frac{2fy_1}{x})$	$-\frac{\mathrm{d}}{y_{\underline{1}}z}(1-\frac{2\mathrm{f}y_{\underline{1}}}{x})$	$\frac{\mathrm{bd}}{\overline{y_1}\overline{z}}(1-\frac{2\mathrm{f}y_1}{\overline{x}})$
H_D	Mind	- bdf 2x	or Zz	- 27 ₁	500	fy ₁	- df - 2x	bdf 2x
II.	- Fil	bdf 2x	$-\frac{df}{2x}$	fy ₁	500	$-\frac{ry_1}{2x}$	df 2x	- bdf 2x
1(MI_	$\frac{\mathrm{bd}}{\mathrm{y}_{\underline{1}}\mathrm{z}}(1-\frac{2\mathrm{f}\mathrm{y}_{\underline{1}}}{\mathrm{x}})$	$-\frac{d}{y_1^2}(1-\frac{2fy_1}{x})$	$\frac{1}{2}(1-\frac{2fy_1}{x})$	$-\frac{1}{z}\left(\frac{x}{iy_1}-2\right)$	$-\frac{1}{2}(7-\frac{2fy_1}{x})$	$\frac{d}{y_{\underline{1}}z}(7-\frac{2fy_{\underline{1}}}{x})$	$-\frac{\mathrm{bd}}{y_{\underline{1}}z}(7-\frac{2\mathbf{\ell}y_{\underline{1}}}{x})$
II _{2G}	- 7M ₇₀	$-\frac{\mathrm{bd}}{\mathrm{y_1^2}}(1-\frac{2\mathrm{fy_1}}{\mathrm{x}})$	$\frac{\tilde{q}}{\tilde{y}_1^2}(1-\frac{2\tilde{q}_1}{\pi})$	$-\frac{1}{2}(1-\frac{2fy_1}{x})$	$\frac{1}{2}\left(\frac{x}{1y_1}-2\right)$	$\frac{1}{2}(7-\frac{2fy_1}{x})$	$-\frac{d}{y_1^2}(7-\frac{2fy_1}{x})$	$\frac{\operatorname{bd}}{\mathcal{I}_{1}^{2}}(7-\frac{2\operatorname{fy}_{1}}{x})$
II _{GE}	TA CH	- bedou	cdeu ₁	- <u>xx</u>	ceu _l	$-\frac{\operatorname{ceu}_{\underline{1}}}{x}(1+\frac{1}{2})$	$\frac{cu_1}{x}(1+\frac{de}{y_1^2})$	$\frac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_2 - \frac{\mathrm{ceu}_1}{\mathrm{J}_1\mathrm{z}})$
li _{GH}	- PN _{OH}	bodeu <u>1</u>	- cdeul	xs cen ³	- ceul fylz	$\frac{\operatorname{ceu}_{\underline{1}}}{x}(1+\frac{1}{z})$	$-\frac{\operatorname{cu}_1}{\operatorname{x}}(1+\frac{\operatorname{de}}{\operatorname{y}_1\operatorname{z}})$	$-\frac{\mathrm{bd}}{\mathrm{x}}(\mathrm{u}_2-\frac{\mathrm{ceu}_2}{\mathrm{y}_1\mathrm{z}})$
15 _{ii} g	Mt _{liJ}	2abcde	- Zaode	2406 102	- 2ace fyl²	$\frac{2ace}{x}(1+\frac{1}{2})$	$-\frac{2ac}{x}(1+\frac{de}{y_1z})$	$\frac{2a}{x}(y_2 + \frac{bcde}{y_1 x})$
I' _H J	- fi	- Zabode	2acde zy ₁ z	- 2aca	2ace fy ₁ ²	$-\frac{2ace}{x}(1+\frac{1}{z})$	$\frac{2ac}{x}(1+\frac{de}{y_1z})$	$-\frac{2a}{x}(y_2 + \frac{bcde}{y_1^2})$
нлі	FN _{JII}	- abcde	aode	- <u>a.co</u>	ace fy ₁ z	$-\frac{\mathrm{ace}}{x}(1+\frac{1}{2})$	$\frac{\text{ac}}{x}(1 + \frac{\text{de}}{y_1 z})$	$-\frac{a}{x}(y_2 + \frac{bode}{y_1^2})$

Engineering Experiment Station

DESCRIPTION:

Beam with ends built in and freely supported at seven points. Supports unequally spaced.



CONSTANT FUNCTIONS OF NEW DISTRIBUTION FACTOR

$$K_{1} = \frac{1}{L_{1}} K_{2} = \frac{1}{L_{2}} K_{3} = \frac{1}{L_{3}} K_{4} = \frac{1}{L_{4}} K_{5} = \frac{1}{L_{5}} K_{6} = \frac{1}{L_{6}} K_{7} = \frac{1}{L_{7}} K_{8} = \frac{1}{L_{8}}$$

$$a = \frac{K_{1}}{2(K_{1} + K_{2})} b = \frac{K_{2}}{2(K_{1} + K_{2})} c = \frac{K_{2}}{2(K_{2} + K_{3})} d = \frac{K_{3}}{2(K_{2} + K_{3})}$$

$$e = \frac{K_{3}}{2(K_{3} + K_{4})} f = \frac{K_{4}}{2(K_{3} + K_{4})} g = \frac{K_{4}}{2(K_{4} + K_{5})} h = \frac{K_{5}}{2(K_{4} + K_{5})}$$

$$j = \frac{K_{5}}{2(K_{5} + K_{6})} k = \frac{K_{6}}{2(K_{5} + K_{6})} m = \frac{K_{6}}{2(K_{6} + K_{7})} n = \frac{K_{7}}{2(K_{6} + K_{7})}$$

$$p = \frac{K_{7}}{2(K_{7} + K_{8})} r = \frac{K_{8}}{2(K_{7} + K_{8})}$$

$$x_1 = 1 - bc - de$$
 $u_1 = 2 - b$
 $x_2 = 1 - km - np$ $u_2 = 2 - e$
 $y_1 = 1 - bc$ $u_3 = 2 - g$
 $y_2 = 1 - de$ $u_4 = 2 - h$
 $y_3 = 1 - km$ $u_5 = 2 - k$
 $y_4 = 1 - np$ $u_6 = 2 - p$
 $z_1 = \frac{x_1}{fgy_1} - \frac{hjx_1y_4}{fgx_2y_1} - 1$ $z_2 = \frac{x_2}{hjy_4} - \frac{fgx_2y_1}{hjx_1y_4} 1$

TABLE OF FINAL MOMENTS

	TABLE OF FINAL MODERIES							
Moment	Fixed End Moment	r: _{ba} - ri _{bc}	Mi _{CB} - Mi _{CD}	M _{DC} - M _{DE}	ML _D - R _{EF}	M _{PE} - Mi _{PG}	M _{GF} - M _{GH}	M _{HG} - M _{HJ}
M _{AB}	- FM _{AB}	$-\frac{a}{x_1}(y_2 + \frac{bode}{y_1x_1})$	$\frac{\operatorname{ac}}{z_1}(1+\frac{\operatorname{de}}{y_1^{2}1})$	$-\frac{\operatorname{nce}}{x_1}(1+\frac{1}{x_1})$	ace fy ₁ z ₁	$-\frac{\text{acejy}_{4}}{\text{fx}_{2}\text{y}_{1}^{2}\text{1}}$	acejm Tx ₂ y ₁ z ₁	- acejmp fx ₂ y ₁ z ₁
M _{BA}	FM _{BA}	$-\frac{2a}{x_1}(y_2 + \frac{bcde}{y_1z_1})$	$\frac{2ac}{x_1}(1+\frac{de}{y_1z_1})$	$-\frac{2ace}{x_1}(1+\frac{1}{x_1})$	2ace . Ty ₁ z ₁	$-\frac{2acejy_{\underline{\ell}}}{fx_2y_1z_1}$	Zaceju N:2 ^y 1 ^z 1	- 2acejmp fx ₂ y ₁ z ₁
M _{BC}	- FM _{BA}	$\frac{2a}{x_1}(y_2 + \frac{bcde}{y_1 z_1})$	$-\frac{2ac}{x_1}(1+\frac{de}{y_1z_1})$	$\frac{2ace}{z_1}(1+\frac{1}{z_1})$	- 2ace 1y121	2acejy/	$-\frac{2acejn}{m_2y_1z_1}$	2acejmp Tx ₂ y ₁ z ₁
M _{CB}	PM _{CB}	$-\frac{\mathrm{bd}}{\mathrm{x_1}}(\mathrm{u_2}-\frac{\mathrm{ceu_1}}{\mathrm{y_1}^\mathrm{z_1}})$	$-\frac{cu_{1}}{x_{1}}(1+\frac{de}{y_{1}x_{1}})$	$\frac{\operatorname{ceu}_1}{x_1}(1+\frac{1}{z_1})$	- ccu1	ceju ₁ y ₄ fx ₂ y ₁ z ₁	- cojmin	oejmpu <u>l</u> Tr _Z y ₁ z ₁
M _{CD}	- FM _{CB}	$\frac{\operatorname{bd}}{\operatorname{x}_1}(\operatorname{u}_2-\frac{\operatorname{ceu}_1}{\operatorname{y}_1\operatorname{x}_1})$	$\frac{cu_{1}}{x_{1}}(1+\frac{de}{y_{1}z_{1}})$	$-\frac{\text{ceu}_{1}}{x_{1}}(1+\frac{1}{z_{1}})$	ceu ₁ Ty ₁ z ₁	$-\frac{\operatorname{ceju}_1 y_{\underline{\zeta}}}{\operatorname{fx}_2 y_1 z_1}$	cojmu <u>i</u> Tx ₂ y ₁ z ₁	- cojmpu <u>l</u> - fx ₂ y ₁ z ₁
M _{DC}	FM _{DE}	$\frac{\operatorname{bd}}{\operatorname{CT}_1^{\mathbf{Z}_1}}(\operatorname{u}_3 - \frac{\operatorname{2hjy}_L}{\operatorname{x}_2})$	$-\frac{\mathrm{d}}{\mathrm{GV_1}^2\mathrm{1}}(\mathrm{u_3}-\frac{\mathrm{2hjy_4}}{\mathrm{x_2}})$	$\frac{1}{g^2_1}(u_3 - \frac{2hjy_4}{x_2})$	$\frac{1}{z_1}(\frac{x_1}{iy_1}-2)$	$-\frac{jy_{\underline{i}}}{x_{\underline{2}}z_{\underline{1}}}(\frac{x_{\underline{1}}}{iy_{\underline{1}}}-2)$	$\frac{j^m}{x_2^2 z_1} (\frac{x_1}{f y_1} - 2)$	$-\frac{\text{jmp}}{x_2^2 1}(\frac{x_1}{\text{fy}_1}-2)$
MDE	- FM _{DE}	$-\frac{\mathrm{bd}}{\mathrm{gy_1}^2\mathrm{l}}(\mathrm{u_3}-\frac{\mathrm{2hjy_4}}{\mathrm{x_2}})$	$\frac{\mathrm{d}}{\mathrm{GV_1}^2\mathrm{1}}(\mathrm{u_3}-\frac{\mathrm{2hjy_4}}{\mathrm{x_2}})$	$-\frac{1}{6^{2}1}(u_{3}-\frac{2hjy_{4}}{x_{2}})$	$-\frac{1}{z_1}(\frac{x_1}{Iy_1}-2)$	$\frac{Jy_4}{x_2^2}(\frac{x_1}{Iy_1}-2)$	$-\frac{\sin}{x_2 z_1} (\frac{x_1}{i y_1} - 2)$	$\frac{\text{jmp}}{x_2^2 1} \left(\frac{x_1}{\text{Iy}_1} - 2 \right)$
MED	FMED	$-\frac{\mathrm{bdf}}{\mathrm{x_1}^{\mathrm{z}}2}(\frac{2\mathrm{x}_2}{\mathrm{jy}_4}-1)$	$\frac{df}{{}^{2}1^{2}2}(\frac{2x}{jy_{4}}-1)$	$-\frac{fy_1}{x_1x_2}(\frac{2x_2}{3y_4}-1)$	$-\frac{1}{x_1}(\frac{2x_1}{1y_1}-1)$	$\frac{\mathrm{jy}_{4}}{\mathrm{x}_{2}\mathrm{z}_{1}}(\frac{\mathrm{2x}_{1}}{\mathrm{iy}_{1}}-1)$	$-\frac{\ln x_2 x_1}{\ln x_2 x_1} (\frac{2x_1}{\ln x_1} - 1)$	$\frac{\sin x_2 x_1}{x_2 x_1} (\frac{2x_1}{x_1} - 1)$
MEF	- PM _{EF}	$\frac{\text{bdf}}{x_1^2}(\frac{2x_2}{3y_4}-1)$	$-\frac{df}{x_1 x_2} (\frac{2x_2}{jy_4} - 1)$	$\frac{fy_{1}}{x_{1}x_{2}}(\frac{2x_{2}}{Jy_{4}}-1)$	$-\frac{1}{z_2}(\frac{2x_2}{y_4}-1)$	$-\frac{3y_4}{x_2x_1}(\frac{2x_1}{6y_1}-1)$	$\frac{jn}{z_2^2z_1}(\frac{2z_1}{ry_1}-1)$	$-\frac{jmp}{x_2^{-2}1}(\frac{2x_1}{fy_1}-1)$
MFE	FM _{FE}	$\frac{\text{bdf}}{x_1^2}(\frac{x_2}{3y_4}-2)$	$-\frac{df}{x_1^2}(\frac{x_2}{3y_4}-2)$	$\frac{xy_1}{x_1x_2}(\frac{x_2}{3y_4} - 2)$	$-\frac{1}{z_2}(\frac{x_2}{3y_4}-2)$	$-\frac{1}{hz_2}(u_{i_0}-\frac{z_fgy_1}{z_1})$	$\frac{n}{hy_4 z_2} (u_4 - \frac{2f(y_1)}{x_1})$	$-\frac{\operatorname{rsp}}{\operatorname{hy_4^22}}(\operatorname{u_4}-\frac{\operatorname{2fgy_1}}{\operatorname{x_1}})$
M _{FG}	- FM _{FE}	$-\frac{bdf}{x_1x_2}(\frac{x_2}{3y_4}-2)$	$\frac{df}{x_1^2}(\frac{x_2}{y_4}-2)$	$-\frac{\text{fy}_1}{x_1^x_2}(\frac{x_2}{\text{jy}_4}-2)$	$\frac{1}{z_2}(\frac{x_2}{y_4}-2)$	1 (u ₄ - 2fg/1)	$-\frac{n}{hy_4^2}(u_4-\frac{2f_8y_1}{x_1})$	$\frac{np}{hy_4^2z_2}(u_4 - \frac{2fgy_1}{x_1})$
MGF	FM _{GH}	- bdfknu ₆ - 3x ₁ y ₄ z ₂	drknu ₆ Jz ₁ y ₄ z ₂	$-\frac{\operatorname{fknu}_6 y_1}{\operatorname{jx}_1 y_4 z_2}$	kmu ₆	$-\frac{\ln n_6}{\kappa_2}(1+\frac{1}{\kappa_2})$	$\frac{nu_6}{x_2}(1+\frac{km}{y_4x_2})$	$\frac{mp}{x_2}(u_5 - \frac{knu_6}{y_4^2})$
MCH	- FM _{GH}	bdfmu ₆ jx ₁ y ₄ z ₂	$-\frac{\text{dflanu}_{6}}{\text{jx}_{1}\text{y}_{4}\text{z}_{2}}$	fimu ₆ y ₁ jx ₁ y ₄ z ₂	$-\frac{\mathrm{knu}_{6}}{\mathrm{jy}_{4}z_{2}}$	$\frac{\ln u_6}{x_2}(1+\frac{1}{z_2})$	$-\frac{nu_{6}}{x_{2}}(1+\frac{km}{y_{4}x_{2}})$	$-\frac{mp}{x_2}(u_5 - \frac{knu_6}{y_4 z_2})$
M _{HG}	FM _{HJ}	2bdfknr J× ₁ y ₄ z ₂	- 2dffmr - jx ₁ y ₄ z ₂	2fknry ₁ Jx ₁ y ₄ z ₂	$-\frac{2\mathrm{knr}}{\mathrm{jy_{4}^{z_{2}}}}$	$\frac{2lmr}{\pi_2}(1+\frac{1}{\pi_2})$	$-\frac{2nr}{x_2}(1+\frac{lm}{y_4 z_2})$	$\frac{2\mathbf{r}}{\mathbf{x}_2}(\mathbf{y}_3 + \frac{\mathbf{knmp}}{\mathbf{y}_4\mathbf{z}_2})$
M _{HJ}	- FM _{HJ}	- 2bdflar - Jx ₁ y ₄ z ₂	2dilor Jx ₁ y ₄ ² 2	$-\frac{2fknry_1}{jx_1y_4z_2}$	$\frac{2\mathrm{knr}}{\mathrm{jy_4}^{z_2}}$	$-\frac{2\mathrm{knr}}{\mathrm{x}_2}(1+\frac{1}{z}_2)$	$\frac{2n\mathbf{r}}{\mathbf{x}_2}(1+\frac{k\mathbf{m}}{\mathbf{y}_4\mathbf{z}_2})$	$-\frac{2\mathbf{r}}{x_2}(y_3 + \frac{kmnp}{y_4 z_2})$
М	FM _{JH}	bdfknr - jx ₁ y ₄ z ₂	dfknr Jx ₁ y ₄ z ₂	$-\frac{\operatorname{flary}_1}{\operatorname{jx}_1\operatorname{y}_4^2_2}$	lorr Jy _L z ₂	$-\frac{\operatorname{lmr}}{\operatorname{x}_2}(1+\frac{1}{\operatorname{z}_2})$	$\frac{\operatorname{nr}}{x_2}(1+\frac{kn}{y_4^22})$	$-\frac{r}{x_2}(y_3 + \frac{kmnp}{y_4 x_2})$

Section 3



(Courtesy of Texas Highway Department)

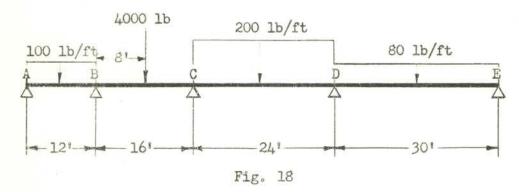
Nueces Bay Causeway on U.S. Highway 181 near Corpus Christi, Texas.

SECTION 3

ILLUSTRATIVE EXAMPLES

1. END MOMENTS FOR A FOUR SPAN FREELY SUPPORTED BEAM

A four span beam is loaded as shown in Fig. 18. The beam has spans of 12 feet, 16 feet, 24 feet and 30 feet and has a constant moment of inertia of 4 ft. Determine the end moments due to the transverse loading.



SOLUTION

- 1. Select table: Case C4 Four Span Beam Cyclosymmetrical
- 2. Determine the stiffness factors:

$$K_{1} = \frac{3}{4} \frac{\text{I}1}{\text{L}_{1}} = \frac{1}{4},$$

$$K_{2} = \frac{\text{I}2}{\text{L}_{2}} = \frac{1}{4},$$

$$K_{3} = \frac{\text{I}3}{\text{L}_{3}} = \frac{1}{6},$$

$$K_{4} = \frac{3}{4} \frac{\text{I}4}{\text{L}_{4}} = \frac{1}{10}.$$

$$\sum K_{B} = K_{1} + K_{2} = \frac{1}{2},$$

$$\sum K_{C} = K_{2} + K_{3} = \frac{5}{12},$$

$$\sum K_{D} = K_{3} + K_{4} = \frac{4}{15},$$

3. Determine the Constant Functions of the New Distribution Factor:

$$a = \frac{K_1}{2\sum K_B} = \frac{1}{4},$$

$$b = \frac{K_2}{2\sum K_B} = \frac{1}{4},$$

$$c = \frac{K_2}{2\sum K_C} = \frac{3}{10},$$

$$d = \frac{K_3}{2\sum K_C} = \frac{1}{5},$$

$$e = \frac{K_4}{2\sum K_D} = \frac{5}{16},$$

$$f = \frac{K_5}{2\sum K_D} = \frac{3}{16},$$

$$x = 1 - bc - de = \frac{69}{80},$$

$$y_1 = 1 - bc = \frac{37}{40},$$

$$y_2 = 1 - de = \frac{15}{16},$$

$$u_1 = 2 - b = \frac{7}{4},$$

$$u_2 = 2 - e = \frac{27}{16}.$$

4. Fixed End Moments:

The first and the last spans must be considered as prop spans in place of the usual fixed end spans when computing the end moments.

$$\begin{split} &\text{FM}_{\text{AB}} = 0, \\ &\text{EM}_{\text{BA}} = \frac{3}{2} \frac{\text{wl}^2}{12} = 1800 \text{ ft-lb}, \\ &\text{FM}_{\text{BC}} = + \text{ FM}_{\text{CB}} = + \frac{\text{Pl}}{8} = + 8000 \text{ ft-lb}, \\ &\text{FM}_{\text{CD}} = + \text{ FM}_{\text{DC}} = + \frac{\text{wl}^2}{12} = + 9600 \text{ ft-lb}, \\ &\text{EM}_{\text{DE}} = + \frac{3}{2} \frac{\text{wl}^2}{12} = + 9000 \text{ ft-lb}, \\ &\text{FM}_{\text{ED}} = 0. \end{split}$$

5. Final Moments:

М	FM	$\Sigma FM_B = -6200$	Σ FM _C = - 1600	Σ FMD = + 600	
M _{AB}	0	0	0	0	
M _{BA}	1800	5430	.1740	0543	
M _{BC}	-1800	. 5430	1740	.0543	
M _{CB}	8000	0978	6080	.1904	
M _{CD}	-9600	.0978	3914	1904	
MDC	9000	.0218	0870	.4022	
MDE	-9000	0218	.0870	4022	
M _{ED}	0	0	0	0	

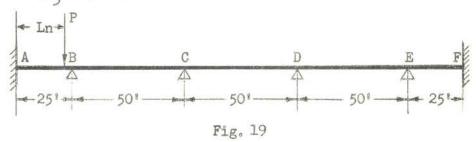
$$M_{AB} = 0$$
,
 $M_{BA} = 1800 - (.5430)(-6200) + (.1740)(-1600) - (.0543)(600)$
= 4857 ft-lb.

In a similar manner

$$M_{BC} = -4857 \text{ ft-lb},$$
 $M_{CB} = -M_{CD} = 9694 \text{ ft-lb},$
 $M_{DC} = -M_{DE} = 9245 \text{ ft-lb},$
 $M_{ED} = 0.$

2. INFLUENCE LINES

A single concentrated load of 10,000 pounds is to move across a five span continuous beam that is fixed at both ends and freely supported at four points. This beam has equal side spans of 25 feet each ($I_1 = 4 \text{ ft}_{\circ}^4$), equal intermediate spans of 50 feet each ($I_2 = 4 \text{ ft}_{\circ}^4$), and a center span of 50 feet ($I_3 = 4 \text{ ft}_{\circ}^4$).



Determine the influence line for moment $M_{\mbox{\footnotesize{BA}}}$.

SOLUTION

- 1. Select table: Case C5 Five Span Beam Symmetrical.
- 2. Determine the stiffness factors:

$$K_{1} = \frac{I_{1}}{L_{1}} = \frac{4}{25},$$

$$K_{2} = \frac{I_{2}}{L_{2}} = \frac{4}{50},$$

$$K_{3} = \frac{I_{3}}{L_{3}} = \frac{4}{50}.$$

$$\sum K_{C} = K_{2} + K_{3} = \frac{4}{25},$$

3. Constant Functions of the New Distribution Factor:

a =
$$\frac{K_1}{2\sum K_B} = \frac{1}{3}$$
,
b = $\frac{K_2}{2\sum K_B} = \frac{1}{6}$,
c = $\frac{K_2}{2\sum K_C} = \frac{1}{4}$,

$$d = \frac{K_3}{2\sum K_C} = \frac{1}{4}$$

$$x = 1 - bc = \frac{23}{24}$$

$$z = \left(\frac{x}{d}\right)^2 - 1 = \frac{493}{36}$$

4. Fixed end moments due to "P":

$$P$$
 2

FM₁₂ = PLn(1 - n)²,

$$FM_{21} = PLn^2(1-n).$$

5. Influence Line equation for moment $M_{\mbox{\footnotesize{BA}}}$, load in 1st span:

$$M_{BA} = FM_{BA} - 2\frac{a}{x}(1 + \frac{bc}{z}) FM_{BA},$$

= .3048 PLn²(1 - n) = 76,250 n²(1 - n).

n	$n^2(1-n)$	M _{BA}
0	.000	0
.2	.032	2,440
.4	.096	7,310
.6	.144	10,970
.8	.128	9,760
1.0	.000	0

If the influence line equation is differentiated with respect to n and the result set equal to zero, the point of maximum moment is obtained.

$$\frac{\partial M_{BA}}{\partial n} = 2n - 3n^2 = 0$$
, and $n = 0$, $\frac{2}{3}$. $M_{BAmax} = 11,120$ ft-lb.

6. Influence Line equation for moment $M_{\mbox{\footnotesize{BA}}^9}$ load in 2nd span:

$$M_{BA} = 2 \frac{a}{x} \left(1 + \frac{bc}{z}\right) FM_{BC} + 2 \frac{ac}{x} \left(1 + \frac{1}{z}\right) FM_{CB},$$

$$= .6952 FM_{BC} + .1865 FM_{CB} = 348,000 n(1 - n)^{2}$$

$$+ 93,200 n^{2}(1 - n).$$

n	$n(1 - n)^2$	$n^2(1-n)$	M _{BA}
0	.000	.000	0
.1	.081	。009	29,000
.2	.128	.032	47,600
۰3	.147	.063	57,100
.4	.144	.096	59,100
.5	.125	.125	55,200
.6	.096	.144	46,900
.7	.063	.147	35,600
.8	。032	.128	23,300
.9	。009	.081	10,700
1.0	.000	.000	0

$$\frac{\partial M_{BA}}{\partial n}$$
 = 348,000 (1 - 4n + 3n²) + 93,200 (2n - 3n²) = 0,
n = 1.201, .379, and M_{BAmax} = 59,120 ft-lb.

7. Influence Line equation for moment MBA, load in 3rd span:

$$M_{BA} = 2 \frac{ac}{x} (1 + \frac{1}{z})(-FM_{CD}) - 2 \frac{ac}{dz} FM_{DC},$$

$$= -.1865 FM_{CD} - .0487 FM_{DC},$$

$$= -.93,300 n(1 - n)^2 - 24,350 n^2(1 - n).$$

n	$n(1 - n)^2$	$n^2(1 - n)$	M _{BA}
0	.000	.000	0
.1	.081	.009	-7,869
.2	.128	.032	-12,719
.3	.147	.063	-15,260
.4	.144	.096	-15,770
.5	.125	.125	-14,700
.6	.096	.144	-12,470
.7	.063	.147	-9,460
.8	.032	.128	-6,110
.9	.009	.081	-1,040
1.0	.000	.000	0

$$\frac{\partial M_{BA}}{\partial n} = -93,300 (1 - 4n + 3n^2) - 24,350 (2n - 3n^2) = 0,$$

$$n = 1.188$$
, .386, and $M_{BAmax} = -15,820$ ft-lb.

8. Influence Line equation for moment $M_{\mbox{\footnotesize{BA}}^9}$ load in 4th span:

$$M_{BA} = -2 \frac{ac}{dz} (-FM_{DE}) + 2 \frac{abc}{dz} FM_{ED},$$

= 24,350 n(1 - n)² + 4,055 n²(1 - n).

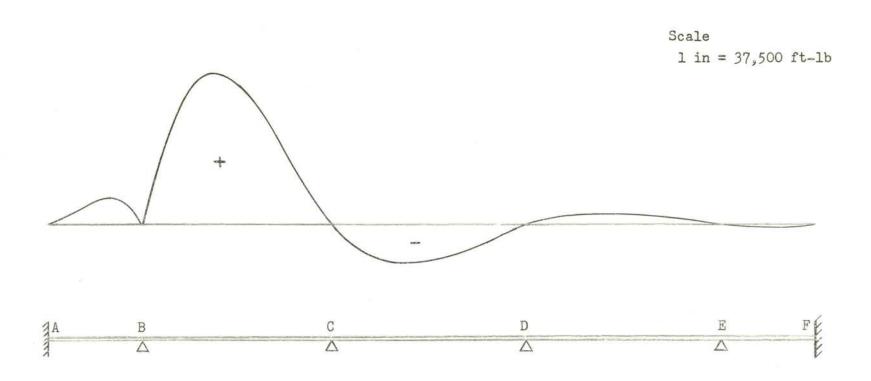
n	M _{BA}	n	M _{BA}
0	0	.6	3,220
.1	2,000	.7	2,130
.2	3,250	.8	1,300
.3	3,850	.9	550
٠4,	3,900	1.0	0
.5	3,550		

9. Influence Line equation for moment M_{BA} , load in 5th span:

$$M_{BA} = 2 \frac{abc}{dz} FM_{EF} = 2028 n(1 - n)^2$$
.

n	$^{ m M}_{ m BA}$
0	0
.2	-254
•4	-290
.6	-195
.8	-65
1.0	0

Fig. 20 INFLUENCE LINE FOR MOMENT AT POINT B



3. SETTLEMENT OF SUPPORTS

Determine the end moments of the six span beam shown in Fig. 21, due to the vertical displacements of the supports.

$$\triangle_{\rm B}=.5~{\rm in},$$

$$\triangle_{\rm C}=.8~{\rm in},$$

$$E=30~{\rm 10}^6~{\rm lb/in},$$

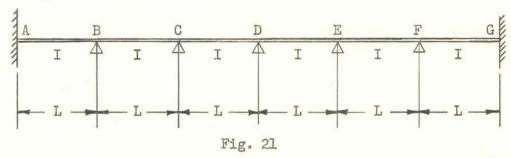
$$\triangle_{\rm D}=1.0~{\rm in},$$

$$I=720~{\rm in}^4,$$

$$\triangle_{\rm E}=.8~{\rm in},$$

$$L=360~{\rm in},$$

$$\triangle_{\rm F}=.5~{\rm in}.$$



SOLUTION

- 1. Select table: Case A60 Six Span Beam Equal Spans.
- 2. Determine the fixed end moments due to the displacements:

$$FM = \frac{6EI}{L^2} \triangle = \frac{6 \times 30 \times 10^6 \times 720}{360 \times 360} \triangle = \triangle 10^6 \text{ lb.}$$

The fixed end moments at the low end of each span must be multiplied by (-1), then the tables may be used.

$$\begin{split} \mathrm{FM}_{\mathrm{AB}} &= -\; \mathrm{FM}_{\mathrm{BA}} = \; 5 \times 10^5 \; \; \mathrm{in\text{-}lb}, \\ \mathrm{FM}_{\mathrm{DE}} &= -\; \mathrm{FM}_{\mathrm{ED}} = -\; 2 \times 10^5 \; \; \mathrm{in\text{-}lb}, \\ \mathrm{FM}_{\mathrm{BC}} &= -\; \mathrm{FM}_{\mathrm{CB}} = \; 3 \times 10^5 \; \; \mathrm{in\text{-}lb}, \\ \mathrm{FM}_{\mathrm{CD}} &= -\; \mathrm{FM}_{\mathrm{DC}} = \; 2 \times 10^5 \; \; \mathrm{in\text{-}lb}, \\ \mathrm{FM}_{\mathrm{CD}} &= -\; \mathrm{FM}_{\mathrm{DC}} = \; 2 \times 10^5 \; \; \mathrm{in\text{-}lb}, \\ \end{split}$$

М	FM	$\Sigma FM_B = -8 \times 10^5$	Σ FM _C = -5×10^5	$\Sigma FM_D = 0$	ΣFM _E = +5×10 ⁵	$\Sigma FM_F = +8 \times 10^5$
MAB	-5x10 ⁵	2680	.0718	0192	.0051	0013
M _{BA}	-5×10 ⁵	5360	.1436	0384	.0102	0026
M _{BC}	5×10 ⁵	•5360	1436	.0384	0102	.0026
M _{CB}	-3×10 ⁵	1242	4973	.1345	0359	.0090
M _{CD}	3×10 ⁵	.1242	•4973	1345	.0359	0090
M _{DC}	-2×10 ⁵	•0333	1333	5000	.1333	0333
M _{DE}	2×10 ⁵	0333	.1333	5000	1333	.0333
M _{ED}	-3x10 ⁵	0090	.0359	1345	.4973	.1242
M _{EF}	3×10 ⁵	•0090	0359	.1345	4973	1242
MFE	-5×10 ⁵	.0026	0102	.0384	1436	•5360
M _{FG}	5×10 ⁵	0026	.0102	0384	.1436	5360
$^{ m M}_{ m GF}$	5×10 ⁵	0013	.0051	0192	.0718	2680

$$\begin{aligned} \text{M}_{\text{AB}} &= -5 \times 10^5 - .2680 (-8 \times 10^5) + .0718 (-5 \times 10^5) \\ &+ .0051 (5 \times 10^5) - .0013 (8 \times 10^5) = -3.205 \times 10^5 \text{ in-lb.} \end{aligned}$$

In a similiar manner

$$\begin{split} & \mathbf{M}_{\mathrm{BA}} = -\; \mathbf{M}_{\mathrm{BC}} = -\; 1.40 \times 10^5 \; \; \mathrm{in\text{-}lb.} \\ & \mathbf{M}_{\mathrm{CB}} = -\; \mathbf{M}_{\mathrm{CD}} = \; .373 \times 10^5 \; \; \mathrm{in\text{-}lb.} \\ & \mathbf{M}_{\mathrm{DC}} = -\; \mathbf{M}_{\mathrm{DE}} = -\; 1.20 \times 10^5 \; \; \mathrm{in\text{-}lb.} \\ & \mathbf{M}_{\mathrm{ED}} = -\; \mathbf{M}_{\mathrm{EF}} = \; .373 \times 10^5 \; \; \mathrm{in\text{-}lb.} \\ & \mathbf{M}_{\mathrm{FE}} = -\; \mathbf{M}_{\mathrm{FG}} = -\; 1.40 \times 10^5 \; \; \mathrm{in\text{-}lb.} \\ & \mathbf{M}_{\mathrm{GF}} = \; 3.205 \times 10^5 \; \; \mathrm{in\text{-}lb.} \end{split}$$

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