A STUDY OF THE IONIZATION PROPERTIES OF HYDROGEN

IN THE PRESENCE OF RELATIVELY STRONG

MAGNETOSTATIC FIELDS

By JAMES RUSSELL McDOUGAL

Bachelor of Science University of Kansas Lawrence, Kansas 1952

Master of Science Tohoku University Sendai, Japan 1957

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STATE UNIVERSITY

Name: James Russell McDougal

OCI 25 1968 Date of Degree: May, 1968

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- Scope of Study: The objective of this study is to present an engineering orientated investigation into the ionization properties of hydrogen in the presence of a relatively strong magnetostatic field (i.e., approximately 0.6 to 5 kilogauss) with rf excitation over the frequency range of 2 to 30 MHz. The method of study involves an analysis based upon solutions to the Lorentz Force Equation, the design and construction of an experimental system and apparatus for taking data in the framework of this theory, and the tabulation and analysis of the experimental results obtained from this system. The experimental work is carried out at reduced pressures, varying approximately from one to six millimeters of mercury.
- Findings and Conclusions: Certain values of magnetic intensity are found to be optimum in the sense that their application allows a minimal degree of ionization to be maintained in the gas with the least magnitude of rf excitation. This is explained as the result of a resonance phenomenom which was predicted in the analysis. An extensive data mapping of these optimum magnetic intensity values throughout the range of excitation frequencies is given and compared with the theory. Reasonable agreement is found at the lower excitation frequencies, but it is concluded that the theory is not adequate to predict conditions at the higher excitation frequencies. It is hoped that the results of this study may be of aid to those searching for new and improved methods of energy conversion.

ADVISER'S APPROVAL

A STUDY OF THE IONIZATION PROPERTIES OF HYDROGEN IN THE PRESENCE OF RELATIVELY STRONG MAGNETOSTATIC FIELDS

Thesis Approved: Thesis Adviser C 0

Dean of the Graduate College

PREFACE

Ionization phenomena of various and sundry sort have been studied for many years. Sir J. J. Thomson in his Conduction of Electricity Through Gases (1928) (1) quotes references dating well back into the 19th Century. This is understandable since such phenomena, generally speaking, exhibits striking effects. One finds literally reams of literature relating to detailed observations of the glow seen in gaseous discharge tubes, and I must admit that even after working with ionization cells for a period of three years there still exists an aura of excitement at the instant in an experiment when the glow first appears. This area has long been very much the exclusive domain of the experimental physicist, but fortunately for the rest of us recent advances in technology, the exploration of space, and increasing concern with the conservation of the earth's dwindling sources of energy have all given the study of ionization phenomena a practical significance which it did not previously enjoy. With the promise of practical and economic benefits to be ultimately obtained, the engineer gains his excuse, and with luck, the money necessary for entering this exciting and extremely complicated field.

This thesis attempts to present an engineering orientated study, for which the incentive came from fuel cell research, of the *i*onization properties of hydrogen in the presence of a relatively strong magnetostatic field with rf excitation. It cumulates in the presentation of an Optimum Magnetic Intensity Mapping of Hydrogen. Here, the optimum

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magnetic intensity is that value of applied magnetostatic field which allows a minimal degree of ionization to be continuously maintained in the gas with the least magnitude of rf excitation. The mapping presented in Chapter V of this thesis covers the measurement of optimum magnetic intensity throughout the range of 2 to 30 MHz in excitation frequencies as a function of the parameters pressure, applied voltage, and current through the ionization cell. All measurements were performed at reduced pressures, varying approximately from one to six millimeters of mercury. It is hoped that this information may aid in the conception of new and improved methods of energy conversion.

I wish to express my gratitude to my thesis adviser, Dr. Wm. L. Hughes, for his invaluable assistance and guidance during my doctoral studies. I also wish to thank Mr. Louis Long of the U. S. Army Research and Development Laboratories for his advice and support in the form of a contract. Thanks and gratitude are also due the other members of my advisory committee; to Dr. E. E. Kohnke for his many illuminating discussions and guidance in areas of physics, and to Professor P. A. McCollum for his instruction and assistance in the use of computers. Others who have helped me were my co-workers; Russell Lawson during the formative stages of the study, and Cheng-I Chen, who took much of the later data and accomplished the greater portion of the computer plotting given in Chapter V.

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CHAPTER I

INTRODUCTION

Townsend's Theory of Ionization

Interest in ionization became strong at about the turn of the century in connection with investigations into the basic structure of matter. One of the earliest investigations, and one which set a pattern for the interpretation of results which is employed to this present day was reported by J. S. Townsend¹ in 1900. Townsend conducted his experiment with air in an evacuated glass tube at pressures of 2 and 8 mm Hg in which parallel plane electrodes had been installed, somewhat as shown in Figure 1 below. Townsend observed that at these



Figure 1. Townsend's Experiment

¹J. S. Townsend, <u>Nature</u>, 62:340 (1900).

pressures and with applied electric intensities, V/d, up to about 10 volts per centimeter, the total current measured was due to the electrons produced through ionization by the X-rays. At greater applied electric intensities, Townsend observed that the current increased and he based his explanation of this phenomenon upon an argument of essentially the following nature.

Suppose that n electrons in moving a distance dx in the gas produce α ndx others, where α is a constant depending upon the applied electric intensity and pressure.

Then,

$$dn = \alpha n dx, \qquad (1.1)$$

or

$$\frac{\mathrm{dn}}{\mathrm{n}} = \alpha \mathrm{dx} \tag{1.2}$$

which integrates to give

$$n = K e^{\alpha x}$$
 (1.3)

where ϵ is the base of natural logarithms. At x = 0, the number of electrons will be those produced by the X-rays, say n_o. Inserting this initial condition, Equation (3) is written

$$n = n_0 e^{CX}$$
(1.4)

and the total number of electrons produced in traveling the distance between the electrodes, d, will be:

$$N = \int \frac{d}{o} n_o e^{\alpha x} dx = \frac{n_o}{\alpha} (e^{\alpha d} - 1). \qquad (1.5)$$

The total number of electrons produced by the X-rays alone is obtained by setting $\alpha = 0$ in the above integral, in which case integration gives

$$\mathbf{N}' = \mathbf{n}_{\mathbf{d}} \tag{1.6}$$

and we may combine Equation (5) and (6) to write:

$$\frac{N}{N} = \frac{1}{\alpha d} \left(e^{\alpha d} - 1 \right).$$
(1.7)

Now, assuming all of the carriers to have the same charge and travel with the same velocity, which would be the case for electrons in the absence of a space charge, Equation (7) may be written as

$$\frac{i}{i_{o}} = \frac{1}{\alpha d} (\epsilon^{\alpha d} - 1). \qquad (1.8)$$

in terms of current. Townsend based his interpretations upon an equation of the form of (8) above. However, in more modern work it is customary to note that Equation (8) approaches the limit

$$i = i_0 \varepsilon^{\alpha d}$$
 (1.9)

as the separation, d, between the electrodes becomes very small. As this is, of necessity (i.e., in order to obtain the desired degree of uniformity in the applied electric field), sufficiently close to the usual case in practice, most references^{2,3,4} refer to Equation (9) as Townsend's equation. The constant α is called the Townsend ionization coefficient, or simply ionization coefficient. It is usually measured by varying the electrode spacing and pressure in such a manner as to maintain a constant ratio of electric intensity to pressure, E/P. The ionization coefficient is then determined as the slope of the curve obtained by plotting the logarithm of the measured values of current versus the electrode spacing at a specified value of E/P.

In a broad sense, Townsend's equation is generally considered to apply up to the point where the gas breaks down, although the equation

² M. A. Harrison and R. Geballe, <u>Phys. Rev.</u>, 91:1 (1953).

- ³D. J. Rose, <u>Phys</u>. <u>Rev</u>., 104:273 (1956).
- ⁴D. J.DeBitetto and L. H. Fisher, <u>Phys. Rev.</u>, 104:1213 (1956).

requires modification⁵ if the finer details are to be taken into account. Contemporary studies⁶ with crossed static electric and magnetic fields also follow this basic pattern, with the addition of a constant magnetic field to pressure ratio, H/P. Apparently, the Townsend equation with extensions and modifications can be used to accurately predict the point of breakdown, the so-called sparking potential, of some gases.⁷ Basically, however, it is to be considered as an adequate description of phenomena observed prior to breakdown.

The Franck and Hertz Experiment

Another experiment of historical interest which has continued to influence the pattern of ionization research to the present day is that performed with a triode gas tube by Franck and Hertz⁸ in 1914. These investigators constructed a triode gas tube filled with mercury vapor as shown in the schematic diagram of Figure 2. The vapor pressure in the tube and the distances between tube elements were adjusted so that the mean free path was considerably smaller than the distance between cathode and grid and somewhat greater than the distance between grid and anode. As shown in the figure; the grid was made slightly positive, e.g., 0.5 volts, with respect to the anode so as to produce

⁷D. J. DeBitetto and L. H. Fisher, <u>Phys. Rev</u>., 111:390 (1958).

⁸ J. Franck and G. Hertz, <u>Phys. Ges.</u>, 16:12 (1914); <u>Phys. Zeits</u>, 17:409 (1916); also discussion in H. E. White, <u>Introduction to Atomic Spectra</u>, McGraw-Hill (1934), p. 92.

⁵D. J. DeBitetto and L. H. Fisher, <u>Phys. Rev</u>., 104:1213 (1956).

⁶S. C. Haydon and A. G. Robertson, Proc. 5th International Conf. on Ionization Phenomena in Gases, 75 (1961).



Figure 2. The Franck and Hertz Experiment

a negative potential gradient in the region between the grid and the anode. Then, for small values of V_g , few of the electrons emitted by the cathode could reach the anode because the energy they had acquired in traversing the region between the cathode and grid was too low, consequently the anode current, i_p , was small. For large V_g , many more of the electrons were able to reach the anode due to the greater energy acquired in traveling between cathode and grid, consequently i_p was large. As the grid voltage, V_g , was increased, Franck and Hertz measured a plate current, i_p , as shown in Figure 3. In the region labeled 0 to A on the curve, i_p increased as would be predicted on the basis of the increasing energy acquired by the electrons. In the region labeled A to B some of the electrons (i.e., those emitted from the cathode with greater energy) had acquired sufficient energy to collide inelastically with Hg atoms near the grid, giving up all of



Figure 3. Sketch of the Measured Results of Franck and Hertz

their energy and consequently were unable to reach the anode, hence i_p decreased. In the region B to C the electrons acquired sufficient energy after collision to reach the anode and i_p increased. In the region C to D the electrons have collided with the Hg atoms both well in front of the grid and then near it, consequently could not reach the anode and i_p again decreased. Throughout the remainder of the curve the process described above was repeated with additional collisions in the region between cathode and grid.

The essentially constant potential difference between successive peaks of the curve demonstrated that the Hg atoms absorbs energy only in specified amounts. This potential difference is the amount of energy, in electron volts, required to raise a valence electron from the ground state to the next higher level and is called the first excitation potential. A more refined apparatus than that employed by

Franck and Hertz allows one to determine the energy required to completely strip an electron from a nucleus giving the first ionization potential. Contemporary apparatus⁹ uses refined electrode systems such as electron-guns, etc., but the principle remains the same as that employed by Franck and Hertz.

The Engineering Need for Further Studies

in Ionization

As stated previously, the experiments of Townsend and Franck and Hertz have set the basic pattern for the investigations into the nature of ionization phenomena which followed. Each of these is a collision experiment, however, the respective results apply to different realms in the subdivisions of matter. As determined by experiment, the Townsend ionization coefficient is essentially a macroscopic idea; it describes the pre-breakdown growth of ion formations in an aggregate of molecules constituting a gas. On the other hand, excitation and ionization potentials are microscopic in character, apply to single atoms and molecules, and are very closely related to atomic structure. As evidenced by the references quoted up to this point, there is a wealth of Townsend ionization coefficient data available in the literature, and virtually any physics handbook will give ionization and excitation potentials for the elements and many compounds. However, in the engineering considerations related to non-conventional and esoteric energy sources where ionized gases are involved, ionization data of a different nature may be desired. Here the concern may not

⁹See, for example, E. W. Rothe, <u>et. al, Phys</u>. <u>Rev</u>., 125:582 (1962).

be primarily with the fundamental structure of matter, nor with conditions leading up to breakdown in the gas, but rather with the problem of economically producing ions in the gas at some steady rate in a specified environment. Of course, ionization potentials, Townsend coefficients, etc., should relate rather directly to this problem, but as this environment becomes more complicated it becomes increasingly more difficult to determine the nature of such relations. Almost invariably reactions with external force fields are involved and this would appear to place such considerations in the realm of that which is called plasma physics. Indeed, the general subject matter of plasma physics does treat the interactions of ionized gases with external force fields, especially electromagnetic fields, in great theoretical detail, but this treatment appears to be normally carried out largely neglecting any experimental necessity to excite the gas. Of course, this is natural in that the plasma physicist is usually treating, at least tacitly, aggregates of ionized particles which have been inadvertently created (i.e., so far as man's intentions are concerned) by the forces of nature.

The ionization study to be discussed on the following pages of this report arose out of attempts to construct a gaseous fuel cell in which a magnetostatic field was to replace the electrolyte. This consideration fixed both the gases to be investigated and the environment in which ionization was to be maintained. In the fuel cell application, the gases of interest are hydrogen and oxygen. The simpler gas, hydrogen, was selected to initiate the study and the remainder of this paper will be devoted to the discussion of the findings with this gas.

CHAPTER II

ANALYSIS

The Initial Assumptions and Solution of the Lorentz Force Equation

From the published results¹ of mass spectrographic analysis of Linde commerical tank hydrogen, i.e., 99.6 per cent H₂, which is the gas employed in conducting this study, one must conclude that the hydrogen phenomena is intimately associated with the physics of the hydrogen molecule. The hydrogen molecule, considering that it is the simplest of all molecules, turns out to be remarkably complex.² Consequently, the quantum mechanics of the molecule has been discarded in an initial attempt at analysis, using its literature² only as a guide to the type of ions to be expected, and taking resort to a much simpler form of theory. This approach has severe limitations, of course, in that the simpler theory can never be completely, or for that matter even satisfactorily, descriptive; but it is hoped that the knowledge gained in continuing the analysis will eventually lead to the incorpora-

¹D. J. Bitetto and L. H. Fisher, <u>Phys</u>. <u>Rev</u>., 104:1213 (1956).

²E. Teller, <u>Z</u>. <u>Physik</u>, 61:458 (1930): W. Hettler and F. London, <u>Z</u>. <u>Physik</u>, 46:47 (1927), 47:835 (1928), 51:805 (1929); S. C. Wang, <u>Phys. Rev.</u>, 31:579 (1928); A. S. Coolidge and H. M. James, <u>J. Chem.</u> <u>Phys.</u>, 1:825 (1933); C. A. Coulson, <u>Trans. Faraday Soc</u>., 33:1473 (1937); also, for an over-all summary account, see J. C. Slater, <u>Quantum Theory</u> <u>of Matter</u>, 1951.

tion of enough of the quantum mechanical ideas to build a sufficiently descriptive theory.

Starting with the common assumption³ that inelastic collision of charged with uncharged particles is the primary mechanism of ionization produced in a gas under the influence of an applied electric field, the motion of charged particles in an environment more or less approximating the physical conditions inside the ionization cells employed in this investigation will be studied. Neglecting the flow of gas through the cell, it is assumed that the motion of an average charged particle in the absence of any collisions will obey the Lorentz force equation, i.e., assuming non-relativistic motion

$$\frac{dV_o}{dt} = \frac{q}{m} \left(\overline{E} + \overline{V}_o \times \overline{B} \right)$$
(2.1)

where \overline{E} is the electric intensity, \overline{B} is the magnetic intensity, and \overline{V}_{O} is the velocity of an average particle of mass m and charge q. Consider the case where \overline{B} is uniform, independent of time, and \overline{E} is uniform and alternating with angular frequency ω ,

$$\overline{E} = \overline{E}_{o} e^{j\omega t}.$$
(2.2)

So long as non-relativistic motion is assumed, the effect of the induced magnetic intensity may be neglected with respect to that of the applied electric intensity since for a plane wave in free space the ratio

$$\left|\frac{\overline{E}}{\overline{V} \times \overline{B}}\right| = c/v$$

where c is the velocity of light and $|\vec{v}| = v$ is the velocity of a

³M. A. Harrison and R. Geballe, <u>Phys</u>. <u>Rev.</u>, 91:1 (1953); L. Gould and L. W. Roberts, <u>The Breakdown of Air at Microwave Frequencies</u>, Microwave Assoc., Inc., MA-1 (1955); L. Gould, NAVSHIPS INDEX NO. NE-111616 (1956)

charged particle under the influence of the fields. Resolve the electric intensity vector into components parallel and perpendicular to the applied magnetic intensity. Denoting the components,

$$\overline{E}$$
 parallel to \overline{B} as \overline{E}_{11} , and
 \overline{E} perpendicular to \overline{B} as \overline{E}_1 ,

and substituting Equation 2.2 into Equation 2.1, the equation

$$\frac{d\overline{V}_{o}}{dt} = \frac{q}{m} \left(\overline{E}_{o} e^{j\omega t} + \overline{V}_{o} \times \overline{B}\right) = \frac{q}{m} \left(\overline{E}_{11} + \overline{E}_{1} + \overline{V}_{o} \times \overline{B}\right)$$
(2.3)

is obtained. Guided by the knowledge that the equation

$$\frac{\mathrm{d}V}{\mathrm{d}t} = \frac{\mathbf{q}}{\mathrm{m}} \, \overline{\mathbf{V}} \, \mathbf{x} \, \overline{\mathbf{B}} \, , \qquad (2-4)$$

where \overline{B} is uniform, independent of time, has solution

$$\overline{\mathbf{v}} = \overline{\mathbf{v}}_{11} + \overline{\Omega} \times \mathbf{r}_{g}$$
(2-5)

where $\overline{\Omega} = -q/m \overline{B}$ is the vector angular velocity, \overline{r}_g is the radius of gyration, and $|\overline{\Omega}| = \omega_g$, the frequency of gyration, assume a solution to Equation (3) of the form

$$\overline{V}_{o} = a_{o} \overline{E}_{11} + a_{1} \overline{E}_{1} + a_{2} \overline{E}_{1} \times \overline{\Omega} . \qquad (2-6)$$

Here, in general, the a's are functions of time. Substituting Equation 2.6 into Equation 2.3 one obtains the following set of simultaneous differential equations in time for the solution of the coefficients:

$$\dot{a}_{0} + j \omega a_{0} = q/m$$

 $\dot{a}_{1} + j \omega a_{1} - \omega g^{2} a_{2} = q/m$
 $\dot{a}_{2} + j \omega a_{2} + a_{1} = 0$
(2-7)

Delcroix⁴ gives the following integrations for Equations 2.7, which, of course, may be readily checked by direct substitution.

Case 1.
$$w = 0$$

 $a_0 = (\frac{q}{m}) t$
 $a_1 = 0$
 $a_2 = -q/m w_g^2$
Case 2. $w \neq 0$ and $w \neq w_g$
 $a_0 = -(q/m) j/w$

⁴J. L. Delcroix, <u>Introduction to the Theory of Ionized Gases</u>, Interscience Publishers, 1960, Chapter 5.

No constants of integration appear in these solutions due to the method of obtaining the complete solution. Delcroix derives a superposition theorem considering \overline{V} as the total velocity of a particle obeying the Lorentz force equation. Then it is assumed that $\overline{V} = \overline{V}_0 + \overline{V}_1$, where \overline{V}_0 is the average velocity of a group of particles, all of charge q and mass m. The velocity \overline{V}_1 is added to account for the particular particle in question. Delcroix breaks this system down into two vector differential equations,

$$\frac{d\overline{V}_{o}}{dt} = \frac{q}{m} (\overline{E} + \overline{V}_{o} \times \overline{B})$$
$$\frac{d\overline{V}_{1}}{dt} = \frac{q}{m} \overline{V}_{1} \times \overline{B}$$

Initial conditions for a particular particle are contained in the integration of this latter equation. An estimate of gas flow velocity based upon the hydraulics of the system is of the order of 10 meters per second. As we shall see in Chapter V, calculated particle velocities range from approximately 10 to 10×10^6 meters per second; and it is believed that the important phenomena is associated with the higher particle velocities. For this reason, it is assumed that initial conditions may be neglected here.

$$a_1 = (q/m) j \omega(\omega_g^2 - \omega^2)$$

 $a_2 = -(q/m)/(\omega_g^2 - \omega^2)$

<u>Case 3.</u> $\omega = \omega$

$$a_{0} = -(q/m) j/w$$
$$a_{1} = (q/2mw) (wt - j)$$
$$a_{2} = (q/2mw) jt$$

The above form of analysis is extremely limited since among the many aspects of the phenomena which are neglected, no accounting is made for the effects of pressure or collision frequency. Nevertheless, much of the observed phenomena appears to be qualitatively deducible in terms of the solutions obtained above. This is discussed below in terms of Cases 1, 2, and 3 and related to the experimental data of Chapter V. Furthermore, it appears that, to a first-order approximation at least, the collision frequency (or mean free path) effects may be inserted into the calculations as an "afterthought."

Development and Discussion of the Solutions for the Various Cases

<u>Case 1.</u> (w = 0): For this case with static applied electric intensity, from the solution for a_0 , charged particles are uniformly accelerated in the \overline{E}_{11} direction with opposite directions for electrons and positive ions. From the solution for a_1 there is no motion in the direction of \overline{E}_1 . However, there is motion in the direction of $\overline{E} \times \overline{B}$. Denoting this by \overline{V}_p , for drift velocity,

$$\overline{V}_{D} = a_{2} \overline{E}_{1} \times \overline{\Omega} = (q/m)^{2} \frac{\overline{E}_{1} \times \overline{B}}{\omega_{q}^{2}}$$
(2.8)

and since $w_{q} = q/m | B |$, Equation 2.8 may be written as

$$\overline{\overline{V}}_{D} = \frac{\overline{\overline{E}} \times \overline{\overline{B}}}{\frac{2}{\overline{B}}}$$
(2.9)

where $\frac{2}{B} = \overline{B} \cdot \overline{B}$. This velocity is independent of the quantity q/m and will, therefore, be the same for both the electrons and ions. Then, if ionization is produced primarily by a collision process, it is seen that the application of \overline{E}_{11} alone would require the same energy to maintain ionization as when there is no \overline{B} applied because the motion is not affected by \overline{B} in this case. If \overline{E}_1 alone were applied, the movement of electrons and ions in the $\overline{E} \times \overline{B}$ direction would, in general, be in a direction parallel to the electrodes and, consequently, might not be so rapidly removed from the cell by the electrodes as in the usual case of no applied \overline{B} . For this reason the \overline{E}_1 orientation might require less applied \overline{E} in order to maintain ionization in certain circumstances. However, the static \overline{E} case is of lesser interest in this investigation since it is a generally accepted fact⁵ that ionization occurs much more readily under the influence of rf fields.

<u>Case 2.</u> $(w \neq 0, w \neq w_g)$: This is the case which is more generally applicable to the measured data presented in Chapter V of this report. Delcroix⁶ discusses this case in terms of tensor mobilities and tensor conductivities, but it appears to more aptly suit the purpose at hand

⁵See, for example, A. B. Cambel, Plasma Physics and Magnetofluid-Mechanics, McGraw-Hill (1963), p. 92.

⁶ J. L. Delcroix, <u>Introduction to the Theory of Ionized Gases</u>, Interscience Publishers (1960) Chapter 5.

to avoid this nuisance by considering first \overline{E}_{11} only and next \overline{E}_{1} only as the directions of applied electric intensity.

)
$$E_{11}$$
 only: The solution becomes:
 $\overline{V}_{11} = -j \ q/m\omega \ \overline{E}_{11}$
 $\overline{V}_{1} = a_1 \ \overline{E}_1 = 0$
 $\overline{V}_{D} = a_2 \ \overline{E}_1 \ x \ \overline{\Omega} = 0$
(2.10)

There is movement in the \overline{E}_{11} direction alone. Suppose there was no applied \overline{B} . The Lorentz force equation becomes

$$\frac{dV_o}{dt} = q/m \,\overline{E}_o \, e^{j\omega t} \tag{2.11}$$

and integrating, the solution

$$\overline{V}_{o} = q/m \overline{E}_{o} \int e^{j\omega t} dt = -j q/m\omega \overline{E} + K \qquad (2.12)$$

is obtained. It is seen that in the case of \overline{E}_{11} the applied \overline{B} has no affect upon the motion. Consequently, one would expect the same energy to be required in order to maintain ionization as when no \overline{B} is applied. The experimental evidence indicates that it is at least as difficult to maintain ionization with \overline{E}_{11} as when no \overline{B} is applied, and perhaps the application of \overline{B} renders ionization more difficult to perform.

(b) \overline{E}_1 only: This represents the normal orientation employed throughout the study and is experimentally found to be that in which ionization is most easily maintained. The solutions become:

$$\overline{V}_{11} = a_0 \ \overline{E}_{11} = 0$$

$$\overline{V}_1 = a_1 \ \overline{E}_1 = j \omega(q/m) / (\omega_g^2 - \omega^2) \ \overline{E}_1 \qquad (2.13)$$

$$\overline{V}_D = a_2 \ \overline{E}_1 \ x \ \overline{\Omega} = (q/m)^2 \ \frac{\overline{E}_1 \ x \ \overline{B}}{(\omega_g^2 - \omega^2)}$$

Now, if $w_g^2 \gg w^2$, then Equations 2.13 become

$$\overline{\overline{v}}_{1} \cong j w/w_{g}^{2} (q/m) \overline{\overline{E}}_{1}$$

$$\overline{\overline{v}}_{D} \cong (q/m)^{2} \frac{\overline{\overline{E}}_{1} \times \overline{B}}{w_{g}^{2}} = \frac{\overline{\overline{E}} \times \overline{B}}{\overline{\overline{B}}^{2}} . \qquad (2.14)$$

On the other hand, if $w_g^2 \ll w^2$, then Equations 2.13 become

$$\overline{\mathbf{v}}_{1} \simeq -j/\omega \ (q/m) \ \overline{\mathbf{E}}_{1}$$

$$\overline{\mathbf{v}}_{D} \simeq - \ (q/m)^{2} \ \frac{\mathbf{E}_{1} \times \mathbf{B}}{\omega^{2}} \qquad . \tag{2.15}$$

Now, from a consideration of the literature⁷ of the hydrogen molecule, it may be presumed that three types of charged particles are present in the cell when ionization is taking place; namely, these are negative electrons e, positively ionized hydrogen atoms H^+ , and singly ionized hydrogen molecules H^+ . In addition, there is the reaction $H_2^+ + H_2 \rightarrow$ $H_3^+ + H$ which frequently occurs between a slow molecular ion and the hydrogen molecule, so that H_3^+ may also be present. The angular frequency of gyration, w_g , for all particles is given by the equation

⁷E. Teller, Z. Physik, 61:458 (1930): W. Hettler and F. London,
Z. Physik, 46:47 (1927), 47:835 (1928), 51:805 (1929); S. C. Wang,
Phys. Rev., 31:579 (1928); A. S. Coolidge and H. M. James, J. Chem.
Phys., 1:825 (1933); C. A. Coulson, Trans. Faraday Soc., 33:1473 (1937);
also for an over-all summary account, see J. C. Slater, Quantum Theory of Matter, 1951.

$$\omega_{g} = q/m \left| \overline{B} \right| . \qquad (2.16)$$

Upon substitution of the applicable values for q and m, Equation 2.16 becomes:

(i) for an electron, e:

$$\omega = 17.6 \times 10^4 | \overline{B} (kg) | \text{ mega-radian/sec.}, \qquad (2.17)$$

or

$$f_{g_e} = 28 | \overline{B} (kg) | GH_{z}$$
(2.18)

(ii) for H⁺:

$$w_{gH} + = 9.6 | B (kg) |$$
 mega-radian/sec., (2.19)

or

f

$$_{gH}^{+} = 1.53 |\overline{B} (kg)| MH_{z}$$
 (2.20)

(iii) and for
$$H_2^+$$
:
 $f_{gH_2}^+ = 0.77 | \overline{B} (kg) | MH_z$
(2.21)

From Equation 2.17 it is seen that Equations 2.14 should always hold for the electron throughout the frequency range of this investigation. However, as \overline{B} and the frequency are varied, conditions for the ions H⁺ and H₂⁺ will range through Equations 2.13 to 2.15, with the "exact" Equations 2.13 being the more generally applicable. From Equations 2.13 (approximations 2.14 and 2.15 also give the same result), the ratio of magnitude of drift to perpendicular velocities is found to be

$$\left|\frac{\overline{v}_{D}}{\overline{v}_{1}}\right| = \frac{q}{\omega m} |\overline{B}| . \qquad (2.22)$$

Upon substitution of the applicable values for q and m, Equation 2.22 may be written:

(iv) for e: $\begin{vmatrix} \overline{V}_{D} \\ \overline{V}_{1} \end{vmatrix} = \frac{2.8 \times 10^{4}}{f (MH_{z})} | \overline{B} (kg) |, \qquad (2.23)$ (v) for H⁺: $\begin{vmatrix} \overline{V}_{D} \\ \overline{V}_{1} \end{vmatrix}_{H^{+}} = \frac{1.53}{f (MH_{z})} | \overline{B} (kg) |, \qquad (2.24)$ (vi) and for H_{2}^{+} ; $\begin{vmatrix} \overline{v} \end{vmatrix}$

$$\left| \frac{\overline{v}_{D}}{\overline{v}_{1}} \right|_{H_{2}^{+}} \stackrel{\stackrel{\circ}{=}}{\longrightarrow} \frac{0.77}{f (MH_{z})} |\overline{B} (kg)| .$$
(2.25)

Reference to the experimental data (presented in Chapter V) shows $|\overline{B}|$ values ranging from approximately 0.6 to 5 kilogauss and f (MH_z) ranging from 2 to 30. Then, the minimum ratio for an electron, $|\overline{v}_D/\overline{v}_1|$, ranges from approximately 10⁴ to 10³; and it is concluded that \overline{v}_D , within the attending assumptions in its derivation, should be the ruling velocity so far as concerns electrons (ratio of diameter to spacing of the electrodes in the cell is approximately 10). This is not, however, the case with the ions H⁺ and H₂⁺. As seen by Equations 2.24 and 2.25, the frequencies and applied magnetic intensities employed in the experimental investigation are of such magnitudes as to sensitively affect the $|\overline{v}_D/\overline{v}_1|$ ratios for the ions. On this basis, one would predict that observed phenomena which is essentially independent of frequency and magnetic intensity must be due to an electron-atom, and/or molecule collision mechanism, while that phenomena which is

sensitive to variations in frequency and/or magnetic intensity must be the result of an ion-atom, and/or molecule collision mechanism. Again, refer to Chapter V. It is noted that the data for the lower values of optimum magnetic intensity is essentially independent of frequency and is nearly constant in magnetic intensity. Further, an effect not explained by this simple theory, this data is essentially constant with respect to mass flow and cell pressure. One concludes that this data perhaps represents the effect of an electron-atom, molecule collision mechanism and that the collision frequency for electrons may tend to remain constant over the range of pressures employed. On the other hand, the data for higher values of optimum magnetic intensity varies radically with both frequency and pressure; magnetic intensity is no longer an essential constant. Then, one concludes that this data probably represents the result of an ionatom, molecule collision mechanism and that the collision frequency for ions is likely more affected by the pressure.

<u>Case 3.</u> $(w = w_g)$: This is the condition of gyromagnetic resonance. Equation 2.13 would indicate infinite perpendicular and drift velocities for this condition; but, going back to the solutions for the a's determined for the special case of resonance, one finds that this is not strictly true.

(a) \overline{E}_{11} only: The solutions become:

$$\vec{v}_{11} = \vec{a}_0 \quad \vec{E}_{11} = -q/m \quad j/\omega \quad \vec{E}_{11}$$

$$\vec{v}_1 = \vec{v}_D = 0$$
(2.26)

and no effect is observed due to the applied \overline{B} . As a matter of fact, Equations 2.26 are identical to Equations 2.10, this case when $w \neq w_g$, so that one would predict no gyromagnetic resonance effects when the electric intensity is applied in the parallel direction.

(b) \overline{E}_1 only: The solutions become:

$$\overline{v}_{11} = 0$$

$$\overline{v}_{1} = q/2\omega m (\omega t - j) \overline{E}_{1}$$

$$\overline{v}_{D} = -j (q/m)^{2} t/2\omega \overline{E} \times \overline{B}$$

(2.27)

and it is observed that the velocities become infinite only as t approaches infinity. In cases where particle and/or other collisions do not severely limit the growth of these velocity components with time, relativistic effects would soon become important thereby nullifying the initial assumptions in this analysis.

CHAPTER III

EXPERIMENTAL SYSTEM AND EQUIPMENT

The General Nature of the Experimental System

The general nature of the experimental system is illustrated in the block diagram of Figure 4. Linde commerical tank hydrogen is inletted into the system as shown at the bottom of the figure. The gas passes through a mass flow transducer for a measurement of mass flow (Hastings model LF-300 Mass Flow Meter) on to a servo-driven variable leak (Granville Phillips Series 213). The variable leak is controlled so as to maintain constant pressure in the system by means of a Granville Phillips Company Automatic Pressure Controller with its feedback loop connected to a modified Pirani Gauge (Consolidated Vacuum Corporation type GP-110). From the variable leak, the gas flows into the cell, which contains the electrodes for producing ionization, and then on to the vacuum pump (Welch Scientific Company Model 1397) from which it is exhausted into the outer atmosphere. Gauge tubes No. 3 and 4 are symmetrically located at approximately 18 inches, respectively, from the outlet and inlet to the cell. The mean of these two gauge tube readings, as indicated by the vacuum gauge (Hastings type VT-4 and/or type VT-6), is taken to be the pressure inside the cell. The function of gauge tube No. 1 and its associated vacuum gauge (Hastings type VT-4 and/or type VT-6) is to provide an indication of the vacuum system performance.





As shown in the diagram, the cell is located between the pole pieces of an electromagnet (Harvey-Wells Corporation Model L-76V), the magnetic intensity of which can be varied by means of varying the output current of the magnet power supply (Harvey-Wells Corporation Model UR1050). The electromagnet is calibrated for the applicable gap by means of a Radio Frequency Laboratories Model 1295A Gaussmeter.

In the early stages of the study an E. F. Johnson Company Viking II amateur transmitter was employed as the rf supply and is referred to in this paper as the "old rf supply." However, due to its limitations, it was replaced by the combination of a Technical Material Corporation PAL-1KW Linear Amplifier and a Hewlett-Packard Company Model 606A HF Signal Generator. This combination gives continuous coverage of the 2 to 32 MH_z frequency range with continuously variable output up to one KW maximum. From the transmitter, the rf power is fed through a coaxial cable to the electrodes of the cell. Samples of the rf voltage applied to the electrodes and rf current through the cell are picked up through rf probes and displayed on a dual-trace oscilliscope. A camera may be used to record the oscilloscope traces of voltage and current.

A photograph of the system described above is shown in Figure 5a. From left to right the equipment shown: dual-trace oscilloscope with camera; magnet with cell in place and magnet power supply immediately below, Hastings vacuum gauges, gauge tube switch gear, variable leak, and mass flow transducer are mounted in the black rack; the larger cabinet in front of the black rack is the automatic pressure controller, the smaller cabinet on top is the mass flow meter, and the other cabinet on top is the Pirani gauge; and the rf power source is shown adjacent to the table in the extreme right of the picture. A close-up of a


Figure 5a. Photograph of the Over-All System



Figure 5b. Cell in Operation

Figure 5. Photographs of the Experimental System

"typical" cell (the aluminum cell with secondary electrodes discussed in the following paragraph) in operation is shown in Figure 5b.

Evolution of the Ionization Cell

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The ionization cells passed through several stages of development before satisfactory cells were produced. The difficulties inherent in the design were manifold, but were chiefly concerned with the problem of obtaining a suitable combination of electrical and mechanical properties along with the capability of maintaining the required vacuum. The dimensions and properties of the Harvey-Wells Corporation electromagnet used in the study necessarily limited the geometry of the ionization cells employed. The pole faces are $6\frac{1}{2}$ inches in diameter which means that the dimensions of the cell must be somewhat less than this in order to obtain the desired uniformity of field. Also, for this same reason it is essential that the cells be constructed entirely of non-ferrous materials. As can be seen by referring to Figure 85, in the Appendix, pole gaps of the order of 31/2 inches allow the magnet to attain a magnetic intensity in the vicinity of 5 kilogauss, consequently this dimension was taken as an approximate parameter. When vacuum joints and fittings are taken into account, this selection restricts the electrodes in a symmetrical plane electrode geometry to circular discs of approximately 1 3/4 inches in diameter. This in turn limits the electrode spacing to something of the order of 0.2 inch, or less, in order to obtain a reasonable degree of uniformity in the applied electric field.

Cell bodies of plexiglass were tried early in the study. The appearance and construction of one of these cells is shown in the two







Figure 6b. Disassembled Cell

Figure 6. Photographs of an Early Model Plexiglass-Bodied Ionization Cell

photographs of Figure 6. This form of design proved to be unuseable for several reasons. First, as shown in the photographs, the electrodes were balanced fed which led to a severe balance to unbalance transformation problem in operating the cell. However, this defect was not so apparent at this stage of the study because the plexiglass cell bodies failed so rapidly in operation due to the heat generated inside the cell with ionization taking place that its full effect was not realized. Another defect in this design was the use of "O" rings to obtain the vacuum seal, as shown in the photograph of Figure 6 b. While these were totally satisfactory for the purpose of obtaining the seal, the "O" rings, being compressible, made it virtually impossible to either maintain a precise positioning of the electrodes or to determine accurately the spacing after assembly. The immediate problem being the failure of the plexiglass cell bodies to withstand the heat of operation, it was decided to construct the cell body from teflon. The previous electrode dimensions and material (approximately 1.70 inch diameter brass discs) were retained, as was the balanced feed geometry, but the "O" ring seals were discarded. Instead, the mating surfaces were tediously ground together with rottenstone (decomposed siliceous limestone) and water to form a vacuum tight joint. Although this procedure required considerable painstaking effort, the results proved superior in operation in every way to the use of the "O" rings. This cell, referred to elsewhere in this paper as the "old teflon cell", is shown in the two photographs of Figure 7. It proved to be wholly satisfactory from the mechanical standpoint, but left much to be desired in the way of electrical design. The principal source of electrical difficulties in the design was the balanced feed geometry. The radiation from









Figure 7. Photographs of the Old Teflon Cell, Scale Shown is a 6-inch Engineer's Scale unbalanced currents on the parallel two-wire transmission line feeding the cell was so severe as to preclude data taking at frequencies above 7 MH_z. An air-core balun transformer was constructed and found to satisfactorily reduce the stray radiation from the transmission line, but its use prevented the direct measurement of the phase relationship between the applied voltage and through the cell. Consequently, the next step in cell development was to redesign so as to have coaxial rf connections to the cell, thereby eliminating the parallel two-wire line.

The cell resulting from the modification of the "old teflon cell" to incorporate coaxial rf connections is shown in the exploded view of The dimensions of this cell, hereafter referred to as the Figure 8. "teflon cell", are shown in the shop drawing of Figure 9. The secondary electrodes for sampling drift components of current, as discussed in the following chapter, were added to the cell at this stage of evolution. The primary electrodes were, as in the previous cells, 1.70 inch diameter brass discs. The teflon cell, as shown in Figures 8 and 9 and also in the photograph of Figure 34 in the following chapter, was the first reasonably successful ionization cell produced during the study. Its use figured prominently in the conceptive stages of the experimental study; however, there were still troublesome deficiencies in the electrical design. The "open" nature of the design did not offer complete rf shielding so that again there was trouble from stray rf currents at the higher frequencies, say 15 MH, and above. This difficulty was eliminated at last in the final stage of development which resulted in the aluminum-bodies cell, hereafter referred to as the "aluminun cell", shown in the shop drawing of



Figure 8. An Exploded View of the Teflon Cell



Figure 9. Shop Drawing of the Teflon Cell

 $T^{(N-N)} v^{*}$

ω L





Figure 10. In addition to totally enclosing the cell within an aluminum shell, this design offers the additional feature of a guard ring structure around the electrodes which insures a greater uniformity in the applied electric intensity. The construction of the insulating parts, which also must function as vacuum seals, around the electrodes in this cell constituted something of a problem. Teflon was tried initially, but it was difficult to machine the parts so as to obtain the essential vacuum tightness. Furthermore, this material had a tendency to break down in the intense field regions inside the cell, especially around the guard ring structure, and cause severe etching of electrodes and cell body. After some experimentation, a casting resin (Emerson & Cuming, Inc. Stycast 2662) was found to offer a satisfactory solution. Two of the aluminum-bodied cells were constructed to almost identical dimensions, except that one of these carried secondary electrodes similar to those of the teflon cell shown in Figure 8, while the other one did not have these additional electrodes. The primary electrodes were, in all of the above cell designs, 1.70 inch diameter discs of brass and the spacing between electrodes varied, as listed in Chapter V, from approximately 2 to 6 millimeters.

CHAPTER IV

GENERAL NATURE OF THE OBSERVED PHENOMENA

General Experimental Observations

Certain effects are observed experimentally with hydrogen in the system and the cell in operation. These are as follows:

- (a) In general, the rf energy available is not sufficient to produce deionization glow (and consequently ionization or excitation) with no static magnetic field applied.
- (b) Application of the magnetic field in a direction parallel to the rf electric field has no apparent effect, but application in the perpendicular direction has the striking effect described below.
- (c) With the rf field applied and applying the static magnetic field in the perpendicular direction, one observes a value of magnetic field intensity at which a glow appears in the window of the cell. Increasing the magnetic intensity while holding the rf voltage fixed will cause the glow to become more intense to a point, after which further increasing the magnetic field causes the glow to become increasingly faint until it is finally extinguished. Within the range of the magnet's capability, further increasing the magnetic intensity causes the glow to appear again with the above observations

being repeated. Values of applied magnetic field such that glow intensity is a maximum for a fixed rf field will hereafter be referred to as the optimum magnetic field intensity.

(d) As the rf electric field intensity is varied, the values of optimum magnetic field intensity will shift somewhat but are quite evident so long as the glow is maintained. Under the condition of the minimum rf field intensity which will maintain a steady glow, current and voltage traces as viewed on the oscilloscope are essentially sinusoidal and displaced in phase by 90 degrees. A photograph of typical traces under these conditions is shown in Figure 11 (photograph shown was taken at $f = 3.5 \text{ MH}_z$, MF = 30 std. cc/min., $\overline{B} = 0.85 \text{ Kg}$). As the output of the rf source is increased, the glow becomes more and more intense, and the current trace becomes nonsinusoidal as shown in the photograph of Figure 12. As shown, the voltage waveform remains sinusoidal under both conditions.

As a consequence of these observations, an experimental mapping of the optimum magnetic field under the condition of a minimum steady deionization glow (i.e., ideally, minimum rf applied electric intensity required to maintain ionization) as a function of pressure, magnetic intensity, applied rf voltage, and applied electric intensity was initiated. The results of this undertaking are presented in detail in the following chapter of this paper.



Figure 11. Typical Oscilloscope Traces with Cell Operating Under the Condition of Minimum Deionization Glow



Figure 12. Typical Oscilloscope Traces with Cell Operating Under the Condition of Intense Deionization Glow

Essential Nature of the Measured Data

The essential nature of the measured data is displayed in the three-dimensional plot of Figure 13. Figure 13 shows a three-dimensional plot of the electric intensity values versus optimum magnetic intensity and the cell pressure. The measured current through the cell is also shown. Due to its essentially total reactive character, hereafter the cell current in this measurement will be referred to as the charging current. Now as viewed in Figure 13, the electric intensity and charging current values lie in planes of intersection defined by curves of cell pressure versus optimum magnetic intensity. These curves can and probably should be idealized by straight lines. Figure 14 shows such an idealization of the cell pressure versus optimum magnetic intensity data given in Figure 13. Along each of these lines a new variable has been defined; $\boldsymbol{\gamma}_1$ along the line of lower optimum magnetic intensity values and γ_{γ} along the line of higher optimum magnetic intensity values. Note that the same linear scale has been picked for γ_1 and γ_2 as is used in plotting magnetic intensity and pressure, and the zero pressure axis has been selected as the reference for defining γ_1 , γ_2 equal to zero. Although little use of this technique is made in this paper, one can easily convert the scale of the γ -values to either optimum magnetic intensity or cell pressure readings by writing the equations for the Y-lines. For example, in the case at hand one finds by graphical analysis:

(a) Along the γ_1 line:

B (Kg) =
$$0.52 + 0.43 \gamma_1$$
 (4.1)
P (mm. Hg.) = $1.066 \gamma_1$ (4.2)







(b) Along the γ_2 line:

$$B (Kg) = 1.50 + 0.45 \gamma_2$$
 (4.3)

P (mm. Hg.) = 0.90 γ_2 (4.4)

Then it appears that the three-dimensional nature of the measured, as well as calculated, data can be better displayed on two-dimensional charts by plotting values relative to the lower optimum magnetic intensity versus $\boldsymbol{\gamma}_1$ and values relative to the higher magnetic intensity versus γ_2 . Then equations like Equations 4.1 through 4.4 above can be employed to translate specific Y values into magnetic intensity and pressure. Figure 15 shows such a plot of electric intensity and charging current versus $\boldsymbol{\gamma}_1.$ Note that if these values were plotted against the magnetic field intensity, all points would fall nearly together. On the other hand, a plot versus pressure would be nearly equivalent to the graph shown. Again, a plot versus mass flow results in a nonlinear distortion of the abscissa due to the nonlinear relation between mass flow and pressure. Figure 16 shows a similar plot of electric intensity and charging current corresponding to the higher optimum magnetic field values plotted against γ_{2} . Note that in this case, relative to the graph shown, plotting against mass flow, pressure, or optimum magnetic intensity would distort the figure. This method of presentation will be further applied to computed data in Chapter V of this thesis.

Secondary Electrode Measurements

As indicated in Chapter II of this thesis, applied field values are such throughout the measurements that the drift component of velocity of the electrons should be governed by the equation



Figure 15. Electric Intensity and Charging Gurrent Corresponding to the Lower Optimum Magnetic Intensity Values at 7 MH Plotted Against the γ_1 Variable

41.5



Figure 16. Electric Intensity and Charging Current Corresponding to the Higher Optimum Magnetic Intensity Values at 7 MH₂ Plotted Against the Y₂ Variable

In the experimental system, values of the magnetic intensity \overline{B} are always such that the minimum $|\overline{v}_D/\overline{v}_1|$ ratio for the electrons is of the order of 10^3 to 10^4 ; i.e., it is predicted that the electron motion is essentially in the direction perpendicular to the plane containing \overline{E} and \overline{B} . Further, Equation 4.5 indicates that the magnitude of this velocity should be nearly independent of frequency. However, these conditions do not apply to the ions. The experimental values of ω and \overline{B} are such as to indicate velocity directions and magnitudes which are, generally speaking, somewhat sensitive to variations in both \overline{B} and ω .

 $V_{\rm D} \approx \frac{\overline{E \times B}}{-2}$

In view of the above considerations, it appeared that information of very pertinent character concerning the ionization processes could be found by sampling the drift velocity components. These cannot be measured directly, of course, but it was reasoned that measuring the voltage buildup due to particle bombardment on electrodes placed in the direction of the drift velocity components should give relative data. This was the purpose of the secondary electrodes as shown in the exploded view of the teflon cell in Figure 8. A pictorial diagram of the electrode and applied field configurations is shown in Figure 17. The secondary electrodes are of brass, approximately 3/16-inch in diameter, and are threaded into the teflon body of the cell as far as is feasible without running the risk of electrically shorting the primary electrodes. As recorded previously, the primary electrodes to which the rf voltage is applied are also of brass, approximately 1.70 inches in diameter. For the case of the secondary electrode

(4.5)





measurements discussed below, the primary electrode spacing was 0.169 inch. In Figure 17 this spacing between primary electrodes has been shown greatly exaggerated in order to be able to indicate the appropriate field and velocity directions. Actually, the diameter of the secondary electrodes is slightly greater than the spacing between primary electrodes.

Since experimental data taken with two primary electrodes at 7 MH_ had been the more extensively analyzed and discussed in previous reports, this frequency was selected for the initial measurements taken with secondary electrodes in order to enable comparison with the results of the previous work. The experimental system was exactly the same as that shown in Figures 4 and 5, with the exception of the addition of equipment to view and measure the potentials developed on the secondary electrodes. The electrical connections are shown in Figure 18. Referring to Figure 17, note that the secondary electrode labeled "front" is located adjacent to the hydrogen inlet while the one labeled "back" is located adjacent to the hydrogen outlet. That is, the gas flow velocity is directed from the front toward the back secondary electrode. In Figure 18, two vacuum tube voltmeters are shown connected so as to measure the respective secondary electrode voltages with respect to ground, and a dual trace oscilloscope is connected to enable one to view the nature of the voltages developed. Under the condition of optimum applied static magnetic intensity with minimal rf excitation, these voltages were always found to be as nearly pure dc as could be ascertained from the traces displayed on the oscilloscope. Their essential character remained dc at all other levels of rf excitation and values of B investigated, but when



Figure 18. Electrical Circuit Showing Added Connections for Measuring the Voltages Developed at the Secondary Electrodes

conditions were such as to produce an observed flicker in the deionization glow, this would result in an apparently periodic trace of widely varying waveforms superimposed upon the dc. The level of this "flicker trace" was generally considerably lower than that of the dc, its waveform never approached anything remotely similar to a sinusoid, and its period was always observed to be considerably greater than that of the applied rf frequency.

The dc potentials, as observed on the two vacuum-tube voltmeters, prove to be effective indicators of the optimum magnetic field intensity. With the rf level held constant, the dc voltages developed on the secondary electrodes peak quite sharply as the magnetic field is varied through its optimum value as indicated by observing the glow intensity. That is, as near as one can determine with the present experimental setup, the points of maximum deionization glow and the peak dc levels developed on the secondary electrodes coincide.

This data at 7 MH_z was taken in the manner of that reported previously, with the addition of the secondary electrode measurements. A plot of the cell pressure versus the optimum magnetic intensity is shown in Figure 19. This plot follows the procedure established of idealizing this data to γ_1 and γ_2 lines. Figure 20 shows a plot of the dc voltages measured with respect to ground on the secondary electrodes along the γ_1 line. The γ_1 line corresponds to the lower values of optimum B. As the cell is operated, the secondary electrodes blacken slightly on portions of their surfaces. Since this discoloration requires some effort with metal polish to remove, it is believed to be the result of a slight oxidation rather than the deposit of foreign matter. Reference to the point in Figure 20 where the







Figure 20. The DC Voltage Developed on the Secondary Electrodes Along the γ_1 Line

electrodes were cleaned vividly illustrates the effect. The dc voltages developed on the secondary electrodes are greatly reduced by any amount of discoloration, no matter how faint. Consequently, this data is impossible to repeat unless one insures that the secondary electrodes are always maintained in the same condition. This would appear to necessitate cleaning and polishing these electrodes prior to each measurement, a procedure which the construction of the cell would not tolerate. For this reason, the data was taken by employing the scheme of alternately measuring successive points along the γ_1 and γ_2 lines so as to obtain comparative data. The plot of the voltages measured on the secondary electrodes along the γ_2 line is shown in Figure 21.

In Chapter II, as a result of the analysis, it has been proposed that an electron collision mechanism is primarily responsible for the ionization at lower values of optimum \overline{B} , i.e., along the γ_1 line, and that an atomic ion, H^+ , collision mechanism is the principle mode of ionization at the higher optimum \overline{B} values, i.e., along the γ_2 line. While this experiment cannot identify the ionization mechanisms, reference to Figures 20 and 21 does establish that ionization along the $\boldsymbol{\gamma}_1$ line is a phenomenon which is distinctly different from that produced along the γ_2 line. The higher dc voltages developed at the secondary electrodes along the γ_1 line, as shown in Figure 20, would tend to indicate a mechanism associated with the more mobile particles, i.e., the electrons with their high q/m ratio. The higher potential developed at the back electrode can be explained by its location; it is adjacent to the hydrogen outlet. Hence, the gas flow velocity would tend to direct the larger number of charged particles in its direction. Our simple theory does not, however, explain why the





potentials should be of opposite polarity, i.e., the potential developed on the front electrode positive and that on the back electrode negative with respect to ground everywhere along the $\boldsymbol{\gamma}_1$ line. Refer to Equations 2.13 in Chapter II for the drift velocity. Along the γ_1 line the angular frequency of gyration for the electron is many times greater than that of the applied rf, but the opposite is the case for the atomic and molecular ions so that \overline{v}_{D} for the ions in the system is in the opposite direction to that for the electrons. A like condition exists for the perpendicular velocity components (again, see Equations 2.13 in Chapter II). That this effect may influence the situation with respect to the sign of the secondary electrode potentials, is substantiated by reference to Figure 21. Here, along the γ_2 line, the conditions described above still apply to the electrons and molecular ions, but the atomic ions undergo gyromagnetic resonance. Prior to resonance, their velocity components are directed opposite to those of the electrons, but after resonance the directions are reversed. In Figure 21, we note that a potential reversal occurs on the front electrode and that the potentials remain opposite from that point on. This occurs in the vicinity of the predicted gyromagnetic resonance of the atomic hydrogen ions.

The problem of contamination precluded the possibility of making extensive meaningful measurements with the secondary electrodes. However, their use in the measurements decribed above, coupled with periodic monitoring of the potentials developed on them as the experimental optimum magnetic intensity mapping progressed did aid in establishing a definite conclusion with respect to the minimum deionization glow condition. The basic experimental observation is that with deionization glow present in the cell, dc potentials are developed on the secondary electrodes. This is true no matter how weak the glow intensity. However, in the absence of any glow there was never any observable secondary electrode potentials irrespective of any other conditions. On the basis of these observations it was concluded that the presence of the glow is essential to establishing any substantial ordered motion of charge carriers. In these terms, the minimum glow condition described at the beginning of this section may be interpreted as the result of the minimum rf excitation which will maintain ordered motion of carriers. All of the optimum magnetic intensity data presented in this paper was taken under this condition.

Effects of Variations in Primary Electrode Spacing

An effort was made to correlate the measurements made employing the teflon cell with those taken with the aluminum cell. The data was taken with different primary electrode spacings in the two cells, but with careful attention to insure that the frequencies were the same. The electrode spacing for the aluminum cell was 0.002794 meters, and it was 0.00208 meters for the teflon cell. Refer to Figures 22 and 25. We note that the γ lines as determined for the two cells at a frequency of 7 MH_z fall at essentially the same place. Now, refer to Figures 28 and 31. Here we note that the γ_1 lines fall in essentially the same place, but there is a substantial deviation in the slope of the γ_2 lines. Further, note that the γ lines for the aluminum-bodied cell at 8 MH_z have remained practically the same as at 7 MH_z. In other words, only the γ_2 line for the teflon-bodied cell has radically shifted in going from 7 to 8 MH_z. One would





Figure 23. Applied Voltage, Charging Current, and Electric Intensity Corresponding to the Lower Magnetic Intensity Values at 7 MH Shown Plotted Against the Y₁ Variable ² for the Aluminum Cell











Figure 26. Applied Voltage, Charging Current, and Electric Intensity Corresponding to the Lower Magnetic Intensity Values at 7 MH Shown Plotted Against the γ_1 Variable ^Z for the Teflon Cell




expect the field structure in the teflon-bodied cell to be more subject to change with frequency than that of the aluminum-bodied cell. This is because the teflon-bodied cell is of more "open" design, while the aluminum-bodied cell is carefully shielded and incorporates a guard ring structure. On the basis of these observations it is concluded that these deviations are probably due to a small frequency difference in the two sources. It also became evident that larger variations with respect to frequency are to be expected in an "open"

Refer to Figures 23, 24, 26 and 27. Note that at 7 MH_z the values of applied voltage and charging current are essentially the same for the two cells despite the difference in electrode spacing. The applied electric intensity is calculated by dividing the applied voltage by the electrode spacing. Consequently it appears at quite different levels for the two cells. This would infer, on the basis of the previously presented theory, that while all other things were nearly equal, the particle velocities in the teflon-bodied cell were much larger than in the aluminum-bodied cell. This, of course, is not reasonable. Now, refer to Figures 29, 30, 32 and 33. At 8 MH_z we noted above that the slopes of γ_2 lines were quite different for the two cells, and here we see that this evidently implies quite different applied voltage and charging current characteristics.

Again, refer to the 7 MH_z curves, Figures 22 through 27. The γ line, applied voltage, and charging current data for the two cells are essentially the same despite the fact that the electrode spacings were considerably different. Since the data is taken under conditions where the applied voltage and current through the cell are 90 degrees







Figure 29. Applied Voltage, Charging Current, and Electric Intensity Corresponding to the Lower Magnetic Intensity Values at 8 MH Shown Plotted Against the γ_1 Variable for the Åluminum Cell



















Figure 34. A Photograph Showing the Teflon Cell Installed the Electromagnet



Figure 35. A Photomacrograph Showing the Darkening shich Appeared on the Secondary Electrode Surfaces (Actual Size is 3/16 Inch Diameter) Upon Removal from the Teflon-Bodied Cell



Figure 36. A Photomacrograph Showing an Enlarged View of the Back Secondary Electrode Alone



Figure 37. A Photomacrograph Showing the Deionization Glow as it Appears in the Cell Window

out of phase, so far as can be determined by the dual-trace oscilloscope, and hence we refer to this current as the charging current, the ratio of applied voltage to charging current represents the capacitative reactance of the cell. The indication from the 7 MH data is that this reactance, and hence the capacitance of the two cells, did not change with electrode spacing. It is believe that this effect can only be explained in terms of a space-charge layer, or layers, lying in the region between the primary electrodes which tend to become thicker with increased electrode spacing. Velocity computations and the photomacrographs of Figures 36 and 37 tend to support this view. Velocity computations based upon the previously reported theory and the applied electric intensity computation have consistently given lower kinetic energies than required to produce ionization by collision. This could be justified in terms of space charge. Figures 36 and 37 suggest the nature of such a space charge. Figure 36 is a photomacrograph of the face of the back electrode (3/16 inch diameter actual size) as it appeared upon removal from the teflon-bodied cell at the conclusion of the data taking. This is the secondary electrode which is least disturbed by adjacent attachments to the cell. Note that the darkened portions of the surface, which indicate regions of particle bombardment, tend to describe two lines separated by a region in which no darkening occurred. This indicates layers of ordered particle activity parallel and adjacent to the surfaces of the primary electrodes and that these layers are separated by a central layer in which no such ordered activity occurs. It is believed that this central region represents the space-charge region. Figure 37 is a photomacrograph of the deionization glow as viewed through the cell window.

Much difficulty was experienced in obtaining this photograph due to vibrations and mechanical restrictions; consequently, it leaves much to be desired in clarity. However, note that the glow fades out to give two dark layers which are a fraction of a millimeter in thickness at the surfaces of the primary electrodes. These dark layers correspond at least approximately to the darkened lines shown on the surface of the back secondary electrode in Figure 36.

Voltage and Current Traces at 2 MH,

In the previous paragraphs of this Chapter, much mention has been made of the fact that the applied voltage and current through the cell were both linear and 90 degrees out of phase (so far as could be determined by the methods employed to measure) under the minimum glow condition where data was taken. However, this condition did not quite hold true at 2 MH_. An example of the phenomenon observed at this frequency is illustrated in Figures 38 and 39. Figure 38 shows that voltage and current waveforms were linear and 90 degrees out of phase with no gas in the cell. Figure 39 shows the voltage and current waveforms with gas flowing into the cell at 20 std. cc/m and under the best minimum glow condition which could be obtained. Note that although the phase displacement appears to be essentially the same as before, a non-linearity has been introduced into the current waveform. This condition tended to become slightly more prominant following section was taken under the condition described above. However, this effect disappeared quite rapidly with increasing frequency. The data at 8 MH and above was taken under the previously described

conditions with current and voltage waveforms as illustrated in

Figure 11.



Figure 38. Voltage and Current Traces as Observed at 2 MH_z with no Gases in the Cell



Figure 39. Voltage and Current Traces as Observed at 2 MH with a Mass Flow of 20 std. cc/min Under the Condition of Best Obtainable Minimum Glow

CHAPTER V

AN OPTIMUM MAGNETIC INTENSITY MAPPING OF HYDROGEN

This chapter presents the optimum magnetic intensity mapping of hydrogen discussed in the preceding chapters of this thesis. The results of all measurements taken throughout the course of the study are given, listed by frequency, starting at 2 MH_, in chronological order. This approach adds considerable bulk to the thesis, however, after much thought was deemed essential in order to show as clearly as possible the true nature of the measured phenomena and the extent to which agreement was and was not found with the predictions of the analysis. For example, reference to the 3.5 MH data contained in Figures 44 through 46 gives a comparison between the old teflon cell, with its relatively unshielded open structure and balanced feed geometry, and the fully shielded, coaxially fed aluminum cell with a guard ring structure. It is seen that there is relatively little difference between the results from the two cells at this low frequency, so long as the electrode spacing is nearly the same; consequently one might assume a relative freedom from precise geometric considerations in the design of some device employing these results in this frequency region. The extensive data presented for 7 MH, also serves a like purpose. Comparing Figures 53 and 54 with Figure 55, one notes a much more pronounced difference between the data obtained from these two cells even though the electrode spacing

is nearly the same. This would indicate the need for more care in the geometric design of devices operating in this frequency region. Also, the effects of varying the electrode spacing are more thoroughly investigated at the 7 MH_z and 8 MH_z frequencies. And so the list continues. Where the objective of a study is to determine the experimental bases of devices not yet conceived, one is quite helpless in deciding which data are superfluous.

In obtaining the data presented in this chapter the optimum magnetic intensities were determined by experimental observation, as discussed in Chapter IV, while holding the cell pressure at some constant value. After the optimum magnetic intensity was determined at minimum glow, this condition was held while the applied voltage and charging current measurements were made. This raw data was then processed on the IBM 7040 digital computer as discussed in the Appendix. At the same time, computations of the charged particle velocities were made in accordance with the analysis of Chapter II, Equations 2.13 through 2.21. This computer program is given in Table I of the Appendix. All of the graphs shown in this chapter were drawn by a combination of the IBM 1620 digital computer with disc storage and the Calcomp 565 automatic plotter with liquid ink conversion. It was found essential to use india ink in order to obtain satisfactory photographic reproduction. The Y-lines of the graphs shown in this chapter were all computed as a least squares fit to the appropriate sets of points by the IBM 1620 as a portion of the plotting program. Slightly over 40 hours actual computer time was required in order to plot the graphs. This program is given in Table III of the Appendix. The caption information listed on the graphs employs the following

D - is the primary electrode spacing in meters.

Aluminum Cell - refers to the Aluminum Cell as shown in the shop

drawing of Figure 10.

Aluminum Cell No. 2 - refers to the nearly identical cell with secondary electrodes.

Old Teflon Cell - refers to the first model teflon-bodied cell with balanced-line fed primary electrodes.

Teflon Cell - refers to the Teflon Cell as shown in Figures 8

and 9.

Old rf Source - refers to the E. F. Johnson Company Viking II

amateur transmitter employed early in the study. Where no mention is made with respect to the rf source, the combination of the Technical Material Corporation Pal - 1KW Linear Amplifier and Hewlett-Packard Company Model 606 A HF Signal Generator was used. The greater credibility is to be given to data taken with either of the Aluminum Cells and the linear amplifier - signal generator combination.



Figure 40a. Data taken June 27, 1966 at 2 MH with the Aluminum Cell, D = 0.002794 meters.











Figure 41a. Data taken June 30, 1966 at 2 MH with the Aluminum Cell, D = 0.002794 meters











Figure 42a. Data taken July 2, 1966 at 3 MH with the Aluminum Cell, D = 0.002794 meters















Figure 43b. Data taken August 24, 1966 at 3 MH with the Aluminum Cell, D = 0.002794 meters







Figure 43d. Data taken August 24, 1966 at 3 MH with the Aluminum Cell, $D = {}^{2}0.002794 \text{ meters}$



Figure 43e. Data taken August 24, 1966 at 3 MH with the Aluminum Cell, $D = {}^{2}0.002794$ meters



Figure 44a. Data taken April 15, 1965 at 3.5 MH with the Old Teflon Cell and Old rf Source, $D^{Z} = 0.004292$ meters











Figure 44d. Data taken April 15, 1965 at 3.5 MH with the Old Teflon Cell and Old rf Source, D = 0.004292 meters 9.3



Figure 44e. Data taken April 15, 1965 at 3.5 MH with the Old Teflon Cell and Old rf Source, D = 0.004292 meters






Figure 45b. Data taken April 20, 1965 at 3.5 MH with the Old Teflon Cell and Old rf Source, D z = 0.004292 meters







Figure 45d. Data taken April 20, 1965 at 3.5 MH with the Old Teflon Cell and Old rf Source, D = 0.004292 meters



Figure 45e. Data taken April 20, 1965 at 3.5 MH with the Old Reflon Cell and Old rf Source, D = 0.004292 meters



Figure 46a. Data taken August 16, 1965 at 3.5 MH with the Aluminum Cell and Old rf Source, D^{Z} = 0.005937 meters



Figure 46b. Data taken August 16, 1965 at 3.5 MH with the Aluminum Cell and Old rf Source, D²= 0.003937 meters







Figure 46d. Data taken August 16, 1965 at 3.5 MH with the Aluminum Cell and Old rf Source, D = 0.003937 meters



Figure 46e. Data taken August 16, 1965 at 3.5 MH with the Aluminum Gell and Old rf Source, D = 0.003937 meters

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Figure 47a. Data taken August 9, 1966 at 4 MH with the Aluminum Cell, D = 0.002794 meters







Figure 47c. Data taken August 9, 1966 at 4 MH with the Aluminum Cell, D = 0.002794 meters



Figure 47d. Data taken August 9, 1966 at 4 MH with the Aluminum Gell, D $\stackrel{\mathbb{Z}}{=}$ 0.002794 meters







Figure 48a. Data taken August 23, 1966 at 4 MH with the Aluminum Cell, D = 0.

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Figure 48d. Data taken August 23, 1966 at 4 MH with the Aluminum Cell, $D = \overline{0.002794}$ meters



Figure 48e. Data taken August 23, 1966 at 4 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters



Figure 49a. Data taken June 24, 1966 at 5 MH with the Aluminum Cell, D = 0.002794 meters



Figure 49b. Data taken June 24, 1966 at 5 MH with the Aluminum Cell, D = 0.002794 meters



Figure 49c. Data taken June 24, 1966 at 5 MH with the Aluminum Cell, D = 0.002794 meters







Figure 49e. Data taken June 24, 1966 at 5 MH with the Aluminum Cell, D $\stackrel{\Xi}{=}$ 0.002794 meters



Figure 50a. Data taken June 29, 1966 at 5 MH with the Aluminum Cell, D = 0.002794 meters



Figure 50b. Data taken June 29, 1966 at 5 MH with the Aluminum Cell, D = 0.002794 meters







Figure 50d. Data taken June 29, 1966 at 5 MH with the Aluminum Cell, D $\stackrel{\mathbb{Z}}{=}$ 0.002794 meters



Figure 50e. Data taken June 29, 1966 at 5 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters



Figure 51a. Data taken July 9, 1966 at 6 MH with the Aluminum Cell, D = 0.02794 meters



Figure 51b. Data taken July 9, 1966 at 6 MH with the Aluminum Cell, D = 0.002794 meters



Figure 51c. Data taken July 9, 1966 at 6 $\frac{MR}{Z}$ with the Aluminum Cell, D = 0.002794 meters







Figure 51e. Data taken July 9, 1966 at 6 MH with the Aluminum Cell, D $\stackrel{\mathbb{Z}}{=}$ 0.002794 meters

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Figure 52a. Data taken August 18, 1966 at 6 MH with the Aluminum Cell, D = 0.002794 meters





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Figure 52c. Data taken August 18, 1966 at 6 MH with the Aluminum Cell, D = 0.002794 meters







Figure 52e. Data taken August 18, 1966 at 6 MH with the Aluminum Cell, D $\stackrel{\mathbb{Z}}{=}$ 0.002794 meters















Figure 53d. Data taken April 13, 1965 at 7 MH with the Old Teflon Cell and Old rf Source, D = 0.004292 meters







Figure 54a. Data taken April 16, 1965 at 7 MH with the Old Teflon Cell and Old rf Source, \vec{D} = 0.004292 meters

















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Figure 55a. Data taken September 1, 1965 at 7 MH with the Aluminum Cell and Old rf Source, $D^2 = 0.003937$ meters



Figure 55b. Data taken September 1, 1965 at 7 MH with the Aluminum Cell and Old rf Source, D^Z = 0.003937 meters











Figure 55e. Data taken September 1, 1965 at: 7 MH with the Aluminum Cell and 51d rf Source, D = 0.003937 meters



Figure 56a. Data taken March 2, 1966 at 7 MH with the Aluminum Cell, D = 0.007902 meters













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Figure 56e. Data taken March 2, 1966 at 7 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.007902$ meters



Figure 57a. Data taken March 2 and 4, 1966 with the Teflon Cell, D = 0.005879 meters

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Figure 57b. Data taken March 2 and 4, 1966 with the Teflon Cell, D = 0.005879 meters

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Figure 57c. Data taken March 2 and 4, 1966 with the Teflon Cell, D = 0.005879 meters



Figure 57d. Data taken March 2 and 4, 1966 with the Teflon Cell, D = 0.005879 meters 158



Figure 57e. Data taken March 2 and 4, 1966 with the Teflon Cell, D = 0.005879 meters



Figure 58a. Data taken September 2, 1966 at 7 MH with the Aluminum Gell No. 2, D = 0.005931 meters















Figure 58e. Data taken September 2, 1966 at 7 MH with Aluminum Cell No. 2, $D^2 = 0.005931$ meters



Figure 59a. Data taken September 7, 1966 at 7 MR with Aluminum Cell No. 2, D = 0.005931 meters










Figure 59d. Data taken September 7, 1966 at 7 MH with Aluminum Cell No. 2, D^Z= 0.005931 meters















Figure 60c. Data taken April 29, 1966 at 8 MH with the Teflon Cell, D = 0.00208 meters²

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Figure 60d. Data taken April 29, 1966 at 8 MH with the Teflon Cell, D = 0.²00208 meters







Figure 61a. Data taken May 2, 1966 at 8 MH with the Aluminum Cell, D = 0.002794 meters



Figure 61b. Data taken May 2, 1966 at 8 MH with the Aluminum Cell, D = 0.002794 meters



Figure 61c. Data taken May 2, 1966 at 8 MH with the Aluminum Cell, D = 0.002794 meters



Figure 61d. Data taken May 2, 1966 at 8 MH with the Aluminum Cell, $D = \frac{z}{0.002794}$ meters







Figure 62a. Data taken May 24, 1966 at 8 MH with the Aluminum Cell, D = 0.002794 meters



Figure 62b. Data taken May 24, 1966 at 8 MH with the Aluminum Cell, D = 0.002794 meters





10,000,000 8,000,000 7,000,000 6,000,000 5,000,000 4.000,000 3,000,000 2,000.000 $v_e v_{De}$ 1,000,000 800,000 700,000 600,000 500,000 400,000 300,000 + v_1 v_H1 200,000 24 V_{H2} ,⁺v_{1H2}+ 100,000 80,000 70,000 60,000 50.000 ---40,000 30.000 V_{DH}+ 20,000 8-9-4 10.000 8,000 7,000 6,000 5,000 4.000 second V_{DH2}+ -94 meters per 3,000 . 2,000 Particle Velocity in 1,000 800 700 600 500 1e 400 300 200 100 80 70 60 50 40 30 20 10 I 7.0 2.0 1.0 0.0 5.0 4.0 3.0 6.0 γ_1









Figure 63a. Data taken July 20, 1966 at 9 MH with the Aluminum Cell, D = 0.002794 meters



Figure 63b. Data taken July 20, 1966 at 9 MH with the Aluminum Cell, D = 0.002794 meters ²



Figure 63c. Data taken July 20, 1966 at 9 MH with the Aluminum Cell, D = 0.002794 meters



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Figure 64a. Data taken August 17, 1966 at 9 MH with the Aluminum . Cell, D = 0.002794 meters





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Figure 64d. Data taken August 17, 1966 at 9 MH with the Aluminum Cell, D = 0.002794 meters



Figure 64e. Data taken August 17, 1966 at 9 MH with the Aluminum Cell, D = 0.002794 meters



Figure 65a. Data taken May 26, 1966 at 10 MH with the Aluminum Cell, D = 0.002794 meters











Figure 65d. Data taken May 26, 1966 at 10 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters



Figure 65e. Data taken May 26, 1966 at 10 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters



Figure 66a. Data taken June 16, 1966 at 10 MH with the Aluminum Cell, D = 0.002794 meters ^Z











Figure 66d. Data taken June 16, 1966 at 10 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters
10,000,000 8,000,000 7,000,000 6,000,000 5,000,000 4,000,000 3,000,000 2,000.000 1,000,000 800,000 700,000 600,000 500,000 400,000 300,000 ve v_{De} 200,000 V_H + v.+ 100.000 80,000 70,000 60,000 50.000 40,000 v_{H2} 4 Ð 30,000 VDH 20,000 v_{DH2} + second 10.000 8,000 7,000 6,000 5,000 ГЭ С 4.000 3,000 meters 2,000 <u>2</u> 1,000 800 700 600 500 Particle Velocity 400 300 200 100 80 70 60 50 40 30 20 vje ٠ 10 7.0 4.0 3.0 2.0 1.0 0.0 6.0 5.0 γ_2





Figure 67a. Data taken July 22, 1966 at 11 MH with the Aluminum Cell, D = 0.002794 meters







Figure 67c. Data taken July 22, 1966 at 11 MH with the Aluminum Cell, D = 0.002794 meters



Figure 67d. Data taken July 22, 1966 at 11 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters







Figure 68a. Data taken August 11, 1966 at 11 MH with the Aluminum Cell, D = 0.002794 meters















Figure 68e. Data taken August 11, 1966 at 11 MH with the Aluminum Cell, D = 0.002794 meters



Figure 68f. Data taken August 11, 1966 at 11 MH with the Aluminum Cell, D = 0.002794 meters



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Figure 69a. Data taken July 14, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters $^{\rm Z}$



Figure 69b. Data taken July 14, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters



Figure 69c. Data taken July 14, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters











Figure 69f. Data taken July 14, 1966 at 12 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters



Figure 69g. Data taken July 14, 1966 at 12 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters



Figure 70a. Data taken August 1, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters



Figure 70b. Data taken August 1, 1966 at 12 MH with the Aluminum Gell, D = 0.002794 meters







Figure 70d. Data taken August 1, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters



Figure 70e. Data taken August 1, 1966 at 12 MH with the Aluminum Cell, D = 0.002794 meters







Figure 70g. Data taken August 1, 1966 at 12 MH with the Aluminum Cell, D $\stackrel{\mathbb{Z}}{=}$ 0.002794 meters



Figure 71a. Data taken July 29, 1966 at 13 MH with the Aluminum Cell, D = 0.002794 meters



Figure 71b. Data taken July 29, 1966 at 13 MH with the Aluminum Cell, D = 0.002794 meters



Figure 71c. Data taken July 29, 1966 at 13 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters



Figure 72a. Data taken August 22, 1966 at 13 MH with the Aluminum Cell, D = 0.002794 meters ^Z



Figure 72b. Data taken August 22, 1966 at 13 MH with the Aluminum Cell, D = 0.002794 meters











Figure 72e. Data taken August 22, 1966 at 13 MH with the Aluminum Cell, D = 0.02794 meters

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Figure 73a. Data taken August 22, 1966 at 13 MH with the Aluminum Cell, D = 0.002794 meters

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Figure 73b. Data taken August 22, 1966 at 13 MH with the Aluminum Cell, $D = 0.002794 \text{ meters}^Z$





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Figure 74a. Data taken August 8, 1966 at 14 MH with the Aluminum Cell, D = 0.002794 meters $^{\rm Z}$











Figure 74d. Data taken August 8, 1966 at 14 MH with the Aluminum Cell, D = 0.002794 meters



Figure 74e.) Data taken August 8, 1966 at 14 $D \stackrel{\text{\tiny E}}{=} 0.002794$ meters



Figure 75a. Data taken August 15, 1966 at 14 MH with the Aluminum Cell, D = 0.002794 meters







Figure 75c. Data taken August 15, 1966 at 14 MH with the Aluminum Cell, D = 0.002794 meters



Figure 76a. Data taken June 15, 1966 at 15 MH with the Aluminum Cell, D = 0.002794 meters







Figure 76c. Data taken June 15, 1966 at 15 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters



Figure 77a. Data taken June 20, 1966 at 15 MH with the Aluminum Cell, D = 0.002794 meters



Figure 77b. Data taken June 20, 1966 at 15 MH with the Aluminum Cell, D = 0.002794 meters







Figure 77d. Data taken June 20, 1966 at 15 MH with the Aluminum Cell, D $\stackrel{\underline{Z}}{=}$ 0.002794 meters



Figure 77e. Data taken June 20, 1966 at 15 MH with the Aluminum Cell, $D \stackrel{Z}{=} 0.002794$ meters



1.500.50

Figure 78a. Data taken May 27, 1966 at 20 MH with the Aluminum Cell, D = 0.002794 meters $^{\rm Z}$



Figure 78b. Data taken May 27, 1966 at 20 MH with the Aluminum Cell, D = 0.002794 meters



Figure 78c. Data taken May 27, 1966 at 20 MH with the Aluminum Cell, D $\stackrel{Z}{=}$ 0.002794 meters



Figure 79a. Data taken June 13, 1966 at 20 MH with the Aluminum Cell, D = 0.002794 meters







Figure 79c. Data taken June 13, 1966 at 20 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters

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Figure 80a. Data taken September 6, 1966 at 20 MH with the Aluminum Cell No. 2, D = 0.0059309 meters









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Figure 81a. Data taken September 9, 1966 at 20 MH with Aluminum Cell No. 2, D = 0.0059309 meters $^{\rm Z}$







Figure 81c. Data taken September 9, 1966 at 20 MH with Aluminum Gell No. 2, D $\stackrel{Z}{=}$ 0.0059309 meters



Figure 82a. Data taken June 10, 1966 at 25 MH with the Aluminum Cell, D = 0.002794 meters



Figure 82b. Data taken June 10, 1966 at 25 MH with the Aluminum Cell, D = 0.002794 meters

n an _{Ma}tri a const



Figure 82c. Data taken June 10, 1966 at 25 MH with the Aluminum Cell, D $\stackrel{\text{\@}}{=}$ 0.002794 meters



Figure 83a. Data taken June 21, 1966 at 25 MH with the Aluminum Cell, D = 0.002794 meters

04.5



Figure 83b. Data taken June 21, 1966 at 25 MH with the Aluminum Cell, D = 0.002794 meters



Figure 83c. Data taken June 21, 1966 at 25 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters


Figure 84a. Data taken May 30, 1966 at 30 MH with the Aluminum Cell, D = 0.002794 meters











Figure 85a. Data taken June 9, 1966 at 30 MH with the Aluminum Cell, D = 0.002794 meters







Figure 85c. Data taken June 9, 1966 at 30 MH with the Aluminum Cell, $D \stackrel{\mathbb{Z}}{=} 0.002794$ meters

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CHAPTER VI

DISCUSSION AND CONCLUSIONS

As illustrated in Figures 40 through 62, pages 76 through 184 of the preceding chapter, the measured data up to approximately 9 MH_ appears to fit the simple analysis presented in Chapter II to a reasonable degree. This is evidenced by the velocity plots, where it is seen that the electron velocity components V $_{\!\!\!\rho}$ and V $_{\!\!\!D_{\!\!\!\rho}}$ are substantially greater than the other components along the $\gamma_1\text{-line},$ while the V_{H}^{+} components and/or the $V_{H_{2}^{+}}$ components are the larger along the γ_2 -line. A feature of these graphs which is not explained by the theory is that the molecular ion, H_0^+ , rather than the atomic ion, H^+ , is in the vicinity of gyromagnetic resonance along the γ_2 -line at the 3 and 4 MHz frequencies. The vicinity of gyromagnetic resonance is indicated by a steep slope at high values on the velocity curve, e.g., Figure 45 e., page 99. The computer, of course, cannot compute an infinity, so the facility for truncating the poles in the velocity curves was built into the program. Then, poles indicating gyromagnetic resonance appear as triangular spikes in the curves, such as V_{H} + in Figure 54 e., page 144. At 5 MH_z we see the indication of a transition from H_{2} + to H^{+} in Figures 49 e. and 50 e., pages 119 and 124. Here, the indication is that H_2^+ is the higher velocity particle at lower values of \overline{B} and pressure while H^{+} is the faster at higher values of \overline{B} and pressure. At 6 MH_z, this transition appears

to be complete, since as indicated by Figures 51 e. and 52 e., pages 129 and 134 respectively, H^{+} is the higher velocity particle. We know of no reason why one should predict that the gas would behave in this manner. However, the above observations from the velocity plots does appear to clear up what was previously thought to be a discrepancy in the theory. Assuming the γ_2 -line to always be due to the H⁺ particle, one would predict on the basis of Equations 2.13, 2.18, and 2.20 of Chapter II that the γ -lines should become increasingly farther apart with increasing excitation frequency. However, as Figures 40 a., 41 a., 42 a., , 52 a., indicate, this is not the case. This may well be due to inability to distinguish between a γ_2 -line brought about by a H^+ collision mechanism and a γ_3 -line brought about by a H_2^+ collision mechanism in this frequency region because of sensitivity limitations in the measuring apparatus. On the other hand, it is equally possible that the two collision mechanisms are so intermingled here that no clear cut distinction can be made. At any rate, a study of the velocity plots does appear to explain why the γ -lines tend to become closer rather that farther apart with increasing frequency.

The effects of primary electrode spacing were discussed in Chapter IV and attributed to space charge, which has been neglected in the theory. One may view the effect upon the velocity computations well in the 7 MH_z plots, Figures 53 to 59, pages 135 to 165, where measurements were performed with a variety of electrode spacings. For spacings up to about 4 millimeters, the gyromagnetic resonance of the H⁺ particle appears on the γ_2 -line, as indicated by Figure 52 e., page 134. However, at spacings of approximately 6 millimeters the

electron is computed to be the faster particle on the $\gamma_2\text{-line,}$ as indicated by Figure 59 e., page 169. Several effects not accounted for in the computations are possibly coming into play here. First, the presence of a space charge would make the true value of electric intensity much different than the computed value used in the velocity computations. Further, it may be that the character of the space charge, such as stratification, etc., changes with electrode spacing so that even relative comparisons of velocity changes with electrode spacings cannot be made. Then, there is the question of the uniformity of the applied electric intensity. Increasing the electrode spacing will naturally affect the uniformity of the applied electric intensity in an adverse manner, but we have no way of determining to what degree and, in addition, no estimate as to what effects to expect with field non-uniformities. It is problematical as to whether this is a true indication of changes in the velocity patterns with change in electrode spacing, or merely a failure in the mathematical model to take a sufficient number of variables into account.

At 9 MH_z we begin to again observe a transition between H^+ and the electron along the γ_2 -line, as indicated by Figures 63 e. and 64 e., pages 189 and 194. Here, the H^+ velocity is computed to be the larger for the lower values of $\overline{\text{B}}$ and pressure while the electron velocity is greater for larger values of $\overline{\text{B}}$ and pressure. Since this occurs at the approximate 2 millimeters electrode spacing, we have more faith in the indication here than the similar effect at 7 MH_z discussed above. The 10 MH_z data, Figures 65 e. and 66 e., pages 199 and 204, shows this same characteristic along the γ_2 -line. At 11 MH_z , Figures 67 a. to 68 g., pages 215 to 216, we begin to pick up a third

 γ -line. This appears at the extreme limit of the magnet's capability and, consequently, was missed in taking the first set of data, so it does not appear in Figure 67. The points shown on the γ_3 -line in Figure 68 a. are much in doubt due to the limitations of the magnet and the slope of the line is to be taken as definitely in error. At 12 MH_z, Figures 69 a. to 70 g., pages 217 to 230, the γ_3 -line is well established. The computed indication is that the electron velocity is larger throughout these measurements with the H⁺ velocity being nearly as great on the γ_3 -line. This is somewhat diverse to what would be expected in the way of charged particle velocities in terms of the theory.

As the excitation frequency increases, we begin to lose Y-lines within the range that data can be taken with the present system. At 13 and 14 MH₂, Figures 71 a. to 75 c., pages 231 to 249, we begin to lose the γ_η -line. This is thought to be due to the insensitivity of the magnet to small changes at very low values of supply current. At 15 MH, Figures 76 a. to 77 e., pages 250 to 257, we tend to lose both the γ_1 and γ_3 -lines. This trend continues for the remainder of the data, out to 30 MH, and is thought to be due to limits in range and control of the magnet at both the high and low extremes of its scale. The electron velocity is computed to be the larger throughout this range in excitation frequencies. An exception to this velocity pattern is found in the data taken at 20 MH. The data taken at approximately 3 millimeter electrode spacings, Figures 78 a. to 79 c., pages 258 to 263, does follow the general pattern described above, but upon increasing the spacing to about 6 millimeters, Figure 80, pages 264 and 266, the data appears to be in the form of a $\gamma_1\text{-line.}$

Note also that the electron velocity is substantially greater than that computed for the data at the lesser electrode spacing which conforms to the notion of a γ_1 -line.

On the basis of the observations discussed above, it is concluded that the simple theory presented in Chapter II is more applicable to the lower excitation frequencies and lesser primary electrode spacings. At frequencies up to about 9 MH_{z} , the computation of applicable frequencies of gyration does give a reasonable indication of the optimum values of magnetic intensity. However, at the higher excitation frequencies, 9 MH, and above, the gyration frequencies cannot be construed to give anything more than a coincidental indication of optimum magnetic intensity. As discussed in Chapter IV, it is believed that the neglection of any space charge is the most serious defect in the theory. Efforts have been made to correct this defeat, but the theoretical problem appears to be most difficult and experimental methods for probing the region between the primary electrodes, other than the rough secondary electrode measurements discussed in Chapter IV, are not feasible due to the restrictive dimensions of the cell geometry. It has been suggested that this discrepancy at the higher frequencies might be due to the excitation frequency becoming sufficiently close to the plasma frequency as to instigate another resonance phenomenon. However, assuming that the degree of ionization is of the order of 10^{-4} , a typical value for a slightly ionized gas, one estimates an electron plasma frequency in the vicinity of 9 GH_z and above for the pressures employed in this study. On the same basis, plasma frequencies for the positive ions H^+ , H_2^+ , and H_3^+ are estimated to range from 100 to a few hundred MH_. Since the degree of

ionization would have to be very much less than the above assumption in order to bring the plasma frequencies down to the vicinity of the excitation frequencies, it is considered unlikely that plasma frequency plays an important part in these measurements.

Despite the limitations in the theory it is believed to be sufficient, when taken in conjunction with the measured data, to be of much aid in the formative stages of conception of non-conventional energy sources. Such devices would naturally employ their own peculiar electrode and field structures which could deviate considerably from the geometry of the ionization cells employed in this study. For this reason, it is recommended with one possible exception that further studies of this nature of the ionization properties of hydrogen should follow rather than precede the conception of such an application. In this way, the more detailed theory and data could be closely tied to the application. The complexities of the ionization processes are such as to lead one to believe that this would generally be found essential to success. The exception to this recommendation is as follows; It might be found profitable to conduct an experimental study in the microwave region of excitation frequencies where one could feasibly create a "super-resonant" condition with the excitation, gyration, and plasma frequencies all being the same. The argument against this proposal is that microwave power is more difficult and extremely expensive to generate, consequently it is difficult to see any application of such a study to the field of energy conversion.

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APPENDIX

COMPUTER PROGRAMS

The velocity computations and all plotting of the Optimum Magnetic Intensity Mapping displayed in Chapter V was accomplished by means of digital computers. The raw data was first processed by the IBM 7040 to obtain listings of optimum magnetic intensity, pressure, applied voltage and electric intensity, charging current and the various charged particle velocities. This program is very much routine, based on Equations 2.13 through 2.25 of Chapter II. The principal problems experienced were in programming the magnet calibration and the correction of vacuum gauge readings to true pressure for hydrogen. Polynomial curve fitting techniques were employed to program the magnet calibration, using the library programs available. The results obtained in fitting the calibration curve for a 8.5 cm gap are shown in Figure 86. Referring to Figure 87, it is seen that the fifth power polynomial gives the closer fit to the measured calibration curve over the entire range of interest. Greater and lesser powers give greater deviations from the measured curve, so the fifth power was used in writing the computer program. However, the polynomial curve fitting techniques failed to suffice for the correction of vacuum gauge readings. Finally, it was discovered that the pressure correction nomogram supplied with the instrument plotted as essentially straight lines on a log-log graph. Then, writing the logarithmic









equations of the straight lines and programming these into the computer resulted in the degree of fit indicated in Figure 87. Undoubtedly, additional effort using the computer could increase the degree of fit of the magnet calibration and pressure correction curves over that shown in Figures 86 and 87, but the accuracy of the experimental data was not considered sufficient to justify this effort.

The complete computer program for processing the raw data is given on the five sheets of Table I. A sample readout of an output from this program is shown in Table II. The first block of entries in Table II, starting with the third card, is the input data to the program. This is in the following order: excitation frequency in MH_z , primary electrode spacing in meters, resistance of the rf load (current sampling resistance) in ohms, the control number \emptyset, \emptyset , the number of data points, and the number \emptyset . The following card lists the coefficients B_1, B_2, \ldots, B_6 for the 5th power polynomial fit of the magnet calibration. Its format is $6FI\emptyset.\emptyset$. The fifth card in this listing contains the powers and coefficients for the pressure correction. The format is $4FI\emptyset.\emptyset$ and the numbers read in are P \emptyset , Pl \emptyset , C \emptyset and Cl \emptyset , where for $P_{read} < 10$ mm Hg,

 $P_{corrected} = (1.0/C\phi)*(P_{read})**(1.0/P\phi).$

For $P_{read} > 10 \text{ mm Hg}$, the form of the equation remains the same, but the coefficients ClØ and PlØ are used in place of CØ and PØ. The remaining cards list the data taken during the operation of the ionization cell. The format is 6FlØ.Ø and the values read in are; magnet current in amperes, the pressure read on gauge tube no. 4 (see Figure 4) in mm Hg, the pressure on gauge tube no. 3 in mm Hg, mass

TABLE I

IBM 7040 PROGRAM FOR PROCESSING THE RAW DATA

	and the second				
BID	A-0001 C CHEN		7303-51000	·	
2008			1909-91000		
21870	B DKNAME NUDECK				
DIBEI	C DKNAME NODECK				
3	DO 630 N=1+2				
4	READ(5+220) F+D+R+G+M+K				
5	READ(5,221) 81,82,83,84,85,86				
6	READ(5,222) P0,P10,C0,C10				· · ·
	DIMENSION A (50,30)		.*		
7	K=K+1	· · ·			
10	READ(5,225) CURB,XX1,XX2,FLOW,CU	R•V			
15	B= B1*(CURB**5)+B2*(CURB**4)+B3*	(CURB**3)+B	4*(CURB**2)+85*C	JRB+B6
19	Y = 1.0/P0				
20	Z = 1.0/P10				
21	IF(XX1-10.0) 22.22.24				
22	XX1C= ((1.0/C0)*XX1)**Y				
23	GO TO 25				
24	XX1C= ((1.0/C10)*XX1)**Z				
25	IF(XX2-10.0) 26, 26, 28	1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 - 1999 -			
26	XX2C= ((1.0/C0)*XX2)**Y				
27	GO TO 30				
28	XX2C= ((1.0/C10)*XX2)**Z				
30	P = (XX1C + XX2C)/2 = 0				
35	F = (V/D) + 0.3536				
40	CURTX= (353.6*CUR)/R				
45	W=2.0*(3.1416*F)				
50	$W_{GE} = (17.6*(10.**4)*B)$				ł.
55	WGH1= 9.6*B				
611	WGH2 = WGH1/2 = 0	Sector Sector Sector			,
65	BWEB= B/10.0				
70					
75	V1E=VDE*(E/(2,8*(10,0**4)))*(1,0)	781			
9 0	$VETOT = SOPT((VDE \pm 2)) + (V)E \pm 2))$				
05	$E_{\rm NEE} + (1.5 + (VETOT + + 2) + 9.11/(10.0 + 1$	*31)			
00	AMT C= (/WCH1+++2)/(W++2))-1.0				
30	APTS = (WONT = 2) (WONT = 100				
100	$TE(AB=1/2.0 \pm B)/(W \pm 2)) = 125 \cdot $	105			
100	$\frac{11}{10} \frac{11}{10} 11$				
105	VINIA VINIA(.77/E)AR	•			
110	$V \cup \{1, \dots, V\} = \{1, 1, \dots, V\} = \{1, \dots, V\} = \{1,$				
110	$\nabla H_1 = \nabla H_$	// N##271			
120	- ENERT# #2*(VNI/0**2)*100702//10#	V**217			
125 .					
130	- LD= AD31() - tereo (D/D 0)#/1 0/0##011 365 36	5.140			
135	1F(CB-(B/2+0)*(1+0/W**2)) 100+10	J914U			
140	V1H2= (7.7/F)=(E/C)		· · · · · · · · · · · · · · · · · · ·		
145	VDH2= V1H2*(*////)*B				
150	VH2TO= SQRT((VDH2**2) + (V1H2**2	11			
155	LNEHZ= •5*(VH210**2)*2•3466/110•	U**Z71			
160	I=K				
165	A(I,1)=F				

	A(I+2)=B
	A(I,3)=E
	A(I,4)=P
	A(I,5)=FLOW
	A(I,6)=CURTX
170	A(I,7)=F
	A(I,8)=B
	A(I,9)=ENEE
	A([,10)=VETOT
	$A(I_{1}) = VIF$
	A(I, 12) = VDE
175	IF(AB-((2.0*B)/(W**2))) 180,180,190
180	$A(1 \cdot 13) = G$
	A(1.14)=F
	A(1.15)=B
	$A(I \cdot 16) = WGH1$
	$\Lambda(1,17) \pm 0$ MTS
	$\Delta(1 + 18) = 6$
195	GO TO 195
100	A(1.13)=E
	A(1,1,1,2)=B
	A(1.15)=ENEH]
	$\Delta(1, 16) = VHITO$
	A(1,17)=VIH1
	A(1,18)=VDH1
105	TELCE-LEVO 018/1 0/08#211 200.200.210
200	A/1.101+C
240	A(1)1777=0
	A(1)23)-C
205	CO TO 215
205	A/T-101-E
210	A(1)17)-P
, ' ·	A(1)22)=VH210
	A(1)/3)=V1H2
	A(1)/24) = VDHZ
215	TE(K•NF•M) GO TO 7
-255	I=U
256	
260	DO = 298 = J = 1 + 3
265	MY= D+ (6*J) MY= C- (6*J)
270	MTESE 6+(6*J) Telali Myesi en en en 70 700
275	IF LALL & MYEST & EU & GT GU TU 298
280	ANGLE= AIAN(A(I (MY)/A(I (MYES))
285	L=24+J
290	JF (A(1+MY)=U+U) 295+297
295	A(I)=ANGLE+3.1416

296	GO TO 298
297	$A(I_{+}L) = ANGLE$
298	CONTINUE
299	IF(I.NE.M) GO TO 256
400	WRITE(7+325)
405	WRITE(7•315)
410	WRITE(7,376) A(1,1)
	WRITE(7•380)
415	WRITE(7.335)
	L = 0
	I = 0
420	I = I + 1
	L = I + 1
425	WRITE(7,300) A(1,4),A(1,5),A(1,2),A(1,3),A(1,6)
	IF(A(I,4).GT.A(L,4)) GO TO 600
426	IF(L.NE.M) GO TO 420
	WRITE(7,300) A(L,4),A(L,5),A(L,2),A(L,3),A(L,6)
430	WRITE(7,310)
435	WRITE(7,315)
440	WRITE(7,375) A(1,1)
445	WRITE(7,316)
	WRITE(7,380)
450	WRITE(7,320)
	L≖0
	I = 0
455	I = I + 1
	L=I+1
460	WRITE(7,305) A(1,4),A(1,10),A(1,25),A(1,11),A(1,12),A(1,9)
÷ .	$IF(A(I_{9}4) \cdot GT \cdot A(L_{9}4)) GO TO 610$
461	IF(L.NE.M) GO TO 455
	WRITE(7,305) A(L,4),A(L,10),A(L,25),A(L,11),A(L,12),A(L,9)
465	WRITE(7,310)
470	WRITE(7,315)
475	WRITE(7,375) A(1,1)
480	WRITE(7•317)
	WRITE(7,380)
495	WRITE(7,320)
	L=0
500	κ=0
505	K=K+1
	L=K+1
510	IF (A(K,13), EQ+G) GO 10 525
515	WRITE(7,305) A(K,4),A(K,16),A(K,26),A(K,17),A(K,18),A(K,15)
	IF(A(K,4),GT,A(L,4)) GO 10 615
	IF(L.EQ.K) GO 10 535
516	IF(L.NE.M) GO TO 505
	60 10 510 Notices and and an are set of starts are set
525	WRITE(7934U) ALK94)9ALK914)9ALK91019ALK91019ALK91/)
	1+ (A(K+4)+G(+A(L+4)) 60 10 015

		IF(L.EQ.K) GO TO 535
		TELLANEAN) GO TO 505
	530	GO TO 505
	535	WRITE(7,310)
	540	WRITE(7,315)
	545	WRITE(7.375) A(1.1)
	660	WDITE(7,318)
	200.	
		WRITE(7,580)
	555	WRITE(7,320)
		, L=0
	560	K=0
	565	K=K+1
		I = K + 1
	570	TE(A(K-19)-E0-6) 60 TO 585
	576	WDITE (7, 205) A/K (3, A/K - 22) - A/K - 27) - A/K - 23) - A/K - 24) - A/K - 21)
	212	WRITE(1,300) = A(K)+1/3A(K)+2/3A(K)+
	· · ·	$F(A(K_{3}4), G(1, A(L_{3}4)), GO(10, B20))$
		IF(L+EQ+K) GO TO 630
	576	$IF(L \cdot NE \cdot M)$ GO TO 565
		K≖L
		GO TO 570
	585	WRITE(7,340) A(K,4),A(K,20),A(K,21),A(K,22),A(K,23)
	202	$IE(A(K \cdot 4) - GT \cdot A(1 \cdot 4)) GO TO 620$
		TELLERAN CO TO B46
		ITTLENE ON JOJ
		GO TO 570
	59 0	GO TO 565
	600	WRITE(7,381)
		WRITE(7,335)
		GO TO 426
	610	WRITE(7+381)
	010	
	615	WRI(E(7)381)
		WRITE(7,320)
·		GO TO 516
	620	WRITE(7,381)
		WRITE(7,320)
		GO TO 576
	620	CONTINUE
	0.50	
	340	FORMAT (5H POLE) 5F10.592E17.77
	220	FORMAT (4F10+0+2110)
	221	FORMAT (6F10+0)
	222	FORMAT (4F10.0)
	225	FORMAT (6F10.0)
	300	FORMAT(4X F10.2,F12.2,F12.3,F15.2,F10.2)
	305	FORMAT (F6.2.F12.1.F11.3.F12.1.F14.1.E15.5)
	1.00	

310	FORMAT (18X 37H VELOCITY AND ENERGY COMPUTATIONS FOR/)
315	FORMAT (31X 9H HYDROGEN /)
316	FORMAT(23X 26H CALCULATIONS FOR ELECTRON ///)
317	FORMAT(23X 28H CALCULATIONS FOR ATOMIC ION ///)
318	FORMAT(21X 31H CALCULATIONS FOR MOLECULAR ION ///)
320	FORMAT (2X 2H P.6X 8H V TOTAL, 5X 6H ANGLE, 5X 8H V PERP.,
	25X 8H V DRIFT+6X 7H ENERGY)
325	FORMAT (25X 22H EXPERIMENTAL DATA FOR /)
335	FORMAT(7X 9H PRESSURE:2X 10H MASS FLOW:5X 2H B:13X 2H E:
	26X BH CURRENT)
375	FORMAT(28X F6+3+11H MEGACYCLES /)
376	FORMAT(28X F6.3,11H MEGACYCLES ///)
380	FORMAT(23X 17H GAMMA-ONE VALUES /)
381	FORMAT(/ 23X 17H GAMMA-TWO VALUES /)
	END
\$ENT	RY
6.100	

\$IBSYS

TABLE II

SAMPLE READOUT OF OUTPUT FROM RAW DATA PROCESSING PROGRAM

	1 10	4				
	DATA TAKEN JU	LY 9 🔸	1966 AL CELL	R ¤52.0 Ø	HMS D=.002794 M	IETERS
6.0	.002794	52.0	0.0		14 0	
•00	.0000034	000	04922.00134888	•115543	10 .06476900	
1.25	8.35	1.48	.0000273			
6.5	2.4	1.55	10.0	6.0	730.0	
55	4.1	2.5	20.0	5.8	710.0	
5.5	6.1	3.5	30.0	5.7	700.0	
5.5	8 • 5	4.6	40.0	5.6	690.0	
5.5	9.0	5.0	50.0	5.6	690.0	
5.7	20.0	7.0	70.0	5.6	690.0	
5.7	33.0	10.0	100.0	5.8	710.0	
5.8	57.0	17.0	150.0	5.9	710.0	
6.2	83.0	24.0	200.0	6.0	720.0	
6.2	150.0	43.0	300.0	6.1	740.0	
22.5	2 • 4	1.55	10.0	4•2	530•0	
30.0	4.1	2.5	20.0	4.8	600.0	
40.0	6.1	3.5	30.0	5.2	640.0	
45.0	8.5	4.5	40.0	5.5	680.0	

EXPERIMENTAL DATA FOR HYDROGEN 6.000 MEGACYCLES

GAMMA-ONE VALUES

PRESSURE	MASS FLOW	.8	E	CURRENT
1+25	10.00	0.860	92386.54	40.80
1.89	20.00	0.733	89855.40	39.44
2.55	30.00	0.733	88589.83	38.76
3.26	40.00	0.733	87324.27	38.08
3.44	50.00	0.733	87324.27	38.08
4.25	70.00	0.758	87324.27	38.08
4.98	100.00	0.758	89855.40	39.44
5.33	150.00	0.771	89855.40	40.12
5.56	200.00	0.822	91120.97	40.80
5.97	300.00	∂ •822	93652.11	41.48

GAMMA-TWO VALUES

PRESSURE	MASS FLOW	В	É	CURRENT
1.25	10.00	2.874	67075.16	28,56
1.89	20.00	3.692	75934.14	32.64
2.55	30.00	4.565	80996.42	35.36
3.24	40.00	4.905	86058.70	37.40

VELOCITY AND ENERGY COMPUTATIONS FOR Hydrogen 6.000 megacycles

CALCULATIONS FOR ELECTRON

GAMMA-ONE VALUES

P	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	1074413.1	0.000	267.7	1074413.0	0.52581E-18
1.89	1225554+2	0.000	358.2	1225554.2	0.68415E-18
2.55	1208292.9	0.000	353.1	1208292.8	0.66502E-18
3.26	1191031.6	0.000	348.1	1191031.5	0.64615E-18
3.44	1191031.6	0.000	348.1	1191031.5	0.64615E-18
4.25	1151376.8	0.000	325.3	1151376.7	0.60384E-18
4.98	1184750.0	0.000	334.7	1184750.0	0.63935E-18
5.33	1165323.9	0.000	323.8	1165323.9	0.61856E-18
5.56	1108852.7	0.000	289.1	1108852.7	0.56006E-18
5.97	1139654.2	0.000	297.2	1139654.1	0.59161E-18
		GAMMA-TW	VALUES		·
P	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	233398.2	0.000	17.4	233398.2	0.24813E-19
1.89	205699.2	0.000	11.9	205699.2	0.19273E-19

VELOCITY AND ENERGY COMPUTATIONS FOR HYDROGEN 6.000 MEGACYCLES

8.3

7.7

177428.4

175460.6

0.14340E-19

0.14023E-19

CALCULATIONS FOR ATOMIC ION

GAMMA-ONE VALUES

0.000

0.000

2.55

3.24

177428.4

175460.6

Ρ	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	248952.0	4.602	-247449.9	-27306.4	0.51853E-16
1.89	238455.4	4.619	-237406.8	-22338.0	0.47573E-16
2.55	235096.8	4.619	-234063.0	-22023.4	0.46242E-16
3.26	231738.3	4.619	-230719.3	-21708.8	0.44930E-16
3.44	231738.3	4.619	-230719.3	-21708.8	0.44930E-16
4.25	232397.7	4.615	-231304.6	-22513.4	0.45186E-16
4.98	239133.8	4.615	-238009.1	-23166.0	0.47844E-16
5.33	239483.5	4.614	-238319.5	-23582.9	0.47984E-16
5.56	244346.8	4.607	-242999.2	-25626.5	0.49952E-16
5.97	251134.2	4.607	-249749.2	-26338.3	0.52766E-16
e An Star		GAMMA-	TWO VALUES		
P	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	392521.9	4.359	-368273.6	-135823.3	0,12891E-15
1.89	1841752.0	4.270	-1664422.3	-788510.6	0.28380E-14
2.55	681325.1	1.041	587870.8	344400.8	0.38838E-15
3.24	463089.5	1.009	391915.0	246687.1	0+17942E-15

VELOCITY AND ENERGY COMPUTATIONS FOR Hydrogen 6.000 Megacycles

CALCULATIONS FOR MOLECULAR ION

GAMMA-ONE VALUES

P	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	120729.6	4.602	-120001.1	-13242.3	0.17102E-16
1.89	116842.0	4.619	-116328.2	-10945.5	0.16018E-16
2.55	115196.3	4.619	-114689.7	-10791.3	0.15570E-16
3.26	113550.7	4.619	-113051.3	-10637.2	0.15128E-16
3.44	113550.7	4.619	-113051.3	-10637.2	0.15128E-16
4.25	113655.6	4.615	-113121.0	-11010.3	0.15156E-16
4.98	116949.9	4.615	-116399.9	-11329.5	0.16048E-16
5.33	117005.4	4.614	-116436.7	-11522.0	0.16063E-16
5.56	118888.6	4.607	-118232.9	-12468.7	0.16584E-16
5.97	122191.0	4.607	-121517.2	-12815.1	0.17518E-16
		GAMMA-	TWO VALUES		
P	V TOTAL	ANGLE	V PERP.	V DRIFT	ENERGY
1.25	105930.5	4.359	-99386.6	-36654.9	0.13166E-16
1.89	138407.6	4+270	-125081.3	-59256.5	0.22477E-16
2.55	181932.9	4.182	-156977.9	-91964.6	0.38836E-16
3.24	213927.8	4.151	-181048.1	-113959.0	0.53696E-16

flow in std. cc/min., peak to peak value of the waveform across the current sampling resistance in volts, and the peak to peak value of the applied voltage in volts. The separation into Y-line values is accomplished by the arrangement of these data cards. They should begin with the low magnet current and low mass flow readings and progress through the highest mass flow readings before introducing the upper magnet current values. The program employs a DO loop to allow the running of any number of data sets at one time. This is controlled by appropriately indexing in statement no. 3 of the program.

The output from the raw data processing program is divided into four parts, as shown in Table II. The first part groups the processed experimental data in accordance with the y-lines. Cell pressure is listed in mm Hg, mass flow in std. cc/min., optimum magnetic intensity B in kilogauss, applied electric intensity E in volts per meter and charging current in milliamperes. The optimum magnetic intensity and applied electric intensity values are appropriately substituted into equations 13 through 25 of Chapter II to order to compute the charged particle velocities shown in the remaining parts of Table II. These following three parts group the velocity computations for the electron, the atomic ion, and the molecular ion respectively in accordance with the γ -lines. The velocities are listed in meters per second and minus signs are used to indicate that the direction of a particular velocity component is opposite to that of the same component for an electron. The angle of total velocity is measured in radians with respect to the electron drift velocity component, $V_{D_{o}}$, and energies computed on the basis of $\frac{1}{2}mv^2$ are listed in Newton-The discovery, discussed in Chapter IV, that the applied meters.

voltage tends to a constant independent of the electrode spacing largely destroys any significance that this later computation may have had otherwise and consequently, it was ignored in plotting the data.

All of the graphs shown in Chapter V were drawn by a combination of the IBM 1620 digital computer with disc storage and the Calcomp 565 automatic plotter with liquid ink conversion. The liquid ink conversion proved to be essential in order to obtain satisfactory photographic reproduction. The program for making the data plots is given in the eight sheets of Table III and consists of five linked sub-programs which were stored on the disc. The total output from the IBM 7040 raw data processing program, as shown in Table II, for example, constitutes the input data to this program. The essential steps written into the program of Table III are as follows: (1) read in the data, (2) plot cell pressure versus optimum magnetic intensity and then compute least squares fits to these points so as to form the γ -lines, (3) scale off the γ -lines to form the abscissa of the following graphs, (4) plot applied voltage, electric intensity, and charging current as a function of the appropriate γ -variable, (5) scale the computed velocity components logarithmically and plot as a function of the appropriate Y-variable.

TABLE III

IBM 1620 PROGRAM FOR LEAST-SQUARES FITTING THE

Y-LINES AND DATA PLOTTING

```
3400032007013600032007024902402511963611300102
ZZJOB
ZZFOR
#LDISKPROG1
    ODIMENSION IND(11)+A(15)+B(15)+C(15)+D(15)+E(15)+GXX(2)+HXX(2)+
     1DISTO(15), DISTT(15)
      COMMON N1,N2,IND,GXX,HXX,DISTO,DISTT
      READ 101.INDEX.N1.N2
      READ 102, DUMMY, DUMYY, DUYYY, DYYYY, YYYYY, DDDDD, DUDUD, UUUUU
      PRINT 102, DUMMY, DUMYY, DUYYY, DYYYY, YYYYY, DDDDD, DUDUD, UUUUU
      DEFINE DISK (10,70)
      FIND (INDEX)
      DO 2 I=1.3
    2 READ 102. DUMMY
      IND(1) = INDEX
      RECORD (INDEX) N1,N2
      IND(2) = INDEX
   READ AND RECORD VOLTAGES
С
      DO 10 I=1.N1
   10 READ 106, A(I)
      RECORD (INDEX) (A(I)+I= 1+N1)
      IF (N2) 16,16,12
   12 DO 14 I= 1.N2
   14 READ 106, A(I)
      IND(3) = INDEX
      RECORD (INDEX) (A(I), I = 1, N2)
   16 \text{ IND}(4) = \text{ INDEX}
      DO 18 I=1.11
   18 READ 102, DUMMY
   READ AND RECORD PRESSURE MASS FLOW B E CURRENT
C
      DO 20 I= 1, N1
   20 READ 109, A(I), B(I), C(I), D(I), E(I)
      RECORD (INDEX) (A(I),B(I),C(I),D(I),E(I),I=1,N1)
      IF (N2) 30,30,21
   21 DO 22 I= 1,4
      DO 22 I= 1,4
   22 READ 102. DUMMY
   24 \text{ IND}(5) = \text{ INDEX}
      DO 26 T=1+N2
   26 READ 109, A(I), B(I), C(I), D(I), E(I)
      RECORD (INDEX) (A(I),B(I),C(I),D(I),E(I),I=1,N2)
   30 DO 50 M=1.3
      MM= 2*M+4
      IND(MM) = INDEX
      DO 31 I=1+13
   31 READ 102, DUMMY
      DO 32 I=1.N1
   32 READ 112, A(I), B(I), C(I), D(I), E(I).
      RECORD (INDEX) (A(I),B(L),C(I),D(I),E(I),I=1,N1)
      IF (N2) 50,50,33
```

III (Continued) 33 DO 34 I= 1.4 DO 34 I=1+4 34 READ 102. DUMMY 36 MM= 2*M+5 IND(MM) = INDEX DO 38 I= 1.N2 38 READ 112+ A(I)+B(I)+C(I)+D(I)+E(I) RECORD (INDEX) (A(I)+B(I)+C(I)+D(I)+E(I)+I=1+N2) 50 CONTINUE 101 FORMAT(5X,15,3X,12,3X,12) 102 FORMAT(20A4) 106 FORMAT(25X,25X,F10.4) 109 FORMAT (4X,F10.2,F12.2,F12.3,F15.2,F10.2) 112 FORMAT (6X,F12.1,F11.3,F12.1,F14.1,E15.5) CALL LINK(PROG2)

END

```
ZZFOR
*LDISKPROG2
     ODIMENSION IND(11) + A(15) + B(15) + C(15) + D(15) + E(15) + GXX(2) + HXX(2) +
     1DISTO(15) + DISTT(15)
      COMMON N1,N2,IND,GXX,HXX,DISTO,DISTT
      DEFINE DISK (10,70)
      FIND(IND(1))
      CALL PLOT (201,0.0.7.0.7.0.7.0.0.0.0.7.0.7.0.7.0)
      M= 1
      J= N1
   20 PS = 0.0
      BS= 0.0
      PQ= 0.0
      PB= 0.0
      IF(M-1) 50,50,60
   50 INDE= IND(4)
      GO TO 70
   60 \text{ INDE= IND(5)}
   70 FETCH(INDE)(A(I),B(I),C(I),D(I),E(I),I=1,J)
      DO 22 I=1+J
      CALL PLOT (0,C(1),A(1))
      PUNCH TAPE 82
   82 FORMAT(23H96601111333355557777922)
      PS = PS + A(I)
      BS= BS+C(I)
      PQ=PQ+A(I)**2.
   22 PB= PB+A(I)*C(I)
      AN=J
      DE= AN*PO-PS**2.
      H= (BS*PQ-PS*PB)/DE
      G= (AN*PB-BS*PS)/DE
      BEND= 7.0+G+H
      PEND = 7.0
      IF (BEND-7.0) 31.31.30
   30 \text{ BEND} = 7.0
      PEND= (7.0-H)/G
   31 CALL PLOT (0,H,0.0)
      CALL PLOT (9,BEND,PEND)
      GXX(M) = G
      HXX(M) = H
      GO TO (40,42),M
   40 IF(N2-2) 42,41,41
   41 M = 2
      J= N2
      GO TO 20
   42 CALL PLOT (99)
      CALL PLOT (9,0,0,0,0)
      CALL LINK(PROG3)
      END
```

```
ZZFOR
#LDISKPROG3
     ODIMENSION IND(11) +A(15) +B(15) +C(15) +D(15) +E(15) +GXX(2) +HXX(2) +
     1DISTO(15),DISTT(15)
      COMMON N1,N2,IND,GXX,HXX,DISTO,DISTT
      DEFINE DISK (10,70)
     M=1
      J= N1
   12 G=GXX(M)
     H= HXX(M)
      ABC= 1.0+ G**2.
     RV= 1.0/SQRTF(ABC)
      RH= G*RV
      CALL PLOT (201,0.0,7.0,7.0,7.0,0.0,7.0,7.0,7.0)
      IF (M-1) 50+50+60
   50 INDE= IND(4)
      GO TO 70
   60 INDE= IND(5)
   70 FETCH(INDE)(A(I),B(I),C(I),D(I),E(I),I=1,J)
      DO 20 I= 1.J
      BB = (G*(G*H-A(I))+C(I))/ABC
      PP=(G*(G*A(I)+H-C(I)))/ABC
      DISTN= SQRTF((C(I)-BB)**2.+(A(I)-PP)**2.)
      PPROJ= DISTN*RV
      BPROJ= DISTN*RH+H
      CALL PLOT (90.BPROJ.PPROJ)
      PUNCH TAPE 82
   82 FORMAT (10H0337777339)
      IF(M-1) 51,51,61
   51 DISTO(I) = DISTN
      GO TO 20
   61 DISTT(I)= DISTN
   20 CONTINUE
      GO TO (21+23)+M
   21 IF (N2-2) 23+22+22
   22 M = 2
      J = N2
      GO TO 12
   23 CALL PLOT (7)
      PRINT 102
  102 FORMAT(20HNEW PAPER THEN START)
      PAUSE
      CALL LINK(PROG4)
      END
```

```
ZZFOR
*LDISKPROG4
     ODIMENSION IND(11)+A(15)+B(15)+C(15)+D(15)+E(15)+GXX(2)+HXX(2)+
     1DISTO(15) + DISTT(15)
      COMMON N1+N2+IND+GXX+HXX+DISTO+DISTT
      DEFINE DISK (10,70)
      CALL PLOT (201,0.0,12.0,12.0,12.0,0.0,7.0,7.0,7.0)
   READ IN DATA OF E AND CURRENT FOR GAMA ONE OR TWO
C
      M=1
      J=N1
      INDE=IND(4)
   70 FETCH (INDE)(A(I),B(I),C(I),D(I),E(I),I=1,J)
С
   SCALE THE DATA AND PLOT
      DO 1 I=1,J
      IF (D(I)-80000.0) 1.1.2
    1 CONTINUE
      GO TO 4
    2 DO 3 I=1+J
    3 D(I) = D(I) - 40000 \cdot 0
      PRINT 103
    4 DO 5 I = 1 J
     IF (E(I)-80.0) 5,5,6
    5 CONTINUE
      GO TO 8
    6 DO 7 I= 1.J
    7 E(I) = E(I) - 40 \cdot 0
      PRINT 104
    8 IF (M-1) 51,51,61
   51 DO 10 I=1,J
   10 A(I)=DISTO(I)
      GO TO 71
   61 DO 12 I=1.J
   12 A(I) = DISTT(I)
   71 DO 14 I=1,J
      D(I) = D(I) / 10000
      CALL PLOT (0,D(I),A(I))
      PUNCH TAPE 84
   84 FORMAT(44H9666601111111333333355555555777777777222220)
   14 CONTINUE
      CALL PLOT (99)
      DO 16 I=1.J
      E(I) = E(I) / 10_{\bullet}
      CALL PLOT (0,E(I),A(I))
      PUNCH TAPE 85
   85 FORMAT (36H977701111143434434676676761111193330)
   16 CONTINUE
      CALL PLOT (99)
   READ IN DATA OF VOLTAGE FOR GAMA ONE OR TWO
C
      IF (M-1) 52+52+62
   52 INDE = IND(2)
      GO TO 72
```

```
62 \text{ INDE} = \text{IND}(3)
   72 FETCH (INDE) (A(I) + I=1+J)
C SCALE THE DATA AND PLOT
      DO 91 I= 1.J
     IF (A(I)-800.0) 91,91,92
   91 CONTINUE
      GO TO 9
   92 DO 93 I= 1,J
   93 A(I) =A(I)-400.0
      PRINT 105
  103 FORMAT (12HE SHIFTED 40 )
  104 FORMAT (18HCURRENT SHIFTED °40)
  105 FORMAT(18HVOLTAGE SHIFTED 40)
   9 IF (M-1) 53,53,63
   53 DO 20 I=1,J
   20 B(I) = DISTO(I)
      GO TO 73
   63 DO 22 I= 1,J
   22 B(I) = DISTT(I)
   73 DO 24 I=1,J
   A(I) = A(I)/100.
      CALL PLOT (0,A(I),B(I))
      PUNCH TAPE 87
   87 FORMAT (42H977777012123233434545565676778781819333330)
   24 CONTINUE
      CALL PLOT (7)
      PRINT 102
  102 FORMAT(20HNEW PAPER THEN START)
      PAUSE
      IF (N2-2) 36,32,32
   32 IF(M-1) 34,34,36
   34 M = 2
      J= N2 -
      CALL PLOT (201,0.0,12.0,12.0,12.0,0.0,7.0,7.0,7.0)
      INDE = IND(5)
      GO TO 70.
   36 CALL LINK(PROG5)
```

END

ZZ	FOR		
*1	DIS	KPROG5	
•		DIMENSION IND(11) A(15) B(1)	5) •C(15) •D(15) •F(15) •GXX(2) •HXX(2) •
		1DISTO(15) $DISTT(15)$	
		DEEINE DISK (10.70)	
		COMMON NI NI TUTTUT	TODICTT
		M- 1	10401311
	9	CALL PLOT (20190-0912-298-2)) + 1 2 • 2 • U • U • (• U • 4 • 362 • / • U)
	• • •	PRINT 200	
	200	FORMATIONBLUE PEN)	
	10	INDE= IND(MM)	
		PEICH (INDE) (A(I))D(I))C(I)	(9D(1)9C(1)91=19J)
		PAUSE	
		CALL PLOT (99)	
		DO = 20 I = I + J	
		A(I) = ABSF(A(I))	
		$A(I) = A(I)/IU \bullet U$	
		IF (A(1)-1.0) 20,60,60	
	60	A(I) = LOGF(A(I))	
	·	CALL PLOT (UAAII) DISID(I)	
		PUNCH TAPE 150	
	20	CONTINUE	
		CALL PLOT (99)	
		$DO 22 I = 1 \cdot J$	
		C(I) = ABSF(C(I))	
		$C(I) = C(I)/10 \cdot 0$	
		IF(C(I)-1.0) 22,62,62	
	62	C(I) = LOGF(C(I))	
		CALL PLOT (0,C(I),DISTO(I))	
		PUNCH TAPE 151	
	22	CONTINUE	
		CALL PLOT (99)	
		DO 24 $I = 1 + J$	
		D(I) = ABSF(D(I))	
		D(I) = D(I)/10.0	
		IF (D(1)-1.0) 24,64,64	
	64	D(I) = LOGF(D(I))	
		CALL PLOT (0,D(1),DIS10(1))	
		PUNCH TAPE 152	
	24	CONTINUE	
		CALL PLOT (99)	
		GO TO (41,42,43,44,45),M	
	41	PRINT 201	
	201	FORMAT(9HBLACK PEN)	
		M= 2	
		MM= 8	
		GO 10 10	
	42	PRINT 202	
	202	FORMAT (7HRED PEN)	

.

```
M= 3
     MM= 10
     GO TO 10
  43 IF(N2-2) 8888,50,50
  50 IF (MM-11) 55+8888+8888
  55 M= 4
     MM= 7
     CALL PLOT (7)
     PRINT 102
 102 FORMAT(20HNEW PAPER THEN START)
     PAUSE
     J = N2
     DO 30 I= 1.J
  30 \text{ DISTO(I)} = \text{ DISTT(I)}
     GO TO 9
  44 M = 5
     MM= 9
     PRINT 201
     GO TO 10
  45 MM= 11
     PRINT 202
     M= 3
     GO TO 10
 150 FORMAT(34H97777702222244444666666888889333330)
 151 FORMAT(34H9667081121323343545565767787193220)
152 FORMAT(44H96666011111111333333335555555577777777922220)
8888 CALL PLOT (7)
     CALL EXIT
     END
```

ZZZZ
VITA

James Russell McDougal

Candidate for the Degree of

Doctor of Philosophy

Thesis: A STUDY OF THE IONIZATION PROPERTIES OF HYDROGEN IN THE PRESENCE OF RELATIVELY STRONG MAGNETIC FIELDS

Major Field: Electrical Engineering

Biographical:

Personal Data: Born near Osawatomie, Kansas, March 21, 1929, the son of Thomas J. and Neva L. McDougal.

- Education: Graduated from Lane High School, Lane, Kansas, in 1947; received the Bachelor of Science Degree in Electrical Engineering from the University of Kansas in February, 1952; received the Master of Science Degree (Kogaku Shushi) in Electrical Communications Engineering from the Tohoku University, Sendai, Japan, in March, 1955; completed requirements for the Doctor of Philosophy Degree in May, 1968.
- Professional Experience: From March, 1952, to April, 1955, was a Lieutenant in the United States Army Corps of Engineers with assignments as an Electrical Engineering Officer except for the period November, 1952, to July, 1953, when in Korea. Industrial experience includes the period June 10, 1957, to September 2, 1961, at the Boeing Company, Wichita, Kansas, as a Research Engineer in the Antenna and Microwave Group. Was employed as an Assistant Professor at the University of Wichita, Kansas, from September, 1961, to September, 1964; also, was retained as a part-time consultant by the Boeing Company during this period. Employed by the School of Electrical Engineering of Oklahoma State University as an Instructor from September, 1964, to November, 1967; and as an Assistant Professor since November, 1967.
- Professional Organizations: Member of the Institute of Electrical and Electronics Engineers and Antennas and Propagation Professional Group, The Institute of Electrical Communications Engineers of Japan, Eta Eta Kappa, Sigma Tau, and Scabbard and Blade.