# DETERMINATION OF RESILIENT MODULUS FOR THE ESTIMATION OF LAYER COEFFICIENTS OF ASPHALT CONCRETE MIXES

By

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## CHAPTER I

## INTRODUCTION

The AASHTO Guide for the Design of Pavement Structures [1-3] requires the estimation of resilient modulus for flexible pavement design. Resilient modulus is considered a fundamental material property and is determined from a repeated load tests and is based on the resilient (recoverable) portion of the strain. The resilient modulus is the ratio of the repeated stress to the corresponding recoverable (resilient) strain during loading i.e., it is the elastic stiffness of a material after a predetermined number of load repetitions have been applied.

The resilient modulus test is designed to simulate the behavior of bituminous materials under in service conditions found in a pavement system. The compaction methodology used in the preparation of specimens should closely correspond to field compaction techniques. Three compaction procedures were evaluated in this study including: gyratory shear, Marshall hammer, and dynamic compaction apparatus. Aggregates were obtained from five different sources located in different parts of Oklahoma. The resilient modulus test was conducted by applying a haversine compressive load on a sample at three temperatures ( 41° Fahrenheit ( 5° Centigrade), 77° Fahrenheit ( 25° Centigrade ), 104° Fahrenheit ( 40° Centigrade ) along two diametral axes ( second axis is oriented 45° to the first ), and at three rest periods ( 0.9s, 1.9s, 2.9s ) using a predetermined stress that differs for each temperature. The stresses to be applied on the sample at three temperatures is obtained by conducting the indirect tensile strength test on a sample. 30.15, and 5 percent of the stresses obtained at failure are the stresses applied at 41° Fahrenheit ( 5° Centigrade ), 104° Fahrenheit ( 40°

Centigrade ) respectively.

Based on the test results obtained during the course of the study, it was concluded that the samples prepared using the gyratory shear compactor and the dynamic compaction apparatus exhibit similar characteristics and the Marshall hammer tends to exhibit a poor behavior. The resilient modulus values of the gyratory shear and dynamic compaction apparatus compacted samples lie close to each other at the three testing temperatures. Overall it was determined that the resilient modulus increases with decreasing temperatures and the effect of three rest periods and two axes is not significant.

## Objective

The primary objective of this study was to determine a representative resilient modulus value for several Oklahoma Department of Transportation "Type B" bituminous mixes. A secondary objective was to establish a reproducible and realistic compaction methodology for molding laboratory specimens.

#### Scope of Work

The compaction procedures used in this study were chosen to determine the best laboratory compaction procedure for simulating field compaction. The devices evaluated were : The gyratory shear compactor, Marshall hammer, and dynamic compaction apparatus (Figure 2). Samples were prepared using these techniques and their engineering properties determined.

The variability in sample preparation was assessed using the following tests : resilient modulus tests, Hveem stability, indirect tensile strength test, and Air voids determination.

Resilient modulus tests were performed at 41° Fahrenheit ( 5° Centigrade), 77° Fahrenheit ( 25° Centigrade ), 104° Fahrenheit ( 40° Centigrade ) to obtain the effect of varying temperature. The samples are tested at different load intensities [( 30.15, and 5 percent of the stresses obtained at failure on the indirect tensile strength test conducted on a sample are the stresses applied at 41° Fahrenheit ( 5° Centigrade), 77° Fahrenheit ( 25° Centigrade ), 104° Fahrenheit ( 40° Centigrade ) for different temperatures respectively], at three rest periods ( 0.9s, 1.9s, 2.9s ), and the sample is tested along two diametral axes ( second axis is oriented 45° to the first )

A typical mix gradation is shown in Table 1. Samples were prepared using aggregates obtained from five different sources within Oklahoma.

## CHAPTER II

## LITERATURE REVIEW

## Background

The implementation of the AASHTO Pavement Design procedure for flexible pavements requires the estimation of "layer coefficients" for bituminous mixes. AASHTO layer coefficients for asphalt concrete and granular materials are defined in terms of resilient or dynamic modulus value.

The AASHTO equation for the design of flexible pavements is as follows :

$$log_{10}(W_{18}) = Z_{R}*S_{o} + 9.36 \text{ Log}_{10}(\text{ SN}+1) - 0.20 + 2.32 \text{ Log}_{10}(M_{R}) + Log_{10} [(\Delta PSI/(4.2-1.5))/(0.4 + \{1094/(SN+1)^{5.19}\})] - 8.07$$

where

 $M_R$  = Resilient modulus of subgrade soil (psi)

S<sub>o</sub> = Overall standard deviation

SN = Structural number

Where the structural number is expressed as

SN =  $a_1 D_1 + a_2 D_2 m_2 + a_3 D_3 m_3$ 

where:

 $a_i = i^{th}$  layer coefficient

- $D_i = i^{th}$  layer thickness (inches)
- $m_i = i^{th}$  layer drainage coefficient
- ΔPSI = Design serviceability loss (Initial serviceability index minus terminal serviceability index)

The structural number is a abstract number expressing the structural strength of the pavement required for a given combination of soil support, total traffic expressed as a 18-kip single axle loads, terminal serviceability and environment. The required structural number must then be converted to the actual thickness of the surface, base, and subbase, using the appropriate layer coefficients representing the relative strengths of these materials. The layer coefficients are based on elastic moduli  $M_R$  and are to be determined based on stress and strain measurements in a multilayered pavement system. The layer coefficient expresses the empirical relationship between SN and layer thickness and is a measure of the relative ability of the material to function as a structural component of the pavement system. The layer coefficient  $a_i$  is related directly to the resilient modulus as follows :

 $a_i = A M_R^B$  (AASHTO Guide 1986)

where

A, B are experimentally derived regression constants

 $M_R$  is the resilient modulus of Asphalt Concrete

An unknown layer coefficient  $a_i$  can also be estimated from a known coefficient  $a_{ref}$  using the following relationship:

$$\mathbf{a}_{i} = \mathbf{a}_{ref} \left[ \mathbf{M}_{Ri} / \mathbf{M}_{RRef} \right]^{B} \qquad [4]$$

where:

 $a_i = i^{th}$  layer coefficient

a<sub>ref</sub> = layer coefficient for the reference material

 $M_{Ri}$  = resilient modulus for the material in the i<sup>th</sup> layer

 $M_{Rref}$  = resilient modulus for the reference material

A, B are experimentally derived regression constants

#### Need For Resilient Modulus in Pavement Design

A study of resilient modulus as used in the AASHTO pavement design procedure was conducted by Elliot and Thorton [5]. The effect of variations in subgrade resilient modulus on the various design parameters and on the AASHTO design thickness were examined. They concluded that resilient modulus is a fundamental material property relating to pavement design and performance. Resilient modulus provide a measure of the load induced stress - strain behavior and governs the load response of the pavement system.

## **Evaluation of Compaction Devices**

A study conducted as part of the Asphalt Aggregate Mixture Analysis System (AAMAS) [6] was to ensure that laboratory molded specimens will be fabricated in a manner that will adequately simulate field conditions and yield reliable engineering properties. Five compaction devices were selected as a part of this study including : The Mobile steel wheel simulator, Texas gyratory compactor, Marshall impact hammer, California kneading compactor, and the Arizona vibratory kneading compactor. The compaction devices were selected on the basis of their availability, uniqueness in mechanical manipulation, and potential for use by agencies responsible for asphalt mixture design.

The ability of these devices to simulate field compaction was based on the similarity between engineering properties such as resilient moduli, indirect tensile strength, strains at failure, and tensile creep data. Project locations were in Texas, Virginia, Michigan, Wyoming, and Colorado. The compaction procedures used at each of the locations was the standard method used by that State Department of Transportation. Indirect tensile and resilient modulus tests were performed at 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), 104° Fahrenheit ( 40° Centigrade ), and the creep compliance tests were performed at 77° Fahrenheit ( 25° Centigrade ). The resilient modulus tests were performed in accordance with ASTM D 4123-82 and the indirect tensile tests were performed. Ten percent of the stress to failure as measured in the indirect tensile strength test was applied to the specimens during the resilient modulus procedure to produce deformations in the elastic range without damaging the sample.

The study concluded that of the five devices evaluated, the Texas gyratory shear compactor demonstrated the ability to produce mixtures with engineering properties nearest those determined from field cores. Because of its operational simplicity and the potential to produce 4 inch (10.16 cm) and 6 inch (15.24 cm) diameter specimens the Texas gyratory was selected as the most applicable device for preparation of specimens used in mix design analysis.

A study of the AASHTO flexible pavement design equation by Baus and Fogg [7] determined the relative importance of the input parameters. This study assessed the relative changes in the required thickness of the pavement structure that would result from errors in input parameters. The design equation for structural number (SN) uses a converging iterative procedure as a basis for the study. The input parameters were chosen to represent a wide range of design values for flexible pavement.

The following parameters were evaluated to assess the change in structural number : 18 kip equivalent single axle load repetitions ( $W_{18}$ ), resilient modulus ( $M_R$ ), reliability (R), and standard Error ( $S_0$ ) was assessed. It was concluded that the variation in resilient modulus value has the most pronounced effect on SN.

Mamlouk and Sarofim [8] conducted research on the numerous moduli values typically used to characterize asphalt mixtures. The moduli evaluated include: Young's, shear's, bulk, complex, dynamic, resilient, and shell nomographic moduli.

An elastic material is defined as the material in which strains completely appear and disappear immediately on the application and removal of stresses. The effects of temperature are neglected in the theory of linear elasticity and a material can be fully characterized by Young's modulus and Poisson's ratio. Young's modulus is the

slope of the straight line representing the stress-strain relation, and Poisson's ratio is defined as the absolute value of the lateral strain divided by the axial strain when an axial stress is applied on an specimen. The other moduli i.e., shear, bulk, can be expressed in terms of Young's modulus. The stress-strain relation of an asphalt concrete specimen is non linear. These moduli are applicable to static loading conditions as opposed to dynamic (repetitive) load conditions. The dynamic modulus was found to be insufficient to explain material response because it ignores the loading frequency and the phase lag between deformation and load.

Viscoelastic materials exhibit a combination of elastic and viscous (time dependent) responses, and are highly temperature dependent. The stress - strain relation depends on the rate of load application and is largely temperature dependent. The responses of asphaltic mixes are time and temperature dependent and they should be analyzed. Repetitive load - type laboratory tests have been developed in an attempt to simulate traffic loads (diametral resilient modulus test, triaxial resilient modulus test, and the sinusoidal unconfined compression tests ). The stress - strain relation of asphalt concrete is essentially linear after several load applications. The resilient modulus is the slope of the stress - strain curve after the application of load repetitions and is the current modulus of the material, given the repetitive nature of the traffic load.

It was concluded that of all the moduli available to characterize asphalt concrete mixtures, the resilient modulus is more appropriate for use in multi layer elastic programs. It represents the elastic stiffness of the material after numerous load repetitions.

A study was conducted by the New York State Department of Transportation [9] to evaluate a resilient modulus device for measuring resilient and creep moduli at 40° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 100° Fahrenheit ( 38° Centigrade ). The study was conducted since the engineering properties of asphalt concrete mixes, including their elastic and fatigue characteristics and their Poisson's ratio, are required for the structural analysis of flexible pavements for cross section design, to

detect problem mixes, and to evaluate alternative materials efficiently.

Marshall specimens were fabricated using seven state - approved top course mixes that were sampled from trucks. The resilient and creep moduli were measured at each temperature and the repeatability of the test was evaluated at the three testing temperatures, and Marshall parameters determined. The repeatability criteria used was the measurements on mutually perpendicular axes of the same specimen should not deviate more than 15 percent of the average of the two values. The test data was analyzed using a single classification analysis of variance model.

The study concluded that acceptable moduli values were obtained at 77° Fahrenheit (25° Centigrade), and 100° Fahrenheit (38° Centigrade), but test results for both the properties were unacceptable at 40° Fahrenheit (5° Centigrade). The differences among the mixes were found to be significant at the 95 - percent confidence level, and the sample sizes required to assure a maximum error of 20 percent 95 percent of the time were found to be 4 and 30 for resilient modulus and 9 and 12 for creep parameters at 100° Fahrenheit (38° Centigrade) and 77° Fahrenheit (25° Centigrade) respectively. The sample sizes were unacceptably large at 40° Fahrenheit (5° Centigrade).

Gonzalez, Kennedy, and Anagnos [10] conducted a study to develop a technique to estimate the resilient elastic characteristics of asphalt mixtures using the repeated load indirect tensile test. The study also evaluated the resilient and static moduli of elasticity and their relationships with fatigue life for the purpose of mixture design.

Laboratory prepared specimens of two asphalt mixtures containing gravel or limestone and various percentages of asphalt were tested at different temperatures. The fundamental elastic properties estimated include the instantaneous resilient modulus of elasticity, the instantaneous resilient Poisson's ratio, static modulus of elasticity, and Poisson's ratio.

The following trends were observed : The instantaneous resilient modulus of elasticity decreased with increasing temperature and increased number of load applications,

and was not affected by the magnitude of applied stress. The instantaneous resilient modulus values were generally higher than the static moduli. The study concluded that a repeated load indirect tensile test. be conducted to estimate the repeated - load elastic properties, i.e., resilient modulus of elasticity, and Poisson's ratio. It was also concluded that, an estimate of resilient modulus can be obtained without conducting a long term repeated load test. Reasonable estimates of the modulus could be obtained after about one percent of the fatigue life, but a test specimen should be subjected to a minimum of twenty five load applications before the modulus is estimated.

Kennedy and Adedimila [11] conducted a similar study on the resilient characteristics of asphalt mixtures. The study concluded that the indirect tensile test is suitable for the study of repeated load characteristics of asphalt mixtures because of the ease and simplicity in conducting the test.

## Variability in Resilient Modulus Test Results

In a 1991 ASTM paper, Brown and Foo [12] evaluated the repeatability of the ASTM D 4123 procedure for determining resilient modulus. The primary factor evaluated was the effect of the repeated stress on the measured resilient modulus. The ASTM D 4123 procedure averages the resilient modulus values of three specimens and two orientations. The following sources of error were investigated:

1. The experimental error ( $\sigma_1$ ), which is a function of the resilient modulus test apparatus and the operator.

2 Orientation variation error associated with the variation of resilient modulus values at different orientations in a specimen.

3. The Sample variation error ( $\sigma_3$ ) which is associated with the variation of resilient modulus of different samples.

Repeatability was measured for a single operator using a specific type of test equipment. The repeatability associated with different operators and different apparatus was not determined. The results were analyzed using the statistical analysis system (SAS) to investigate their repeatability and interaction.

The ASTM method of placing spring loaded Linear Variable Differential Transducers (LVDT's) in direct contact with the sample surface was studied. Two alternate procedures were investigated in which a thin membrane (paper, aluminium foil) is placed between the spring loaded LVDT's and the sample surface. A thin membrane (paper, aluminium foil) between the sample and LVDT tip was used to minimize experimental error associated with placement on small depressions or aggregates. It was concluded that of the three methods of measuring deformation, the ASTM method resulted in the least error. The study quantified the repeatability of the ASTM D 4123 procedure as a function of the stiffness of asphalt concrete. It was also concluded that the repeatability is relatively low and an increase in the number of samples would improve repeatability.

## CHAPTER III

## SAMPLE PREPARATION, EQUIPMENT AND TEST PROCEDURES

Specimens were fabricated in the laboratory using the gyratory shear compactor, Marshall hammer, and the dynamic compaction apparatus. The samples were prepared as per standardized procedures, when available. The laboratory compacted samples resemble as closely as possible the in service mixtures i.e., those produced by mixing, placement, and compaction in the field.

A coding system was developed to identify the specimens prepared with the three compaction techniques and the various sources of aggregates. Every specimen has a unique code by which the compaction technique, source of aggregate, type of mix, and the date of preparation can be identified. An example of the coding system is shown below : Sources were numbered 1 through 5 randomly.

Legend:

First digit:	Mix Type
Second Digit:	Source Number
Third Digit:	Compaction Technique
Fourth through Seventh digit :	Date of Sample Preparation
Eight Digit:	Specimen Number

COMPACTION TECHNIQUE:
MARSHALL HAMMER:
GYRATORY SHEAR:
DYNAMIC COMPACTION:

CODE

1

2

A specimen having a code of B111012-1 can be decoded as the specimen prepared using a type B mix from aggregate source 1 using Marshall compaction technique, and prepared on the 12<sup>th</sup> day in October.

## The Dynamic Compaction Apparatus

The dynamic compaction apparatus (Figure 3) was developed specifically for this study in an attempt to approximate field compaction. The device is used to prepare of 4-inch (10.16 cm), 6-inch (15.24 cm), and 8- inch (20.32 cm) in diameter specimens. The different size specimens require changing the compaction head and mold, refer to Figure 1 for a schematic of the device.

The compaction apparatus is mounted on a 3-foot (91.44 cm) \*3-foot (91.44 cm) \*3/4-inch (1.9 cm) thick base plate which is supported by castors for ease of transport. 2 inch (5.08 cm) diameter vertical pipe supports are provided on both sides of the base plate. The vertical carriage, which supports the compaction hammer slides along the vertical pipe supports. An electric winch with remote switch is provided to raise and lower the compaction hammer.

A spring supported platform 1-foot (30.48 cm)\*1-foot (30.48 cm)\*3/4-inch (1.9 cm) thick is affixed to the base plate. The purpose of the springs is to give a uniform response during compaction i.e., the rebounding plate aids in compaction. The sample base is bolted to the spring supported platform during compaction. A modified Marshall sample mold (Figure 2) and collar are used for preparing 4 inch (10.16 cm) specimens. The Marshall collar has two tabs welded on opposite sides, so that the collar / mold assembly can be bolted to the sample base. The vibration and subsequent misalignment of the sample mold and compaction head necessitated this modification. To ensure that the samples are 2 1/2-inches (6.35 cm) thick, the vertical pipe supports are drilled and pinned to provide a positive stop for the vertical carriage. The stop locations require a different pin location for the 6-inch (15.24 cm) and 8-inch (20.32 cm) diameter specimens.

### Sample Preparation Procedures

Sample preparation procedures are vital in determining realistic resilient modulus values. It is desirable to produce specimens that closely resemble field compacted asphalt concrete. Samples were prepared using the standardized procedures where available. Fifteen samples were prepared for each source and each compaction technique. All samples used identical preparation procedures with the exception of the compaction method.

The aggregates are dried to a constant weight between 105° Centigrade (221° Fahrenheit) to 110° Centigrade (230° Fahrenheit). The aggregates were then blended as per the designated percentages at an optimum asphalt content obtained from the Oklahoma Department of Transportation mix design data. A sample mixture is prepared by weighing 1200 gms of the aggregate as per the design mix requirements at an optimum asphalt content. A two minute mixing time was used on all mixes to assure uniform aggregate coating. The mixture was placed in the heated sample mold in three lifts and the surface smoothened into a convex shape.

The mixture was compacted and the height of the sample measured to ensure that it is 2 1/2-inches ( 6.35 cm ). The samples were allowed to cool prior to removal from the mold until no deformation results while removing it from the mold. The weight of the mix is adjusted ( increased / decreased ) accordingly to obtain a 2 1/2-inch ( 6.35 cm ) specimen if required.

## Texas Gyratory Shear Compactor:

The test specimens were prepared using the ASTM 4013- 81 (Reapproved 87) [15] procedure. The apparatus was set at three revolutions at an gyratory angle of 3 degrees.

## Marshall Hammer:

All Marshall specimens were prepared using the Marshall method described in

ASTM D - 1559- 89 [14]. Seventy Five blows were applied on each face of the specimen to simulate heavy traffic. The seventy five blow criteria is comparable to the gyratory compaction.

#### **Dynamic Compaction Apparatus:**

The final compaction methodology evaluated for specimen preparation is by using the dynamic compaction apparatus (Figure 1). There is no standard procedure for fabricating specimens using this apparatus. The specimens were prepared by following a using a combination of the previous two compaction methods.

The combined weight of the aggregate was equal to the weight of the aggregate used to prepare a specimen using the gyratory shear method of compaction. The percent air voids was used as the basis for comparison between the compaction techniques.

## Material Test System

The Material Test System (MTS) or hydraulic load apparatus used in this study included the following components:

- 1. An electronic hydraulic actuator panel performing the following functions:
  - A. Input control module controls calibration and sensitivity of the internal LVDT and load cell
  - B. Transducer conditioner panel signal conditioning for the load cell and LVDT signals.
  - C. Function generator frequency control of load ram (load rate) and waveform generator for cyclic loading.
- A hydraulic actuator (10 kip hydraulic ram) with an internally mounted LVDT and an externally mounted load cell.
- 3. A rigid frame which supports the hydraulic actuator assembly.
- 4. A high pressure, high volume hydraulic pump, an accumulator, and

assorted valving and piping.

5. A computer interfaced, data acquisition system.

## MTS Control / Computer Interface

An important factor associated with resilient modulus testing is the rate at which load/displacement data can be recorded and processed. A computer interfaced control system was used to control the MTS system and read the load/displacement data [Figure 5].

An analog/digital board (A/D) installed in a 386 - 16MZ computer was used for machine control and data acquisition. The operational details of the system are as follows:

- A Control Program (CP) was developed that initiates the MTS load apparatus and subsequently monitors the load displacement data. A series of three Linear Differential Variable Transducers (LVDT's) are used to measure the displacement data. A subroutine was developed for conducting the indirect tensile test for use in the resilient modulus test procedure.
- 2. The control program operates as follows:
  - A. User prompts request detailed test information including : sample code which includes aggregate source and type of compaction, sample weight, height of the sample, sample diameter, test temperature, rest period, and the axis of testing.
  - B. User prompt also requests the approval of default program parameters that include : the number of channels requiring data translation, number of data points, clock frequency ( sampling rate ), load - voltage and displacement - voltage equivalency factors, gain etc.,.
  - C. A selected load based on percentage of the indirect tensile strength depending on the testing temperature is made to act on the specimen and the MTS is initiated.
  - D. The program inputs voltages from four separate channels corresponding

to the LVDT's and the load cell. The stress and horizontal deformations are measured and the resilient modulus is calculated.

## Test Procedures

A number of tests were selected to evaluate the properties of asphalt concrete mixtures. A comprehensive outline of the test plan is presented in Figure 3

All specimens were prepared with the optimum asphalt content as determined by the Oklahoma Department of Transportation Materials Laboratory. After molding of the specimens, bulk specific gravity of all the samples was determined as per ASTM D 2726-86 [16]. Five random samples were selected from each of the three compaction devices and each of the five sources.

The Hveem stability of the selected specimens was determined as per the standard procedure designated by ASTM D 1560 [16]. The specimens were maintained at a temperature of 140° Fahrenheit ( 60° Centigrade ) for 15 hours prior to measuring the stability. The maximum specific gravity ( Rice's Method ) of the specimens were determined as per ASTM.D 2041 [16]. The percent air voids was then calculated using the bulk specific gravity and maximum specific gravity.

The indirect tensile strength test was conducted on one random sample prepared from each of the three compaction devices and using the five aggregate sources as per the procedure described in SHRP Protocol PO7 [13]. Resilient modulus test was conducted on samples as per the procedure described in SHRP Protocol PO7. The test was conducted at three temperatures (41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade )) along two diametral axes ( second axis is 45° to the first ) at three rest periods ( 0.9s, 1.9s, 2.9s ). Load intensities of 30, 15, and 5 percent of the indirect tensile strength test were used to determine the resilient modulus at 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ) respectively. The load intensity for Marshall samples was reduced to 3 percent when testing was conducted at 104° Fahrenheit to ensure adequate deformations without breaking the sample.

## Indirect Tensile Test

The indirect tensile strength test was conducted by following the procedure described in Strategic Highway Research Program (SHRP) Protocol PO7 [13]. The asphalt concrete specimen is loaded in compression along the diametral axis at a fixed deformation rate (2 inches per minute (5.08 cm per minute)). This test is required to establish the load intensity to be used in the resilient modulus procedure.

The specimen must be allowed to stand at a temperature of 77° F for 24 hours prior to testing. A modification to this procedure was used to assess the tensile strength of specimens at 104° Fahrenheit ( 40° Centigrade ). Load intensities of 30, 15, and 5 percent of the indirect tensile strength test were used to determine the resilient modulus at 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ) respectively. The indirect tensile strength is calculated using the following equation :

 $S_t = 1.273 * P_o / t [(Sin 57.2958 / D) - 1 / 2D]$ 

OR

 $S_t = 0.156 * P_o / t$  for a 4 inch (10.16 cm) diameter specimen

Where

Po = Maximum load in pounds (lbs)

t = Specimen thickness (inches)

D = Specimen diameter (inches)

## **Resilient Modulus Test**

## Introduction:

The resilient modulus test of asphalt concrete is determined by applying repetitive

applications of compressive loads in a haversine wave form. The compressive load is applied along the vertical diametral plane of a cylindrical specimen of asphalt concrete (Figure 7). The resulting vertical and horizontal deformations are measured. The resilient modulus value is calculated using the applied load, specimen dimensions and the vertical and horizontal deformations. The following test procedure is a summarization of the Strategic Highway Research Program (SHRP) Protocol PO7 procedure of July 1991 [13]. Figure 8 shows the specimen setup for resilient modulus testing.

Resilient modulus tests are conducted by repetitive application of compressive loads in a haversine wave form. Determinations are made at testing temperatures of 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade with a tolerance of  $\pm 2°$  Fahrenheit ( 1.1° Centigrade ). The specimens should be maintained at the testing temperature for 24 hours.

## **Temperature Control:**

The temperature control system used for testing consisted of a insulated enclosure with copper tubing running along the inside perimeter of the box. Water maintained at a constant temperature of 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ) was circulated through the tubing in order to maintain the sample at that temperature of testing. In addition to the above the room temperature was maintained at 50° Fahrenheit ( 10° Centigrade ), 77° Fahrenheit ( 25° Centigrade ) , 95° Fahrenheit ( 35° Centigrade ) during the time of testing in order to ensure the proper control of temperature.

## Sample Placement & Machine Setup :

The diameter and the height of each test specimen is measured prior to testing. Two orientations are evaluated for each specimen, axis one and axis two are 45° apart. The

first axis is centered top to bottom within the loading strips, refer to Figure 7. The line of contact between the specimen and the loading strip is critical for reliable results.

## Preconditioning:

The magnitude of applied loads used for preconditioning and testing at the three test temperatures is based on the tensile strength of a similar specimen determined at 77° Fahrenheit ( 25° Centigrade ). The applied load ranges from 30 to 5 percent of the tensile strength. Tensile stress levels of 30, 15, and 5 percent of the tensile strength is used in conducting the resilient modulus determinations at 41° Fahrenheit ( 5° Centigrade ), 77° Fahrenheit ( 25° Centigrade ), and 104° Fahrenheit ( 40° Centigrade ). Minimum specimen contact loads of 3, 1.5, and 0.5 percent of the tensile strength shall be maintained during the testing at all the three test temperatures. The sequence of resilient modulus testing consists of initial testing at 41° Fahrenheit ( 5° Centigrade ), followed by intermediate testing at 77° Fahrenheit ( 25° Centigrade ) and the final testing at 104° Fahrenheit ( 40° Centigrade ).

The test specimen is preconditioned along the axis prior to testing by applying a repeated haversine - shaped load pulse of 0.1 second duration followed by a rest period of 0.9 seconds duration (Figure 6)until a minimum of 10 successive horizontal deformation readings agree within ten percent. The number of load applications depend upon the test temperature. The expected ranges are

41º Fahrenheit	50 - 150
77 <sup>0</sup> Fahrenheit	50 - 100
104 <sup>0</sup> Fahrenheit	20 - 50

The minimum number of load applications for a given situation must be such that the resilient deformations are stable.

#### <u>Testing</u>:

A minimum of 30 load pulses ( each 0.1 second load pulse has a rest period of 0.9 seconds ) are applied and the measured deformations are recorded. The application of load pulses is continued beyond 30 until the range in deformations values of five successive horizontal deformation values ( i.e. from lowest to highest value ) is less than ten percent of the average of the five deformation values. The rest period is then increased to 1.9 seconds and a minimum of 30 load repetitions are applied. The rest period is then increased to 2.9 seconds. The recoverable horizontal and vertical deformations over the last five loading cycles are measured after the resilient deformations have become stable. A loading cycle consists of a load pulse and a subsequent rest period.

Once the testing is completed along the first axis the specimen is then oriented 45° from the first axis and the above procedure is repeated. After the testing is completed along both the axes, the specimen is raised to the next higher temperature and the test is conducted. The resilient modulus is calculated along each axis for each rest period and temperature by averaging the measured deformations for the last five cycles. The resilient modulus is calculated using the following equation

 $E_{RI} = \frac{P*D (0.080 + 0.297V + 0.0425V^2)}{H_I*T}$ 

$$E_{RT} = \underline{P*D (0.080 + 0.297V + 0.0425V^2)}_{H_T* T}$$

Where

E <sub>RI</sub>	=	Instantaneous modulus of elasticity, psi
E <sub>RT</sub>	=	Total modulus of elasticity, psi
Р	=	repeated load, lbf.,
Т	=	thickness of the test specimen, inches.,
D	=	diameter of the specimen, inches.,

HI	=	instantaneous recoverable deformation, inches.,
HT	=	total recoverable horizontal deformation, inches.
v	=	Poisson's Ratio assumed for each temperature.

The values of Poisson's Ratio shall be assumed as follows :

41º Fahrenheit	0.20
77º Fahrenheit	0.35
104º Fahrenheit	0.50

## CHAPTER IV

## **RESULTS AND ANALYSIS**

The following tests were performed in this study : bulk specific gravity, Hveem stability, maximum specific gravity, Air void determination, indirect tensile strength test, and the resilient modulus test. The results of those tests are presented in this chapter.

## **Bulk Specific Gravity**

The bulk specific gravity (BSG) of all the samples was determined. The BSG results are summarized in Tables 2 through 6. The following trend was observed : The BSG of the gyratory shear specimens was the highest, followed by dynamic samples, and Marshall samples. The BSG is an indicator of the relative compaction and percent air voids. The primary reason for the gyratory samples giving consistently a higher BSG is, the gyratory compaction method applies normal forces to both top and bottom faces of the asphalt mix in a cylindrically confined mold. These normal forces supplemented with a gyratory motion work the mix into a denser configuration while it is totally confined resulting in better compaction and lower air voids.

#### Hveem Stability Test

Hveem stability tests were conducted on five random samples from each of the ten source compaction combinations (Five sources and two compaction techniques). Hveem stability determinations on the Marshall samples were not conducted during the course of this study. The results of the stability tests are summarized in Tables 7 through 11. The following trends were observed : The stabilities of the dynamic samples was high followed

by the gyratory compacted samples. Generally, the higher the percent air voids the lower the Hveem stability. But inspite of the higher air voids the dynamic samples resulted in consistently higher stabilities. During the dynamic compaction, some of the larger aggregates may have broken resulting in higher percentage air voids, but still behaves as a well compacted sample resulting in higher stabilities.

### Maximum Specific Gravity and Air Voids

The maximum specific gravity of a five random samples selected from each source and compaction technique was conducted (Rice Method) and the results were summarized. The percentage of air voids in the compacted specimens is then calculated. The results of the air voids and the maximum specific gravities are tabulated in Tables 7 through 11. From the results, it can be observed that the maximum specific gravities of the specimens prepared by the three compaction techniques are relatively close to each other.

The percent air voids in the compacted mixes vary between 3 and 10 percent. The Marshall compacted specimens show a wide variation in air voids (6 percent to 10 percent). The absence of kneading action during the compaction operation, is a primary factor in the higher air voids.

The gyratory shear and dynamic compaction apparatus facilitate reorientation of the aggregate particles. The percent air voids in these range between 3 percent and 9 percent with the gyratory samples having a lower percent air voids. The gyratory samples consistently gave a lower percent air voids for the same weight of the mixture taken. The gyratory compaction method applies normal forces to both top and bottom faces of the asphalt mix in a cylindrically confined mold. These normal forces supplemented with a gyratory motion work the mix into a denser configuration while it is totally confined resulting in better compaction and lower air voids. The better orientation of the aggregate particles as a result of the gyratory action also result in the inter granular voids getting filled with more fines which result in a lower percent air voids.

#### Indirect Tensile Strength Test

The indirect tensile strength test was conducted on one sample from each source and compaction technique for use in estimating the loads to be used in the resilient modulus test. as per the designated test procedure and the results are summarized in Table 12. The following trends were observed : indirect tensile strength of gyratory shear compacted specimens was the highest followed closely by dynamic samples with the Marshall samples having the least strength. For similar type of aggregates prepared with the same percentage of asphalt cement the better the sample is compacted, the higher is the indirect tensile strength. Bulk specific gravity which can be considered a measure of compaction, shows that the gyratory samples have a higher degree of compaction when compared to dynamic and Marshall samples. This agrees with the results obtained.

#### Resilient Modulus Test

Resilient modulus tests were conducted on samples prepared from five different aggregate sources obtained from different locations in Oklahoma. Five sources were evaluated to assess the range in resilient modulus values for a typical type " B " mix. This was done primarily to observe a range in resilient modulus values. The results of the resilient modulus tests are tabulated in Tables 13 through 15. The resilient modulus test was conducted on asphalt concrete samples prepared using different aggregate sources and the three compaction techniques. The resilient modulus was evaluated for the following parameters :

Three temperatures i.e., 41º Fahrenheit ( 5º Centigrade ), 77º Fahrenheit ( 25º Centigrade ), 104º Fahrenheit ( 40º Centigrade ).

2. Three different rest periods (0.9s, 1.9s, 2.9s).

3. Two axes of loading (second axis oriented 45° to the first).

Testing on the two axes at three different rest periods result in six combinations of test conditions. It was observed that all the six combinations of test conditions for every

source of aggregate, compacted with any of the compaction techniques give approximately the same resilient modulus values. The mean for the the different test combinations and the F values for different sources and compaction techniques are tabulated in Tables 16 through 20.

The difference in resilient modulus values along the two axes may be due to the application of repeated load for preconditioning and subsequent testing along the first axis before testing along the second axis ( oriented 45° to the first ). Another reason may be the variability that exists in the sample preparation procedures and general experimental error. Overall, it can be observed that the mean values lie approximately close to each other.

The effect of different compaction techniques on the resilient modulus for each aggregate source will be analyzed on an individual basis. Testing for any interaction between the compaction technique and testing temperature, it can be observed that all the samples behave similarly with temperature, irrespective of the compaction technique. The test results also show that there is a significant difference in the resilient modulus values of samples prepared using different compaction techniques. The difference in resilient modulus values may be a result of the different actions used to compact the aggregates i.e shearing, dynamic, and impact by gyratory, dynamic and marshall compaction procedures respectively. The gyratory and dynamic compacted samples exhibit similar characteristics and the resilient modulus values are close to each other. The difference in the resilient modulus values that can be observed among different sources may be due to the variation in source as a result of their location. Figures 14 through 16 show the range in the resilient modulus values for different sources, different compaction techniques, and different temperatures. Since the differences in the resilient modulus values were not found to be significant, the mean value will be used for further analysis.

## Aggregate Source One :

Analyzing the resilient modulus values of samples obtained from source one using

three different compaction techniques (Table 16), it can be observed that there is no interaction between the compaction technique and temperature i.e., the resilient modulus values vary similarly with temperature for three different compaction techniques. The following trends were observed : The resilient modulus of gyratory samples was the highest followed by dynamic samples , and Marshall samples. Figure 9 shows that gyratory and dynamic compaction techniques exhibit similar characteristics when compared to Marshall compaction technique.

#### Aggregate Source Two :

Analyzing the resilient modulus values of samples prepared from aggregates obtained from from source two using three different compaction techniques at three temperatures (Table 17), it can be observed that there is no interaction between the compaction technique and temperature. The following trends were observed : The resilient modulus of gyratory samples was the highest followed by dynamic samples , and Marshall samples. Figure 10 shows that gyratory and dynamic compaction techniques exhibit similar characteristics when compared to Marshall compaction technique.

#### Aggregate Source Three :

Analyzing the resilient modulus values of samples prepared from aggregates obtained from source three using three compaction techniques at three temperatures (Table 18), it can be observed that the resilient modulus for three different compaction techniques does not vary similarly with temperature (F = 11.56, OSL < 0.05). Therefore overall comparison of three techniques is not feasible for this source. The following trends were observed : The resilient modulus of gyratory samples was the highest followed by dynamic samples , and Marshall samples. Figure 11 shows that gyratory and dynamic compaction techniques exhibit similar characteristics when compared to Marshall

#### compaction technique.

## Aggregate Source Four :

Analyzing the resilient modulus values of samples prepared from aggregates obtained from from source four using three different compaction techniques at three temperatures (Table 19), it can be observed that there is no interaction between the compaction technique and temperature. The following trends were observed : The resilient modulus of gyratory samples was the highest followed by dynamic samples , and Marshall samples. Figure 12 shows that gyratory and dynamic compaction techniques exhibit similar characteristics when compared to Marshall compaction technique.

#### Aggregate Source Five :

Marshall samples prepared using this particular aggregate source were not tested, since the depressions present in the sample were higher than the minimum values [11], and hence the samples were not used for testing and evaluation. Analyzing the resilient modulus values of samples prepared from aggregates obtained from source five (Table 20) it can be observed that there is no interaction between compaction technique and temperature. The following trends were observed : The resilient modulus of gyratory samples was higher than the dynamic samples. Figure 13 shows that gyratory and dynamic compaction techniques exhibit similar characteristics when compared to Marshall compaction technique.

## **Overall Trends**

The resilient modulus test was conducted on asphalt concrete samples prepared using different aggregate sources and the three compaction techniques. The resilient modulus was evaluated for the following parameters :

1. Three temperatures i.e., 41º Fahrenheit ( 5º Centigrade ), 77º Fahrenheit

(25º Centigrade), 104º Fahrenheit (40º Centigrade).

- 2. Three different rest periods (0.9s, 1.9s, 2.9s).
- 3. Two axes of loading ( second axis oriented 45<sup>o</sup> to the first ).

Based on the test results obtained and their discussion the following trends can be inferred : The resilient modulus of the gyratory samples are the highest followed by the dynamic samples, and the Marshall samples for all the five aggregate sources and different temperatures. A similar trend is observed between resilient modulus and temperature i.e., the resilient modulus decreases with an increase in temperature for each of the five different sources and the compaction techniques. Previous studies [7] on the evaluation of compaction devices have shown that gyratory compaction is more effective than Marshall compaction. Thus Marshall hammer will not be evaluated for further analysis. A comparison of the resilient modulus values on samples prepared by using the gyratory shear compactor and the dynamic compaction apparatus at three different temperatures can be observed in Figure 17.
#### CHAPTER V

#### CONCLUSIONS

The purpose of the study was to determine the resilient modulus of a typical type 'B' mix. A secondary objective was the evaluation of three compaction devices to determine their ability to approximate field compaction. The resilient modulus tests were conducted at three temperatures, three rest periods, and along two axes. Based on the results obtained from this test program, it can be concluded that

- 1. The resilient modulus values of all compacted mixes increase with decreasing temperatures independent of the compaction technique.
- 2. The effect of varying the rest period is not significant.
- 3. The difference in resilient modulus values measured on the two axes (second is 45 degrees to the first) is not significant.
- 4. The resilient modulus of the gyratory compacted samples and the dynamic compacted samples are approximately equal.
- 5. The dynamic compaction apparatus, to produce reliable results may be used for preparing and testing large size samples [ 6 inch ( 15.24 cm ), and 8 inch ( 20.32 cm ) diameter respectively ] by changing the compaction head and mold.

#### CHAPTER VI

#### RECOMMENDATIONS

- Field cores of "newly constructed" pavements using the aggregate sources considered in the study should be tested for resilient modulus. A comparison should be made with laboratory compacted samples and the effectiveness of the dynamic compaction apparatus to approximate field compaction should be determined
- 2. The effectiveness of the dynamic compaction apparatus to produce large size samples [i.e., 6 inch (15.24 cm) and 8 inch (20.32 cm)] prepared using aggregate with sizes greater than 1 inch (2.54 cm) should be determined by comparing it with 4 inch (10.16 cm) diameter samples prepared using the same aggregate size.
- 3. Since different rest periods, and two axes do not give any significantly different resilient modulus values, future resilient modulus testing can be conducted along one axis with one rest period..
- 4. The effect of applying various percentages of indirect tensile strength on the specimen for resilient modulus determinations should be studied.
- There is a need to look at an increased scope of temperature effect on the resilient modulus.
- 6. Assess gradation changes in aggregate due to various compaction techniques, principally the dynamic compaction apparatus.

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# APPENDIXES

APPENDIX A

**FIGURES** 



Figure 1. Schematic of the Dynamic Compaction Apparatus



Figure 2. Sample Mold Assembly For Dynamic Compaction Apparatus



Figure.3. The Dynamic Compaction Apparatus



Figure 4. Flow Chart for the Project



Figure 5. Overall View of Resilient Modulus Testing System



Figure 6. Typical Load and Deformation Versus Time Relationships for Resilient Modulus Test



Figure 7. Loading of the Asphalt Concrete Specimen



Figure 8. Specimen Setup For Resilient Modulus Testing



Figure 9 Resilient Modulus at 41, 77, 104 degree Fahrenheit for Source 1



Figure 10 Resilient Modulus at 41, 77, 104 degree Fahrenheit for Source 2



Figure 11 Resilient Modulus at 41, 77, 104 degree Fahrenheit for Source 3



Figure 12. Resilient Modulus at 41, 77, 104 degree Fahrenheit for Source 4



Figure 13. Resilient Modulus at 41, 77, 104 degree Fahrenheit for Source 5



Figure 14. Resilient Modulus for Marshall compacted samples for different sources



TEMPERATURE (FAHRENHEIT)

Figure 15. Resilient Modulus for Gyratory compacted samples for different sources



TEMPERATURE (FAHRENHEIT)

Figure 16. Resilient Modulus for Dynamic compacted samples for different sources



Figure 17 Comparison of Resilient Modulus values for Gyratory and Dynamic compacted specimens

# APPENDIX B

# LIST OF TABLES

Mixture	Asphalt Concrete Mixture Type			
Туре	A	В	C	D
Sieve Size		Percent Passing		
				- <u> </u>
1 1/2"	100	-	-	-
1"	90-100	-	-	-
3/4"	•	100	-	-
1/2"	70-90	90-100	100	-
3/8"	-	70-90	90-100	100
No 4	40-65	45-70	60-80	80-100
No 10	25-45	25-50	35-60	50-90
No 40	10-26	12-30	15-35	20-50
No 80	6-18	7-20	8-22	10-30
No 200	*	*	*	5-15

#### AGGREGATE GRADATION FOR OKLAHOMA DEPARTMENT OF TRANSPORTATION ASPHALT CONCRETE MIXES

\* For types A, B, C asphalt concrete, the ratio of the percent passing the no. 200 sieve to the percent asphalt cement shall be a minimum of 0.6 to a maximum of 1.2. The ratio will establish the master range for the job mix on the no. 200 sieve.

.

		Type of Compaction	1	
Sample No.	Marshall	Gyratory	Dynamic	
1	2.279	2.400	2.338	
2	2.276	2.412	2.352	
3	2.246	2.400	2.323	
4	2.233	2.412	2.343	
5	2.270	2.410	2.321	
6	2.250	2.411	2.341	
7	2.261	2.411	2.330	
. 8	2.255	2.395	2.350	
9	2.260	2.408	2.336	
10	2.269	2.405	2.329	
11	2.252	2.402	2.333	
12	2.266	2.401	2.346	
13	2.278	2.394	2.320	
14	2.270	2.400	2.345	
15	2.267	2.405	2.326	

		Type of Compaction	l	
Sample No.	Marshall	Gyratory	Dynamic	
1	2.343	2.458	2.403	
2	2.348	2.450	2.399	
3	2.339	2.420	2.394	
4	2.354	2.458	2.370	
5	2.336	2.417	2.417	
6	2.320	2.423	2.417	
7	2.312	2.413	2.392	
8	2.341	2.404	2.399	
9	2.335	2.425	2.437	
10	2.310	2.429	2.420	
. 11	2.338	2.386	2.412	
12	2.331	2.426	2.436	
13	2.334	2.418	2.405	
14	2.349	2.420	2.399	
15	2.340	2.406	2.416	

		Type of Compaction	L	<u> </u>
Sample No.	Marshall	Gyratory	Dynamic	
1	2.197	2.302	2.252	
2	2.220	2.301	2.257	
3	2.242	2.303	2.254	
4	2.172	2.294	2.262	
5	2.217	2.318	2.260	
6	2.209	2.294	2.267	
7	2.202	2.289	2.280	
8	2.210	2.294	2.258	
9	2.223	2.301	2.276	
10	2.201	2.315	2.249	
11	2.217	2.280	2.247	
12	2.213	2.313	2.234	
13	2.209	2.285	2.249	
14	2.223	2.303	2.258	
15	2.216	2.292	2.245	

		Type of Compaction		
Sample No.	Marshall	Gyratory	Dynamic	
1	2.202	2.335	2.337	
2	2.154	2.330	2.333	
3	2.222	2.353	2.345	
4	2.221	2.379	2.319	
5	2.180	2.349	2.243	
6	2.204	2.246	2.259	
7	2.194	2.265	2.288	
8	2.212	2.355	2.304	
9	2.203	2.285	2.269	
10	2.200	2.358	2.282	
11	2.215	2.309	2.253	
12	2.179	2.345	2.261	
13	2.215	2.366	2.259	
14	2.214	2.367	2.261	
15	2.190	2.254	2.244	
· ·				

	F	Type of Compaction				
Sample No.	Marshall	Gyratory	Dynamic			
1	2.293	2.408	2.363			
2	2.308	2.410	2.351			
3	2.314	2.404	2.362			
4	2.289	2.410	2.368			
5	2.306	2.402	2.373			
6	2.284	2.419	2.372			
7	2.298	2.406	2.332			
8	2.302	2.407	2.335			
9	2.290	2.409	2.314			
10	2.332	2.411	2.319			
. 11	2.307	2.417	2.343			
12	2.322	2.406	2.361			
13	2.296	2.414	2.332			
14	2.291	2.413	2.332			
15	2.302	2.413	2.315			

<sup>1</sup> SOURCE	AC (%)	<sup>2</sup> BSG	<sup>3</sup> MSG	VOIDS (%)	STABILITY
12-2	5.0	2.412	2.485	3.1	45
12-4	5.0	2.410	2.484	3.3	42
12-8	5.0	2.408	2.479	2.9	44
12-9	5.0	2.405	2.476	3.0	41
12-15	5.0	2.405	2.480	3.1	45

# AIR VOIDS AND STABILITY TEST RESULTS FOR SOURCE 1 SPECIMENS

<sup>1</sup> Include Source, Type of Compaction, Sample Number

<sup>2</sup> BSG = Bulk Specific Gravity

<sup>3</sup> MSG = Maximum Specific Gravity

<sup>1</sup> SOURCE	AC (%)	<sup>2</sup> BSG	<sup>3</sup> MSG	VOIDS (%)	STABILITY
21-2	4.6	2.348	2.527	7.1	-
21-7	4.6	2.312	2.532	8.7	-
21-10	4.6	2.310	2.530	8.7	-
21-12	4.6	2.331	2.531	7.9	-
21-13	4.6	2.334	2.527	7.6	-
22-5	4.6	2.417	2.523	4.2	42
22-9	4.6	2.425	2.520	3.8	43
22-12	4.6	2.426	2.524	3.9	44
22-13	4.6	2.418	2.517	3.9	43
22-14	4.6	2.420	2.520	4.0	42
23-3	4.6	2.394	2.519	5.0	48
23-4	4.6	2.370	2.524	6.1	47
23-6	4.6	2.417	2.528	4.4	58
23-11	4.6	2.412	2.525	4.5	57
23-15	4.6	2.416	2.520	4.1	62

#### AIR VOIDS AND STABILITY TEST RESULTS FOR SOURCE 2 SPECIMENS

<sup>1</sup> Include Source, Type of Compaction, Sample Number
<sup>2</sup> BSG = Bulk Specific Gravity

<sup>3</sup> MSG = Maximum Specific Gravity \* Marshall Stability not determined

<sup>1</sup> SOURCE	AC (%)	<sup>2</sup> BSG	<sup>3</sup> MSG	VOIDS (%)	STABILITY
31-2	5.3	2.220	2.461	9.7	-
31-9	5.3	2.223	2.443	8.8	-
31-11	5.3	2.217	2.450	9.5	-
31-12	5.3	2.213	2.446	10.0	-
31-15	5.3	2.216	2.440	9.0	-
32-1	5.3	2.302	2.439	5.6	39
32-3	5.3	2.303	2.440	5.6	40
32-9	5.3	2.301	2.432	5.4	41
32-11	5.3	2.280	2.461	7.3	39
32-13	5.3	2.285	<b>2.4</b> 51	6.8	39
33-3	5.3	2.254	2.455	8.2	48
33-6	5.3	2.267	2.456	7.7	51
33-8	5.3	2.258	2.449	7.8	49
33-10	5.3	2.249	2.460	8.6	47
33-15	5.3	2.245	2.464	8.9	51

# AIR VOIDS AND STABILITY TEST RESULTS FOR SOURCE 3 SPECIMENS

<sup>1</sup> Include Source, Type of Compaction, Sample Number
<sup>2</sup> BSG = Bulk Specific Gravity

<sup>3</sup> MSG = Maximum Specific Gravity Marshall Stability not determined

<sup>1</sup> SOURCE	AC (%)	<sup>2</sup> BSG	<sup>3</sup> MSG	VOIDS (%)	STABILITY
42-1	5.2	2.335	2.441	4.3	40
42-2	5.2	2.330	2.448	4.8	45
42-7	5.2	2.265	2.458	7.8	41
42-11	5.2	2.309	<b>2.44</b> 1	5.4	45
42-13	5.2	2.366	2.442	3.1	45
43-1	5.2	2.337	2.454	4.8	55
43-3	5.2	2.345	2.445	4.1	41
43-8	5.2	2.304	2.446	5.8	50
<b>43-</b> 11	5.2	2.253	2.449	8.0	50
43-14	5.2	2.261	2.449	7.7	49

#### AIR VOIDS AND STABILITY TEST RESULTS FOR SOURCE 4 SPECIMENS

<sup>1</sup> Include Source, Type of Compaction, Sample Number

<sup>2</sup> BSG = Bulk Specific Gravity

<sup>3</sup> MSG = Maximum Specific Gravity

<sup>1</sup> SOURCE	AC (%)	<sup>2</sup> BSG	<sup>3</sup> MSG	VOIDS (%)	STABILITY
51.0	4.5	0 000	0.400	<b>7</b> 0	
51-2	4.7	2.308	2.483	7.0	-
51-3	4.7	2.314	2.472	6.4	-
51-4	4.7	2.289	2.471	7.4	-
51-7	4.7	2.298	2.478	7.3	-
51-11	4.7	2.307	2.471	6.6	-
52-3	4.7	2.404	2.274	2.8	43
52-5	4.7	2.402	2.476	3.0	44
52-12	4.7	2.406	2.479	2.9	42
52-13	4.7	2.414	2.475	2.5	41
52-14	4.7	2.413	2.476	2.54	38
53-1	4.7	2.363	2.480	4.7	61
53-3	4.7	2.362	2.482	4.8	50
53-6	4.7	2.372	2.480	4.3	57

#### AIR VOIDS AND STABILITY TEST RESULTS FOR SOURCE 5 SPECIMENS

<sup>1</sup> Include Source, Type of Compaction, Sample Number

<sup>2</sup> BSG = Bulk Specific Gravity

<sup>3</sup> MSG = Maximum Specific Gravity Marshall Stability not determined

Sample Code	Indirect Tensile Strength (psi)
B11-07	68.402
B12-05	127.075
B13-03	97.730
B21-14	89.569
B22-08	99.902
B23-07	117.627
B31-14	73.887
B32-08	95.367
B33-02	91.992
B41-12	52.548
B42-04	132.840
B43-12	89.862
B51-06	90.454
B52-04	117.801
B53-12	103.629

# INDIRECT TENSILE STRENGTH DATA
SOURCE							
		0	.9s	1.	9s	2.	.9s
	TEMP	Axis 1	Axis 2	Axis 1	Axis 2	Axis 1	Axis 2
1	41 <sup>0</sup> F	630	652	577	649	642	644
		589	519	561	565	553	546
		812	514	802	497	735	781
		568	765	556	714	520	752
2		661	544	639	524	565	502
		741	412	726	404	664	408
		470	570	459	592	423	576
		656	600	645	605	592	580
3		381	571	367	478	355	464
		415	403	431	400	395	384
		425	225	421	230	413	229
		399	318	404	327	387	545
4		451	498	419	417	387	381
		573	295	600	299	575	293
		426	438	413	425	419	499
		361	689	369	630	366	620
5		415	633	440	661	442	630
		539	953	507	877	510	715
		564	564	564	564	564	564
		519	450	504	433	482	439
1	77 <sup>0</sup> F	205	233	200	223	195	241
		320	337	318	329	306	330
		252	251	288	241	333	261
		282	249	285	249	290	255
2		398	314	391	300	387	304
		355	264	370	243	375	235
		394	402	380	389	396	388
		267	286	293	288	294	278
3		194	283	195	283	198	295
		315	260	305	247	301	247
		194	344	200	322	204	313
		256	225	247	222	240	237

## RESILIENT MODULUS ( KSI ) AT 41, 77, 104 DEGREE FAHRENHEIT USING MARSHALL COMPACTION

.

.

SOURCE				REST PE	RIODS		
		0.	.9s	1.	9s	2.	9s
	TEMP	Axis 1	Axis 2	Axis 1	Axis 2	Axis 1	Axis 2
4	-	322	259	345	378	323	395
		525	390	465	441	478	440
		363	353	365	351	363	357
		351	360	361	360	388	360
5		339	501	320	513	324	509
		415	408	480	427	512	435
		268	204	276	200	288	196
		251	283	246	283	251	292
1	104 <sup>0</sup> F	106	77	114	72	108	91
		114	86	117	87	110	90
		96	117	91	111	81	107
		99	99	99	99	99	99
2		174	111	168	118	180	131
		172	276	186	288	195	295
		183	150	189	133	199	122
		126	151	140	123	148	117
3		164	146	151	158	153	170
		114	129	121	137	147	129
		182	127	181	150	185	131
		77	100	79	68	82	55
4		202	183	184	189	190	186
		196	118	155	106	182	105
		147	204	142	191	138	167
		143	234	132	265	139	218

SO	URCE			REST PE	RIODS		
		0	.9s	1.	9s	2.	9s
	TEMP	Axis I	Axis 2	Axis 1	Axis 2	Axis 1	Axis 2
1	41 <sup>0</sup> F	1210 574	1209 572	1085 573	1326 575	1159 548	1205 747
		894 972	1008	918 930	1018	970 963	680 1006
2		629	538 772	634 852	536 776	618 816	584 786
		1003	1206	991	1197	866	1387
		928	810	1004	783	996	767
3		869	800 845	803	950 804	800	735
		903 • 796	845 544	920 795	804 550	003 780	524
		902	565	883	567	900	576
4		627 572	365	567	398	539	341
		575 618	489 766	280 585	490 713	021 577	482 669
		583	398	577	405	576	399
5		1285	354	1054	371	1004	458
		407 556	1350	626	1272	700	1213
		574	531	599	548	602	528
1	770F	510	589	507	606	498	579
		388 545	515 610	523	511 607	590 530	400 593
		774	593	658	637	637	628
2		479	639	473	663	471	625
		465	698 610	436	668 503	393 607	000 574
		588	525	563	533	563	558
3		372	486	374	493	389	497
		487	262	444	250	455	238
		557 375	451 375	507 391	439 367	520 390	446 359

### RESILIENT MODULUS (KSI) AT 41, 77, 104 DEGREE FAHRENHEIT USING GYRATORY COMPACTION

SO	URCE		- ·	REST PE	RIODS	_	
	TEMP	0 Axis 1	.9s Axis 2	1. Axis 1	9s Axis 2	2. Axis 1	9s Axis 2
		•					
4		567	359	558	345	571	334
		530	605	499	571	512	598
		498	394	476	415	512	428
		550	355	548	347	572	345
5		252	441	243	458	251	458
		309	287	310	291	315	289
		488	417	477	400	473	416
		247	236	265	243	260	253
1	104 <sup>0</sup> F	407	411	408	416	441	409
		329	206	287	204	290	211
		370	217	318	215	357	217
		298	310	289	307	216	294
2		243	275	245	271	262	273
		192	238	210	260	189	247
		212	227	217	217	228	219
		326	217	325	218	321	213
3		213	154	218	152	236	153
		177	201	165	218	157	224
		153	179	159	178	163	165
		226	145	229	151	226	170
4		291	239	316	302	312	292
		261	197	260	202	268	213
		192	147	182	153	169	160
		231	211	223	219	238	222
5		238	122	250	114	261	122
		228	169	208	177	218	173
		177	116	182	124	179	124
		143	179	140	186	144	227

SO	URCE			REST PE	RIODS		
		0	.9s	1.	9s	2.	9s
	TEMP	Axis 1	Axis 2	Axis 1	Axis 2	Axis 1	Axis 2
1	41 <sup>0</sup> F	1020 804 859	720 750 886	1020 700 848	706 720 859	1004 720 876	722 614 719
		771	863	724	834	705	817
2		505 580 560 860	547 825 910 466	654 684 576 806	541 837 865 457	692 671 544 803	536 700 950 477
3		491 517 495 525	668 721 626 679	470 473 516 480	690 730 636 687	465 445 522 479	700 680 608 579
4		357 931 486 401	428 700 922 829	374 905 487 452	433 699 845 771	397 937 500 469	426 616 864 836
5		525 1230 370 575	767 1015 697 633	514 1013 380 550	808 947 622 668	548 1065 404 531	761 900 616 700
1	77 <sup>0</sup> F	380 276 475 214	387 235 421 279	434 313 510 217	412 310 413 285	406 296 485 216	423 301 420 306
2		789 563 620 669	491 643 592 486	743 530 615 615	461 617 580 479	743 545 610 590	457 604 552 509
3		275 308 309 397	280 308 214 363	264 307 304 392	267 324 212 355	266 329 300 413	272 342 206 371

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## RESILIENT MODULUS ( KSI ) AT 41, 77, 104 DEGREE FAHRENHEIT USING DYNAMIC COMPACTION

RCE	0	0.5	REST PE	RIODS	2	Q.c
TEMP	Axis 1	Axis 2	Axis 1	Axis 2	Axis 1	Axis 2
	315	466	296	525	300	545
	335	487	330	475	358	496
	509	419	507	396	532	387
	423	387	443	431	436	425
	334	546	315	561	320	581
	247	187	225	187	249	186
	335	374	353	370	291	363
	284	329	274	315	296	334
104 <sup>0</sup> F	144	114	166	130	161	126
	190	142	186	150	201	168
	126	259	154	275	168	279
	140	125	148	122	150	130
	273	237	233	267	246	264
	182	248	178	239	196	256
	189	145	194	143	195	155
	139	228	157	231	162	227
	243	149	238	168	256	176
	109	161	113	140	107	143
	178	125	175	123	176	126
	83	87	66	87	73	102
	223	157	223	140	221	145
	332	223	328	242	367	224
	296	228	269	240	257	236
	246	212	247	224	250	234
	109	122	113	125	115	129
	141	137	156	143	150	130
	159	99	147	100	149	97
	224	177	223	187	243	173
	RCE TEMP	RCE       0         TEMP       Axis 1       0         315       335       509         423       334       247         335       284       104°F       144         104°F       144       190       126         140       273       182       189         139       243       109       178         83       223       332       296         246       109       141       159         224       109       141       159	$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$\begin{array}{c c c c c c c c c c c c c c c c c c c $

COMPACTI	ON		TEMPERATURE	ļ		
METHOD	)	41	77	104		
MARSHAL	L	632 556 690 646	216 323 271 268	95 101 101 99		
	TOTAL TRT. MEAN	2524 631	1078 270	396 99		
GYRATORY		1199 598 809 983	548 445 568 655	415 255 282 286		
	TOTAL TRT. MEAN	3589 897	2216 554	1238 310		
DYNAMIC		865 718 991 1086	308 233 454 253	140 173 210 136		
	TOTAL TRT. MEAN	3210 803	1403 351	659 165		
			ANOVA			
SOURCE		DF	SS	MS	F	OSL>
TOTAL		35	2830168			
TREATMENT 8		8	2538067	317258.4	29.32	5
METHOD 2		2	389765.1	194882.5	18.01	4
TEMPERATURE 2		2	2127778	1063889	98.33	9
MXT		4	20524.44	5131.111	0.474	0.10
ERROR 2		27	358335.5	13271.69		

COMPACT	ION		]	<b>TEMPERATURE</b>			
METHOI	)		41	77	104		
MARSHAL	L		573	349	147		
			559	307	255		
			515	392	163		
			613	284	134		
	TOTAL		2260	1332	679		
	TRT. MEAN	1	565	333	170		
GYRATOR	Y		590	558	262		
			820	555	223		
			1108	641	220		
			881	222	270		
	TOTAL		3399	2309	975		
	TRT. MEAN	J	850	577	244		
DYNAMIC			579	514	253		
			716	584	217		
			734	495	170		
			645	558	191		
	TOTAL		2674	2351	831		
	TRT. MEAN	J	669	488	208		
				ANOVA			
				AIOVA			
SOURCE		DF		SS	MS	F	OSL>
TOTAL		35		1907999			
TREATME	NT	8		1726128	215765.9	32.03	31
METHOD		2	•	242430.5	121215.3	17.99	95
TEMPERATURE 2		2		1426823	713411.6	105.9	91
MXT		4		56873.83	14218.46	2.110	0.10
ERROR		27		181871.5	6735.981		

METHOD         41         77         104           MARSHALL         436         241         157           405         279         130           324         263         159           397         238         77           TOTAL         1562         1021         523           TRT. MEAN         391         255         131           GYRATORY         826         435         188           873         356         190         665           655         487         166           732         376         191           TOTAL         3096         1654         735           TRT. MEAN         774         414         184           DYNAMIC         581         271         205           594         320         129         567         258           572         382         83         142           ANOVA         X         X         X         X           SOURCE         DF         SS         MS         F         OSL           TOTAL         35         1536296         1536296         1536296         1116001         205 02	COMPACTIO	N	T	EMPERATURE			
MARSHALL       436       241       157         405       279       130         324       263       159         397       238       77         TOTAL       1562       1021       523         TRT. MEAN       391       255       131         GYRATORY       826       435       188         873       356       190         665       487       166         732       376       191         TOTAL       3096       1654       735         TRT. MEAN       774       414       184         DYNAMIC       581       271       205         594       320       129         567       258       151         572       382       83         TOTAL       579       308       142         ANOVA       X       X       X         SOURCE       DF       SS       MS       F       OSL         TOTAL       35       1536296       1536296       1469288       183660.9       74.003         METHOD       2       237668.7       118834.4       47.883       225.02	METHOD		41	77	104		
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	MARSHALL		436 405 324 397	241 279 263 238	157 130 159 77		
GYRATORY $\begin{array}{c} 826 \\ 873 \\ 873 \\ 665 \\ 732 \end{array}$ $\begin{array}{c} 435 \\ 190 \\ 666 \\ 732 \end{array}$ $\begin{array}{c} 188 \\ 190 \\ 666 \\ 732 \end{array}$ TOTAL TRT. MEAN $\begin{array}{c} 3096 \\ 774 \end{array}$ $\begin{array}{c} 1654 \\ 144 \end{array}$ $\begin{array}{c} 735 \\ 732 \end{array}$ DYNAMIC $\begin{array}{c} 581 \\ 594 \\ 572 \end{array}$ $\begin{array}{c} 271 \\ 320 \\ 129 \\ 567 \\ 258 \end{array}$ $\begin{array}{c} 205 \\ 151 \\ 572 \end{array}$ TOTAL TRT. MEAN $\begin{array}{c} 2314 \\ 579 \end{array}$ $\begin{array}{c} 1231 \\ 308 \end{array}$ $\begin{array}{c} 568 \\ 142 \end{array}$ TOTAL TRT. MEAN $\begin{array}{c} 2314 \\ 579 \end{array}$ $\begin{array}{c} 1231 \\ 308 \end{array}$ $\begin{array}{c} 568 \\ 142 \end{array}$ SOURCE       DF       SS       MS       F       OSL         TOTAL       35 $\begin{array}{c} 1536296 \end{array}$ TOTAL $\begin{array}{c} 735 \\ 1536296 \end{array}$ TREATMENT       8 $\begin{array}{c} 1469288 \\ 183660.9 \end{array}$ $\begin{array}{c} 74.003 \\ 74.003 \end{array}$ METHOD       2 $\begin{array}{c} 237668.7 \\ 118834.4 \end{array}$ $\begin{array}{c} 47.883 \\ 47.883 \end{array}$ $\begin{array}{c} 74.003 \\ 74.003 \end{array}$ METHOD $\begin{array}{c} 2 \\ 237668.7 \\ 118834.4 \end{array}$ $\begin{array}{c} 74.003 \\ 74.003 \\ 74.003 \end{array}$ $\begin{array}{c} 74.003 \\ 74.0$	ר ז	TOTAL TRT. MEAN	1562 391	1021 255	523 131		
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	GYRATORY		826 873 665 732	435 356 487 376	188 190 166 191		
DYNAMIC $581$ $594$ $572$ $271$ $320$ $258$ $382$ $205$ $129$ $151$ $83$ TOTAL TRT. MEAN $2314$ $579$ $1231$ $308$ $568$ $142$ NOVASOURCEDFSSMSFOSL TOTAL $35$ $1536296$ TREATMENT8 $1469288$ $183660.9$ $74.003$ METHOD2 $237668.7$ $118834.4$ $47.883$	ר ז	IOTAL IRT. MEAN	3096 774	1654 414	735 184		
TOTAL TRT. MEAN       2314 579       1231 308       568 142         ANOVA       ANOVA       F       OSL         SOURCE       DF       SS       MS       F       OSL         TOTAL       35       1536296       TREATMENT       8       1469288       183660.9       74.003         METHOD       2       237668.7       118834.4       47.883	DYNAMIC		581 594 567 572	271 320 258 382	205 129 151 83		
ANOVA         SOURCE       DF       SS       MS       F       OSL         TOTAL       35       1536296		TOTAL TRT. MEAN	2314 579	1231 308	568 142		
SOURCE       DF       SS       MS       F       OSL         TOTAL       35       1536296				ANOVA			
TOTAL       35       1536296         TREATMENT       8       1469288       183660.9       74.003         METHOD       2       237668.7       118834.4       47.883	SOURCE	I	OF	SS	MS	F	OSL<
TREATMENT       8       1469288       183660.9       74.003         METHOD       2       237668.7       118834.4       47.883	TOTAL	3	35	1536296			
METHOD 2 237668.7 118834.4 47.883	TREATMEN	s ۲	3	1469288	183660.9	74.00	03
	METHOD	2	2	237668.7	118834.4	47.88	83
IEMPERATURE 2 1110891 558445.4 225.02	TEMPERATU	JRE 2	2	1116891	558445.4	225.0	02
MXT 4 114727.9 28681.99 11.557 0.0	MXT	2	4	114727.9	28681.99	11.5	57 0.05
ERROR 27 67008 2481.778	ERROR	2	27	67008	2481.778		

COMPACT	ION		TEMPERATUR	E	·
METHOI	)	41	77	104	
MARSHAL	L	426 436 427	337 457 250	189 144	
		437 506	363	189	
	TOTAL TRT. MEAN	1805 451	1516 379	687 172	
GYRATORY		473 540 655 490	456 553 454 453	292 234 167 224	
	TOTAL TRT. MEAN	2158 540	1916 479	917 229	
DYNAMIC		403 798 684 626	408 414 458 424	185 286 254 236	
	TOTAL TRT. MEAN	2511 628	1704 426	961 240	
			ANOVA		
SOURCE		DF	SS	MS	F OSL>
TOTAL		35	889828.8		
TREATME	NT	8	750948	93868.5	18.249
METHOD		2	65687.17	32843.58	6.3852
TEMPERA	FURE	2	657793.5	328896.8	63.941
MXT		4	27467.33	6866.833	1.3350 0.10
ERROR		27	13880.8	5143.731	

COMPACT	ION			TEMPE	RATURI	E	<u></u>	·
METHOI	D		41		77	104		
GYRATOR'	Y		754 533 953		351 300 445	185 196 150	· · ·	
	TOTAL TRT. MEA	N	564 2804 701		251 1347 337	170 701 175		
DYNAMIC			654 1028 515 610		443 214 348 305	119 143 125 205		
	TOTAL TRT. MEA	N	2807 702		1310 328	592 148		
				ANC	VA			
SOURCE		DF		SS		MS	F	OSL>
TOTAL		23		153	8574			
TREATMEN	NT	5		1219	9646	243929.3	13.76	57
METHOD		1		852.	.0417	852.0417	0.048	81
TEMPERATURE 2		2		121	7989	608994.5	34.37	71
MXT		4		805.	.3333	402.6667	0.022	27 0.10
ERROR	•	18		318	927.3	17718.18		

SOURCE	TEMPERATURE	COMPACTION	MEAN	F
1	41	M	631	0.283
1	41	G	897	0.020
1	41	D	803	0.809
1	77	M	270	0.084
1	77	G	554	0.426
1	77	D	351	0.086
1	104	M	99	0.603
1	104	G	309	0.468
1	104	D	165	0.121
2	41	M	565	0.981
2	41	G	850	0.047
2	41	D	669	0.064
2	77	M	333	0.100
2	77	G	577	0.771
2	77	D	488	0.404
2	104	M	170	0.043
2	104	G	244	0.053
2	104	D	208	0.377
3	41	M	390	0.165
3	41	G	774	2.330
3	41	D	578	0.516
3	77	M	255	0.677
3	77	G	414	0.383
3	77	D	307	0.286
3	104	M	131	0.121
3	104	G	184	0.429
3	104	D	142	0.140
4	41	M	451	0.059
4	41	G	539	0.862
4	41	D	628	0.476

#### F VALUES AT 41, 77, 104 DEGREES FAHRENHEIT

SOURCE	TEMPERATURE	COMPACTION	MEAN	F
4	77	M	379	0.394
4	77	G	479	1.950
4	77	D	426	0.616
4	104	M	172	0.372
4	104	G	229	0.533
4	104	D	240	2.070
5	41	M	564	1.080
5	41	G	701	0.006
5	41	D	702	0.308
5	77	M	343	0.066
5	77	G	337	0.073
5	77	D	327	0.425
5	104	G	175	1.304
5	104	D	148	0.460

TABLE 21 (Continued)

M = MARSHALL COMPACTION G = GYRATORY COMPACTION D = DYNAMIC COMPACTION

# DESIGN MIX FOR SOURCE 1 PROJECT NO : IR-40-5(171)181 05487(04)

Percent	5/8" Chips	5/8" Mill	Stone Sand	Sand	Job Formula
passing		Run			
3/4"	100	100			100
1/2"	95	. 97			98
3/8"	70	86	100		87
No 4	14	61	99		63
No 10	3	49	68	100	47
No 40	2	19	20	77	24
No 80	1	13	9	19	10
No 200	0.9	8.8	4.3	1.5	4.8

% Asphalt Cement Used : 5.0 %

MATERIAL	SOURCE	% USED
5/8" Chips	Meridian Aggregate @ Mill	25
-	Creek, OK.	
5/8" Mill Run	Meridian Aggregate @ Mill	40
	Creek, OK.	
Stone Sand	Dolese Co. @ Konawa OK	20
Sand	White Pit @ Harrah, OK	15

# DESIGN MIX FOR SOURCE 2 PROJECT NO: IR-40-4(340)86 11255(04)

Percent	5/8" Chips	Screening	Stone Sand	Fill Sand	Job Formula
passing					
3/4"	100				100
1/2"	93				98
3/8"	42	100			80
No 4	5	96	100	100	66
No 10	3	53	81	99	48
No 40	2	21	24	90	25
No 80	2	14	8	27	11
No 200	1.5	10.1	43	3.1	5.0

% Asphalt Cement Used : 4.6 %

SOURCE	% USED	
The Dolese Co. @	35	
Cooperton, OK.		
The Dolese Co. @	32	
Cooperton, OK.		
Dolese Co. @ Richard	18	
Spur, OK		
The Dolese Co. @ Yukon,	15	
OK.		
	SOURCE The Dolese Co. @ Cooperton, OK. The Dolese Co. @ Cooperton, OK. Dolese Co. @ Richard Spur, OK The Dolese Co. @ Yukon, OK.	SOURCE% USEDThe Dolese Co. @35Cooperton, OK.32The Dolese Co. @32Cooperton, OK.32Dolese Co. @ Richard18Spur, OK15OK.15

# DESIGN MIX FOR SOURCE 3 PROJECT NO: RS-4720(110) 06877(04)

Percent Passing	3/4" Rock	Mine Chat	Mfg Sand	Job Formula
3/4"	100			100
1/2"	76	100		94
3/8"	49	99	100	87
No 4	9	76	78	60
No 10	5	45	36	32
No 40	3	21	13	14
No 80	3	14	7	9
No 200	2.4	10.0	3.2	6.1

% Asphalt Cement Used : 5.3 %

MATERIAL	SOURCE	% USED
3/4" Rock	Cummins Materials @	25
	Tulsa, OK.	
Mine Chat	Bingham Sand & Gravel @	45
	Treece, Kansas	
Manufactured Sand	Cummins Materials @	30
	Tulsa, OK.	

# DESIGN MIX FOR SOURCE 4 PROJECT NO : CMC-66(286) 12247(04)

Percent	3/4" Chips	Mine Chat	Stone	Screenings	Sand	Job
Passing	-		Sand	Ũ		Formula
3/4"	100					100
1/2"	86	100	100			96
3/8"	46	99	100	100	100	90
No 4	7	49	61	95	98	66
No 10	3	6	19	64	88	39
No 40	3	1	6	26	21	13
No 80	3	1	4	23	2	9
No 200	2.4	0.3	2.4	15.5	0.2	6.0

% Asphalt Cement Used : 5.2 %

MATERIAL	SOURCE	% USED
3/4" Chips	Anchor Stone Co. @ Tulsa,	18
	OK.	
Mine Chat	Bingham S & G @ Miami,	23
	OK.	
Stone Sand	Anchor Stone Co. @ Tulsa,	10
a' .	OK.	
Screenings	Anchor Stone Co. @ Tulsa,	34
	OK.	
Sand	Loman Sand Co. @ Bixby,	15
	OK.	

Percent	3/4" Chips	3/8" Chips	Screening	Sand	Job Formula
passing		-			
3/4"	100				100
1/2"	65	100			97
3/8"	33	92	100		91
No 4	4	6	96	100	56
No 10	2	1	61	98	40
No 40	2	1	25	80	23
No 80	2	1	17	17	10
No 200	1.4	0.3	12.3	2.5	5.5

#### DESIGN MIX FOR SOURCE 5 PROJECT NO : VARIOUS PURCHASE ORDERS

% Asphalt Cement Used : 4.7 %

	COLIDOD	A LIGED
MATERIAL	SOURCE	% USED
3/4" Chips	Bellco Materials Co. @	10
•	Snyder, OK.	
3/8" Chips	Bellco Materials Co. @	35
•	Snyder, OK.	
Screenings	Bellco Materials Co. @	40
0	Snyder, OK.	
Sand	CC Sand @ Jenks, OK.	15

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#### Candidate for the Degree of

#### Master of Science

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Biographical:

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